

# Direct Bundle Simulation approach for the compression molding process of Sheet Molding Compound

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## Abstract

The manufacturing process of Sheet Molding Compounds (SMC) induces a reorientation of fibers during the flow, which influences local properties and is of interest for structural computations. Typically, the reorientation is described with an evolution equation for the second order fiber orientation tensor, which requires a closure approximation and multiple empirical parameters to describe long fibers. However, CT scans of SMC microstructures show that fiber bundles stay mostly intact during molding. Treating hundreds of fibers in such a bundle as one instance enables direct simulation on component scale. This work proposes a direct simulation approach, in which bundle segments experience Stokes' drag forces and opposing forces are applied to the fluid field. The method is applied to specimens with a double-curved geometry and compared to CT scans. The Direct Bundle Simulation provides increased accuracy of fiber orientations and enables prediction of fiber-matrix separation with affordable computational effort at component scale.

*Keywords:* Process simulation, CT analysis, Compression molding, Discontinuous reinforcement

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## 1. Introduction

Sheet Molding Compound (SMC) is a composite material with thermoset matrix material and discontinuous glass or carbon reinforcement fibers. SMC compression molding is an economic process to mass produce complex parts with considerably higher fiber lengths compared to injection molding. Typical parts include automotive body panels due to high surface qualities and in-mold coating capability.

The mechanical properties of SMC depend on local fiber orientation and fiber volume fraction, which can change significantly during flow. However, these properties are difficult to determine after molding, and predicting these properties in the early development process can reduce expensive corrections of mold design. Additionally, utilization of process induced fiber orientations improves the predictive quality of structural simulations [1].

The production of SMC typically starts with the production of semi-finished sheets on an SMC line. The first step is the application of resin to a carrier foil. Chopped fibers fall on this carrier foil in a random transversely isotropic orientation. Afterwards, a second carrier foil is placed on top of the first carrier foil and both of them run through sets of rolls that ensure proper impregnation of the fibers. Then, the material is coiled and stored. The viscosity increases in a maturing process. After maturing, the foils are removed and the material can be cut and stacked to an initial charge for molding. This initial charge has room temperature and is placed into a mold at elevated temperature ( $\approx 150^\circ\text{C}$ ). The mold is closed and SMC flows with its fibers in a complex shape. The mold can be opened after a few minutes of curing and the final part is released.

SMC rheology was first described with generalized Hele-Shaw type models treating SMC as a one-phase material [2]. Barone and Caulk [3] developed a model with lubrication layers at the mold and a central plug flow. This approach was extended by several authors [4–7].

Fiber reorientation is typically modeled based on fiber orientation tensors in-

30 introduced by Advani and Tucker [8] as

$$\mathbf{A} = \int_S \mathbf{p} \otimes \mathbf{p} \Psi(\mathbf{p}) dp \quad (1)$$

31 and

$$\mathbb{A} = \int_S \mathbf{p} \otimes \mathbf{p} \otimes \mathbf{p} \otimes \mathbf{p} \Psi(\mathbf{p}) dp. \quad (2)$$

32 Here,  $\mathbf{p}$  describes a fiber direction and  $dp$  is the surface element on a unit sphere  
 33  $S := \{\mathbf{p} \in \mathbb{R}^3 : \|\mathbf{p}\| = 1\}$ . These fiber orientation tensors  $\mathbf{A}$  and  $\mathbb{A}$  represent the  
 34 second and fourth moment of the fiber orientation distribution function  $\Psi(\mathbf{p})$   
 35 and thus are a statistical representation of the microstructure. The evolution  
 36 of the second order fiber orientation tensor  $\dot{\mathbf{A}}$  is often described with equations  
 37 that are based on Jeffery's pioneering work [9]. Assuming that fibers have a  
 38 large aspect ratio, his result may be written as

$$\dot{\mathbf{A}} = \nabla \mathbf{v} \cdot \mathbf{A} + \mathbf{A} \cdot (\nabla \mathbf{v})^\top - \mathbb{A} : (\nabla \mathbf{v} + (\nabla \mathbf{v})^\top) \quad (3)$$

39 for a given velocity gradient  $\nabla \mathbf{v}$ . Several empirical modifications have been  
 40 introduced to his work to account for fiber interactions [10], experimentally  
 41 observed orientation delays [11] and anisotropic diffusivity [12]. These models  
 42 require a closure of the fourth order fiber orientation tensor  $\mathbb{A}$ , which can be  
 43 expressed as an approximation only. Additionally, these models require the fiber  
 44 length to be much shorter than structural features of the part (scale separation),  
 45 which does not hold in a lot of cases for SMC.

46 As an alternative to these statistical descriptions, several authors have developed  
 47 models for single flexible fibers based on inextensible threads [13], bead chains  
 48 [14] and linked rigid bodies [15–18]. Typically, these models use lubrication  
 49 theory and contact formulations to model fiber-fiber interactions [19, 20] as well  
 50 as hydrodynamic drag forces to describe the long range interaction between fluid  
 51 and fiber [21, 22]. Two-way coupling using a field of body forces was presented  
 52 by Lindström and Uesaka [23, 24] to conserve momentum in the direct bundle  
 53 simulation. However, they utilize the drag of prolate spheroids, which leads to  
 54 a total drag force on a fiber that depends on discretization [25]. Direct models

55 have been utilized in representative volume elements to determine rheological  
56 properties [26, 27], contact properties in microstructures [28] and parameters  
57 of macroscopic fiber orientation models [21]. This approach of “computational  
58 rheometry” has also been applied to SMC represented as a planar network of  
59 fiber bundles that interact through local shear forces at contact points [29–31].

60 The application of direct fiber simulations at the component scale has been  
61 reported only scarcely due to the sheer number of fibers and a reduced number  
62 of fibers is typically computed. A bead chain model was used by Kuhn et al.  
63 [32] and constrained beams were suggested by Hayashi et al. [33] at this scale.  
64 A commercial tool utilizing direct fiber simulation is 3D TIMON by TORAY  
65 Engineering. However, the tool neglects anisotropy and two-way coupling, as it  
66 is run after the determination of the flow field. Additionally, it does not include  
67 any interactions between fibers and it seems to use only a small subset of test  
68 fibers.

69 The evolution of the fiber microstructure is a complex phenomenon. However,  
70 CT scans in this work show that most fiber bundles in the core of a part stay  
71 intact during SMC molding, while few bundles at the mold surface are disen-  
72 tangled. This observation is also reported in literature [34–37]. This behavior  
73 allows at least in some flow situations the simplifying assumption to treat hun-  
74 dreds of fibers as one bundle instance. This drastically reduces computational  
75 costs compared to direct fiber simulations, while improving disadvantages of  
76 approaches based on fiber orientation tensors. Hence, the compression mold-  
77 ing process of a full component with thousands of bundles is demonstrated in  
78 this contribution. Two-way coupling is achieved using a similar approach to  
79 Lindström and Uesaka [23] and results in anisotropic material flow.

## 80 **2. Direct Bundle Simulation**

81 The fundamental idea of Direct Bundle Simulation is the full description of fiber  
82 bundles as a chain of one-dimensional finite elements that experience hydrody-  
83 namic drag forces of the surrounding flow. Bundles are represented as truss

84 elements that transfer tensile load, but do not transfer bending torque due to  
 85 an assumed thread-like nature of the bundle mechanics. Bundle elements may  
 86 collide with walls or each other. Thus, the direct simulation eliminates the need  
 87 of empirical interaction parameters in common fiber orientation models such as  
 88 the Folgar-Tucker constant [10]. Further, this approach allows for the simula-  
 89 tion of fiber-matrix separation, as bundles move independently from the matrix  
 90 material flow.

### 91 *2.1. Matrix model*

92 The matrix material is subjected to large deformations when it fills the cavity.  
 93 Thus, the flow of matrix material during molding is described in a Eulerian  
 94 framework and interacts with the molds through contacts in a Coupled Eulerian-  
 95 Lagrangian approach [38]. An operator split is utilized to solve the conservation  
 96 of mass

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0 \quad (4)$$

97 with mass density  $\rho$  and fluid velocity  $\mathbf{v}$  as well as the conservation of momentum

$$\frac{\partial \rho \mathbf{v}}{\partial t} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = \nabla \cdot \boldsymbol{\sigma} + \mathbf{f}_h \quad (5)$$

98 with the stress tensor  $\boldsymbol{\sigma}$  and a point-wise body force field imposed by bundles  
 99  $\mathbf{f}_h$ . Such a point-wise body force field was also applied by Lindström et al. [39]  
 100 to model fiber-fluid coupling.

101 The conservation equations are split into a Lagrangian step containing only  
 102 source terms

$$\left. \frac{\partial \rho}{\partial t} \right|_L = 0 \quad (6)$$

$$\left. \frac{\partial \rho \mathbf{v}}{\partial t} \right|_L = \nabla \cdot \boldsymbol{\sigma} + \mathbf{f}_h \quad (7)$$

103 and a Eulerian step containing the convective terms

$$\left. \frac{\partial \rho}{\partial t} \right|_{\mathbf{E}} + \nabla \cdot (\rho \mathbf{v}) = 0 \quad (8)$$

$$\left. \frac{\partial \rho \mathbf{v}}{\partial t} \right|_{\mathbf{E}} + \nabla \cdot (\rho \mathbf{v} \otimes \mathbf{v}) = \mathbf{0}. \quad (9)$$

104 The first step is solved analogously to standard Lagrangian procedure on a  
 105 deforming mesh. In the second step, the deformed mesh is moved back to its  
 106 original position and the solution variables are updated using a second order  
 107 advection transport algorithm [40].

108 The problem is closed with a constitutive model that relates stress to the defor-  
 109 mation rate. The stress tensor may be decomposed to a spherical part  $\boldsymbol{\sigma}^\circ$  and  
 110 deviatoric part  $\boldsymbol{\sigma}'$  according to

$$\boldsymbol{\sigma} = \boldsymbol{\sigma}^\circ + \boldsymbol{\sigma}'. \quad (10)$$

111 Then, the spherical relation is expressed using an equation of state as

$$\boldsymbol{\sigma}^\circ = \rho_0 c_0^2 \left( 1 - \frac{\rho_0}{\rho} \right) \mathbf{I}, \quad (11)$$

112 where  $\rho_0$  denotes the mass density of the matrix at rest,  $c_0$  describes the speed  
 113 of sound, and  $\mathbf{I}$  is the second order identity tensor. The deviatoric relation is  
 114 expressed as isotropic Newtonian viscous behavior

$$\boldsymbol{\sigma}' = \eta \dot{\boldsymbol{\gamma}} \quad (12)$$

115 with the deviatoric engineering shear strain rate  $\dot{\boldsymbol{\gamma}}$  and the dynamic shear vis-  
 116 cosity  $\eta$ .

117 If fiber bundles are neglected, the interaction term  $\mathbf{f}_h$  vanishes and the model  
 118 describes homogeneous isotropic Newtonian flow of the matrix material in the  
 119 mold. However, fiber bundles move with the flow and the presence of fiber  
 120 bundles subjects the matrix to an additional force. The determination and  
 121 application of this contribution to the conservation of momentum is described  
 122 in the next two sections.

123 *2.2. Hydrodynamic interaction*

124 Stokes' law describes the total hydrodynamic drag force on a sphere with radius  
125  $\hat{R}$  as

$$\mathbf{F}_d = 6\pi\eta\hat{R}\Delta\mathbf{v}, \quad (13)$$

126 where the relative velocity  $\Delta\mathbf{v} = \mathbf{v} - \mathbf{v}_0$  describes the difference between the  
127 velocity of the surrounding viscous fluid  $\mathbf{v}$  and the velocity of the suspended  
128 sphere itself  $\mathbf{v}_0$ .

129 For incompressible Newtonian flows with negligible inertia, the absolute hydro-  
130 dynamic resistance is proportional to  $\eta R \|\Delta\mathbf{v}\|$ , independent of the actual shape  
131 of a suspended rigid body [41]. Thus, an equivalent radius  $\hat{R} = k_d R$  may be used  
132 for shapes different from spheres, where  $k_d$  describes a dimensionless correction  
133 coefficient and  $R$  is a typical linear dimension of the shape, e.g. the cylinder  
134 radius. The drag force is always opposing the direction of the relative velocity.  
135 Contrary to spheres, cylinders also create lift if they are subjected to a flow with  
136 an orientation angle  $\phi$ . Thus, a second analogous coefficient is used to describe  
137 lift forces. These lift forces act perpendicular to the relative velocity in direction  
138  $\mathbf{q}$  and are computed using a coefficient  $k_l$ . In this section, both coefficients are  
139 computed for a range of aspect ratios and orientation angles. The coefficients  
140 are interpolated using a fitting function that is later used to efficiently compute  
141 hydrodynamic forces on bundle segments in the Direct Bundle Simulation.

142 Multiple cylindrical segments are chained together to represent a bundle, as  
143 illustrated in Figure 1. Thus, only forces at the lateral bundle surface  $A$  con-  
144 tribute to the total hydrodynamic drag and lift. The ends of the bundle are  
145 neglected, as the surface is small compared to the lateral surface. Let  $\mathbf{p}$  be  
146 the direction of a cylinder positioned at  $\mathbf{x}_0 \in \Omega \subset \mathbb{R}^3$ , then any point of the  
147 cylinder can be described as  $\mathbf{r} = r\mathbf{e}_r + \psi\mathbf{e}_\psi + \zeta\mathbf{p}$ , where  $\{\mathbf{e}_r, \mathbf{e}_\psi, \mathbf{p}\}$  describes  
148 the local cylinder coordinate system. With this parametrization, the lateral  
149 cylinder surface is defined as

$$A := \{(r, \psi, \zeta) \in \mathbb{R}^3 \mid r = R, 0 < \psi < 2\pi, 0 < \zeta < L\}, \quad (14)$$

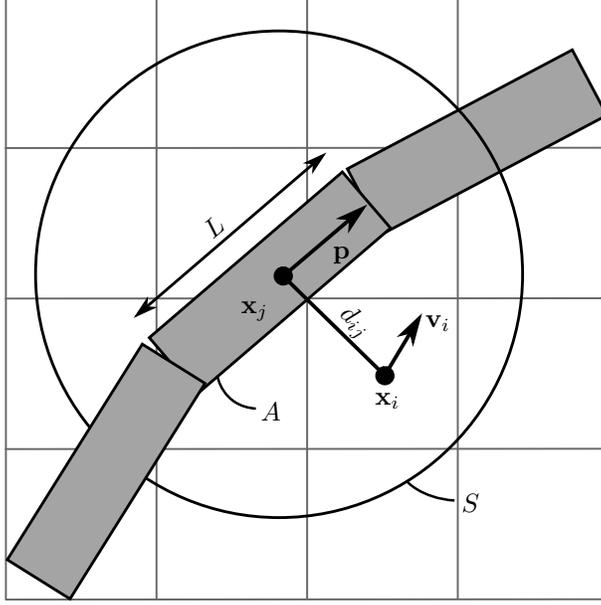


Figure 1: A bundle segment of length  $L$  and direction  $\mathbf{p}$  is placed in a mesh. The velocity of one exemplary element in the neighborhood  $S$  is shown with its current velocity  $\mathbf{v}_i$ .

150 where  $R$  is the cylinder radius and  $L$  is the length of a bundle segment. The  
 151 total hydrodynamic force exerted on the cylinder can be determined using an  
 152 integral over the lateral surface  $A$  as

$$\mathbf{F}_h = \int_A \boldsymbol{\sigma} \cdot \mathbf{n} \, dA \quad (15)$$

153 with surface normal  $\mathbf{n}$ .

154 To obtain this resistance force for cylinder aspect ratios and orientation angles  
 155 of interests, a parametric numerical study is performed. A cylinder with radius  
 156  $R = 0.5 \text{ mm}$  and aspect ratio  $r_p \in \{1, 2, 3, 5, 8, 13, 25\}$  is placed in the center of a  
 157 cube of fluid  $\Omega$  with edge length  $50 \text{ mm}$ . A uniform inlet velocity  $v_\infty = 1 \text{ mm s}^{-1}$   
 158 is applied at  $x_{\min}$  and a zero-pressure outlet is applied at  $x_{\max}$ . A slip condition  
 159 with no flux perpendicular to the wall is applied to all other four faces of the  
 160 cube. At the cylinder surface, a no-slip condition is applied. For each aspect  
 161 ratio, the orientation angle  $\phi$ , which describes the angle between the velocity

162 direction  $\mathbf{v}_\infty = v_\infty \mathbf{e}_x$  and the cylinder axis  $\mathbf{p}$ , is varied and a Finite Element  
 163 Analysis is performed to solve the incompressible steady Stokes flow problem

$$\mathbf{0} = \nabla \cdot \left( -p\mathbf{I} + \eta \left[ \nabla \mathbf{v} + (\nabla \mathbf{v})^\top \right] \right) \quad (16)$$

$$\mathbf{0} = \nabla \cdot \mathbf{v}. \quad (17)$$

164 After computing the velocity field for each configuration, the first and second  
 165 components of Eq. (15) are used in combination with Eq. (13) to compute the  
 166 coefficients

$$k_d = \frac{1}{6\pi\eta Rv_\infty} \int_A \boldsymbol{\sigma}_x \cdot \mathbf{n} \, dA \quad (18)$$

167 and

$$k_l = \frac{1}{6\pi\eta Rv_\infty} \int_A \boldsymbol{\sigma}_y \cdot \mathbf{n} \, dA \quad (19)$$

168 from the vertical and horizontal surface stress components  $\boldsymbol{\sigma}_x$  and  $\boldsymbol{\sigma}_y$ .

169 Figure 2 illustrates computed results for different aspect ratios and orientations  
 170 as points. Additionally, two fits have been determined as

$$k_d(r_p, \phi) = 1 - \alpha(r_p - 1) \cos(2\phi) + \beta(r_p - 1) \quad (20)$$

171 and

$$k_l(r_p, \phi) = \alpha(r_p - 1) \sin(2\phi) \quad (21)$$

172 with  $\alpha = 0.09$  and  $\beta = 0.3125$ .

173 The fitted Eq. (20) and Eq. (21) are plotted as solid lines in Figure 2. For aspect  
 174 ratio  $r_p = 1$ , the drag is similar to a sphere with  $k_d(1, \phi) \approx 1$  and  $k_l(1, \phi) \approx 0$ .  
 175 For other aspect ratios, the drag increases in a cosine-shape with orientations  
 176 closer to  $\phi = 90^\circ$  and with increasing aspect ratios. The lift force peaks, as  
 177 expected, at  $\phi = 45^\circ$  and follows a sine-shape with an amplitude increasing  
 178 with the aspect ratio.

179 Subsequently, it is assumed that micro-scale hydrodynamic effects of the veloc-  
 180 ity field are included in drag force and lift force. Therefore, bundle segments

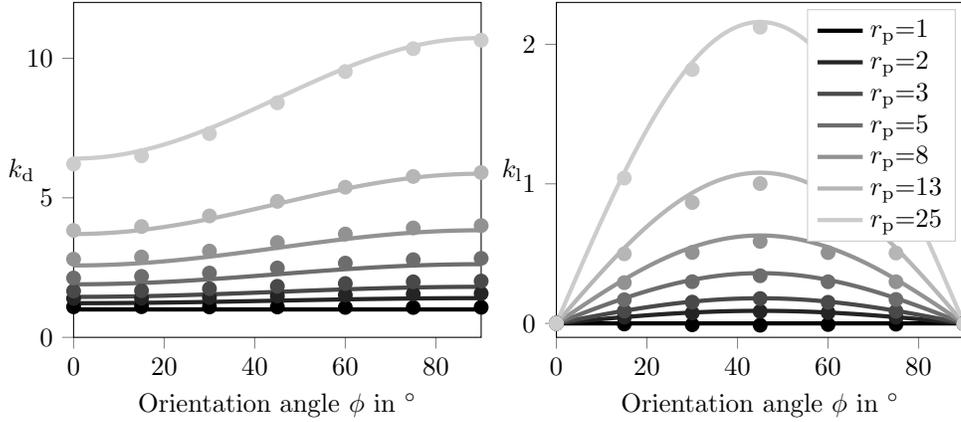


Figure 2: Dimensionless drag coefficient  $k_d$  and lift coefficient  $k_l$  from computation (dots) and fit according to Eq. (20) and Eq. (21). An orientation angle  $\phi = 90^\circ$  means that the cylinder is placed perpendicular to the flow direction and induces maximum drag, while  $\phi = 0^\circ$  refers to a cylinder aligned with the velocity  $\mathbf{v}_\infty$ .

181 experience only resulting forces and the computation does not need to account  
 182 for velocity gradients that occur at the subgrid micro-scale.

183 The surrounding fluid field is computed with a mesh-based approach in this  
 184 work. Hence, the relative velocity  $\Delta \mathbf{v}$  for drag computation has to be deter-  
 185 mined from nearby matrix elements, as illustrated in Figure 1. The search  
 186 radius for nearby elements is set to the length of a bundle segment  $L$  which  
 187 leads to the definition of the neighborhood of bundle segment  $j$  as  $S := \{i \in \mathbb{N} \mid$   
 188  $0 < \|\mathbf{x}_i - \mathbf{x}_j\| < L\}$ . Using this neighborhood definition, the relative velocity is  
 189 computed by a Gaussian weighting approach as

$$\Delta \mathbf{v}_j = \sum_{i \in S} \frac{w_{ij}}{W_j} (\mathbf{v}_i - \mathbf{v}_j) \quad (22)$$

190 with Gaussian weighting factors

$$w_{ij} = \exp\left(-\frac{9}{2} \frac{d_{ij}^2}{L^2}\right) \quad (23)$$

191 and  $W_j = \sum_{i \in S} w_{ij}$ . The Gaussian weights depend on the distance of a bundle  
 192 center to a neighboring element  $d_{ij}$ . The total hydrodynamic force on a bundle

193 segment  $j$  with aspect ratio  $r_p^{(j)}$  is computed as

$$\mathbf{F}_h^{(j)} = 6\pi\eta R \left( k_d(r_p^{(j)}, \phi)\Delta\mathbf{v} + k_1(r_p^{(j)}, \phi) \|\Delta\mathbf{v}\| \mathbf{q} \right) \quad (24)$$

194 utilizing the orientation angle

$$\phi = \arccos \left( \frac{\Delta\mathbf{v} \cdot \mathbf{p}}{\|\Delta\mathbf{v}\|} \right). \quad (25)$$

195 The direction of  $\mathbf{q}$  is computed from a projection as

$$\mathbf{q} = -\text{sgn}(\mathbf{p} \cdot \Delta\mathbf{v}) [\mathbf{p} - (\mathbf{p} \cdot \llbracket \Delta\mathbf{v} \rrbracket) \llbracket \Delta\mathbf{v} \rrbracket] \quad (26)$$

196 Here,  $\mathbf{p}$  is a unit vector and the operator  $\llbracket \cdot \rrbracket$  computes a unit vector in the  
197 direction of its input and is defined as  $\llbracket \cdot \rrbracket = (\cdot) / \|\cdot\|$ .

198 After computation of drag forces, the same weights  $w_{ij}$  are used to apply an  
199 opposing force to each mesh element  $i \in S$ . The contribution of each bundle  $j$   
200 to the coupling body force in element  $i$  is expressed as

$$\mathbf{f}_h^{(ij)} = -\frac{1}{V_i} \frac{w_{ij}}{W_j} \mathbf{F}_h^{(j)} \quad (27)$$

201 with the volume  $V_i$  of the  $i$ -th element. The total body force field  $\mathbf{f}_h$  is then  
202 obtained by summing over contributions from all bundles in each element.

### 203 *2.3. Interaction between fiber bundles*

204 Fiber bundles may collide with mold walls, other bundles or themselves. The  
205 collision is treated with a kinematic contact constraint normal to the collision di-  
206 rection utilizing Abaqus' built-in general contact algorithm. All artificial damp-  
207 ing parameters are set to zero, because the fluid interaction provides sufficient  
208 damping. The tangential friction between fiber bundles is neglected for now,  
209 which is a significant simplification. The implication of this simplification is  
210 discussed in more detail in Section 5.

### 211 *2.4. Implementation*

212 The described model is implemented in Abaqus explicit using several subrou-  
213 tines. A VUFIELD subroutine is called at each node to copy node velocities

214 and positions to field variables. The field variables are then interpolated at each  
 215 integration point by a VUSDFLD subroutine and copied to global arrays. The  
 216 main task of drag force computation is then treated in a VDLOAD subroutine.  
 217 Eq. (22) is used to compute the relative velocity at each bundle segment, which  
 218 is then used to compute drag forces based on Eq. (24) utilizing the coefficients  
 219 in Eq. (20) and Eq. (21). An opposing force is saved for all neighboring Eulerian  
 220 elements  $i \in S$ . Subsequently, Eulerian elements are subjected to a body  
 221 force field  $\mathbf{f}_h$  computed from the stored drag force and its volume according to  
 222 Eq. (27).

### 223 2.5. Verification

224 The motion of a single bundle in shear flow is simulated in order to verify the  
 225 model. The fiber bundle has a length of 25 mm and is subjected to a shear rate  
 226  $\dot{\gamma} = 10 \text{ s}^{-1}$ . The domain for this simulation is

$$\Omega = \{\mathbf{x} \in \mathbb{R}^3 \mid -C < (x_2, x_3) < C, -2C < x_1 < 2C\} \quad (28)$$

227 with  $C = 20 \text{ mm}$ . The bundle is placed at the center, discretized with ten  
 228 segments and positioned vertically, so that the initial orientation is  $\theta = 0$ .  
 229 Figure 3 shows bundle position and velocity in x-direction shortly after starting  
 230 the simulation. The contour plot of the horizontal velocity component depicted  
 231 in Figure 3 indicates the two-way coupled nature of the presented approach.  
 232 Although the bundle is flexible, it behaves like a rigid body until alignment  
 233 with the flow due to the positive normal stress in the direction of the bundle  
 234 axis.

235 A reference solution for this test case is given by Jeffery's equation for a single  
 236 ellipsoid without buoyancy and inertia [9, 42] in the 2D case as

$$\frac{d\theta}{dt} = \frac{\dot{\gamma}}{2} (1 + \xi \cos 2\theta). \quad (29)$$

237 Bretherton [43] showed that this equation is also valid for shapes other than  
 238 ellipsoids, if an equivalent aspect ratio  $r_e$  is used in the shape factor  $\xi = (r_e^2 -$

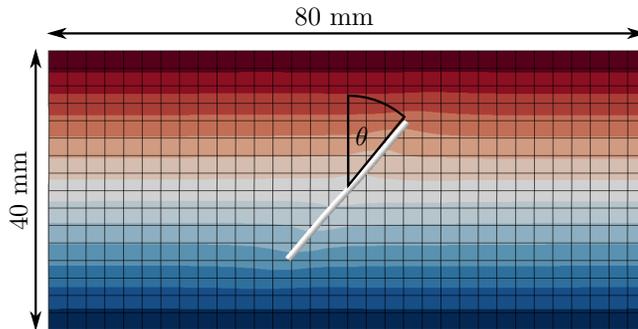


Figure 3: The contour plot shows a fiber bundle discretized with ten segments in a shear flow. The color codes indicate the velocity in x-direction (dark red is  $200 \text{ mm s}^{-1}$ , dark blue is  $-200 \text{ mm s}^{-1}$ ). The fluctuations at both ends show how the two-way coupling influences the macroscopic velocity field.

239  $1)/(r_e^2 + 1)$ . Such equivalent aspect ratios can be determined from the work of  
 240 Goldsmith and Mason [42] or Cox [44], who suggested the empirical formula

$$r_e = 1.24 \frac{r_p}{\sqrt{\ln r_p}} \quad (30)$$

241 to determine the equivalent aspect ratio  $r_e$  from a cylinder aspect ratio  $r_p$ .

242 Figure 4 compares the orientation evolution of the Direct Bundle Simulation  
 243 with ten truss elements and two truss elements to the solution of Eq. (29). The  
 244 simulation is in good agreement with the reference solution for both discretiza-  
 245 tions. Additionally, a bundle with a bundle aspect ratio  $r_p = 25$  is placed  $90^\circ$   
 246 to the flow under the same conditions as in the parameter identification (see  
 247 section 2.2) and meshed with one and ten segments. The resulting drag force  
 248 normalized with  $6\pi\eta Rv_\infty$  is 9.31 and 9.55, respectively. This is close to each  
 249 other, but slightly smaller than the drag coefficient shown in Figure 2, because  
 250 the averaged velocity around the bundle is smaller than the nominal velocity  
 251 far away. Anyway, the orientation result and the total drag indicate that bun-  
 252 dle motion is generally only slightly affected by discretization. However, the  
 253 effect on the flow field changes and the approach is not entirely independent of  
 254 discretization, as one chooses which effects are included in the drag coefficients  
 255 and which are resolved on the mesh by setting the bundle segment length.

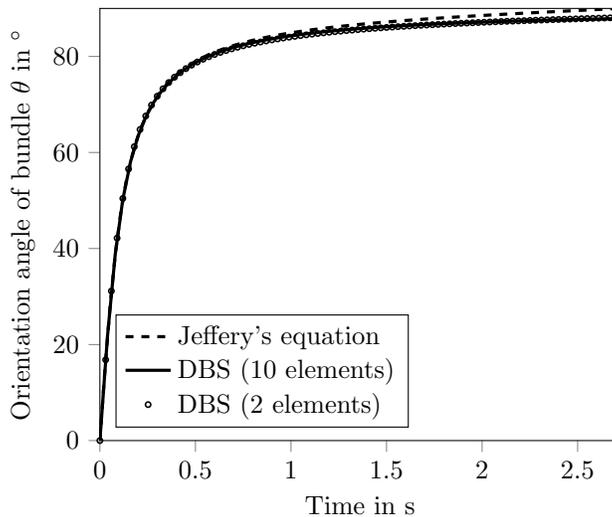


Figure 4: Comparison of bundle orientation angle computed from Direct Bundle Simulation and Jeffery's equation.

256 There is a small difference between simulation and analytical solution at the  
 257 almost horizontal state in Figure 4. At this point, torque induced by friction  
 258 at the lateral surface dominates bundle motion. In SMC, bundles are heavily  
 259 confined by other bundles and the mold. It is assumed that the torque that  
 260 spins a free bundle in a dilute situation is small compared to the confinement  
 261 effects and it is therefore neglected here.

### 262 **3. Application at component scale**

#### 263 *3.1. Molding trials*

264 In this work, a structural SMC based on an UPPH resin system with a com-  
 265 position shown in Table 1 is used. This two-step curing resin was developed  
 266 to improve co-molding with unidirectional carbon fiber patches due to a higher  
 267 viscosity in the B-stage [45].

268 The specimen under investigation is a hat profile with outer dimensions 120 mm  
 269 x 94 mm and a final thickness of 2 mm. Two variants are molded: Variant

Table 1: Composition of UPPH Sheet Molding Compound

Component	Trade name	Quantity
UPPH resin	Daron ZW 14141	100 parts
Flow aid	BYK 9085	2 parts
Impregnation aid	BYK 9076	3 parts
Deaeration aid	BYK A-530	0.5 parts
Inhibitor	pBQ	0.3 parts
Peroxide	Trignox 117	1 part
Isocyanate	Lupranat M20R	24.2 parts
Glass fiber	Multistar 272 4800 80	23 vol%

270 "S" (split configuration) consists of two SMC stacks ("S1" and "S2") with di-  
 271 mensions 80 mm x 30 mm x 5.3 mm that are manually placed in the mold as  
 272 illustrated in Figure 5 with dotted outlines. This split stack allows the inves-  
 273 tigation of weld line formation during the flow. The second variant "A" uses  
 274 an asymmetric placement of a single stack with dimensions 80 mm x 60 mm x  
 275 5.3 mm and enables a longer flow path. The mold is heated to 145 °C and closed  
 276 with a hydraulic press. The maximum press force was limited to 50 kN.

### 277 3.2. CT Analysis

278 The molded samples were analyzed by volumetric imaging using an Yxlon X-ray  
 279 CT system with a Perkin Elmer flat panel Y.XRD1620 detector and a reflection  
 280 tube by Comet. The detector has a resolution of 2048 × 2048 pixels and a pixel  
 281 pitch of 200 μm. Acceleration voltage, current, exposure time and frame binning  
 282 were set to 150 kV, 0.05 mA, 1000 ms and 2, respectively. A 16-bit volumetric  
 283 image gray scale image is reconstructed based on 2400 projections over 360°  
 284 and the Feldkamp, Davis and Kress (FDK) algorithm [46]. The voxel size of the  
 285 resulting volumetric image is 68.7 μm.

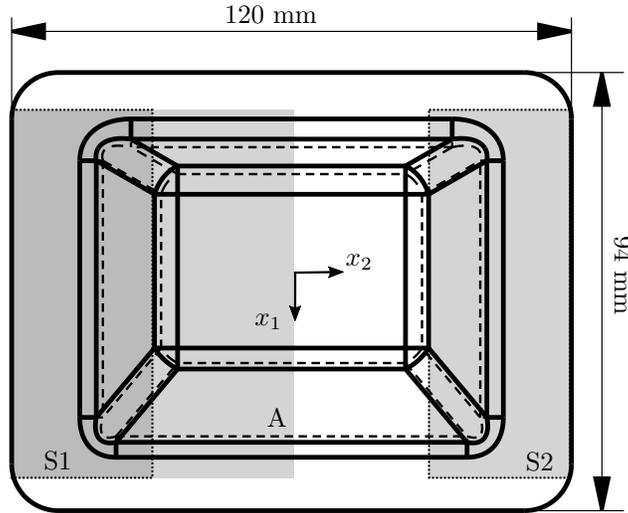


Figure 5: The molded part has outer dimensions 120 mm x 94 mm. For the split stack configuration, two SMC stacks "S1" and "S2" are placed at the light gray areas with dotted outlines. For the asymmetric configuration, a single stack ("A") is placed on one side of the mold.

### 286 3.3. Compression Molding Simulation

287 The molding process is simulated using Abaqus explicit utilizing the Coupled  
 288 Lagrangian Eulerian (CEL) feature. In this method, operator splitting is applied  
 289 to divide the momentum equation in a Lagrangian step and a subsequent  
 290 Eulerian step for material transport, as explained in Section 2.1. The fluid phase  
 291 is represented by an element-wise material volume fraction and an immersive  
 292 boundary is reconstructed at each step for interactions with the molds [38].  
 293 Fiber bundles interact with the SMC phase exclusively through the subroutines  
 294 described in Section 2.4.

295 The total part volume is  $25\,410\text{ mm}^3$ , which leads to a bundle volume of  $5844\text{ mm}^3$   
 296 at the given nominal fiber volume fraction. The roving used for SMC produc-  
 297 tion is a 4800 Tex multi-end roving with 80 strands and fiber diameter of  $14\text{ }\mu\text{m}$ .  
 298 Hence, each bundle is comprised of approximately 200 fibers, which leads to  
 299 a total amount of 7600 bundles with 25 mm length in the part. The initial  
 300 microstructure for the simulation is generated by sampling bundle directions

301 randomly from a uniform planar-isotropic fiber orientation distribution. The  
 302 bundles are then randomly shifted such that at least one node remains in the  
 303 stack volume. This way, a statistically uniform fiber volume fraction is achieved  
 304 in the stack region. Each bundle is discretized with ten linear truss elements  
 305 and all elements outside the stack are cut, similar to the physical process, in  
 306 which the stack is cut from an SMC sheet.

307 Additionally, Eulerian elements are used to represent the molding domain. Only  
 308 those Eulerian elements occupied by initial stack positions are initially filled with  
 309 material. Both mold halves are represented by rigid shell elements. They interact  
 310 with the SMC paste through hydrodynamic friction

$$\boldsymbol{\tau} = -\lambda \left( \frac{\|\mathbf{v}_{\text{rel}}\|}{v_0} \right)^{m-1} \mathbf{v}_{\text{rel}} \quad (31)$$

311 with a friction coefficient  $\lambda$ , a reference velocity  $v_0$ , a power law coefficient  $m$   
 312 and the relative velocity in the contact plane  $\mathbf{v}_{\text{rel}}$ . This formulation is quite  
 313 common and physically motivated by a resin-rich lubrication layer near the hot  
 314 mold [6, 47]. Parameters are estimated from a similar material system [48] and  
 315 listed in Table 2.

316 The explicit time integration requires an extremely small time increment due to  
 317 the high resin viscosity. The mass of the entire model was therefore scaled by  
 318 a factor  $\kappa_m$  to improve the time increment, while ensuring that kinetic energy  
 319 remains negligible small compared to the external work. The viscosity domi-  
 320 nated time step scales linearly with density. Additional simulation parameters  
 321 are listed in Table 2.

322 While the lower mold is constrained at a fixed position, the upper mold is  
 323 closed with the profiles given in Figure 6. These profiles are an idealization  
 324 to save computational time during the initial forming process, before the flow  
 325 of material starts. There is some variation in the experimental profiles, which  
 326 can be attributed partly to a non-uniform thickness of SMC sheets and to the  
 327 reaction time of the press control unit. The simulation stops after a complete  
 328 fill with the final part height and does not include the subsequent holding and

Table 2: Simulation Parameters

Property	Symbol	Value
Resin viscosity	$\eta$	$1 \times 10^5 \text{ Pa s}$
Resin mass density	$\rho_r$	$1900 \text{ kg m}^{-3}$
Resin speed of sound	$c_0$	$1000 \text{ m s}^{-1}$
Bundle elastic modulus	$E$	$73 \text{ GPa}$
Bundle density	$\rho_b$	$2600 \text{ kg m}^{-3}$
Bundle radius	$R$	$0.1 \text{ mm}$
Bundle segment length	$L$	$2.5 \text{ mm}$
Mold friction coefficient	$\lambda$	$1 \times 10^6 \text{ N s m}^{-3}$
Mold friction exponent	$m$	$0.6$
Reference velocity	$v_0$	$0.001 \text{ m s}^{-1}$
Mass scaling factor	$\kappa_m$	$1 \times 10^6$
Time step	$\Delta t$	$3 \times 10^{-4} \text{ s}$

329 curing process. The computational time for the simulation is approximately 22  
330 hours on a single workstation with a Intel Xeon E5 2667V2 CPU.

331 A conventional simulation utilizing fiber orientation tensors and Jeffery’s equa-  
332 tion is used to compare the Direct Bundle Simulation to the macroscopic orien-  
333 tation model given in Eq. (3). A VUMAT subroutine with six state variables  
334 and an IBOF closure approach [49] for the fourth order fiber orientation tensor  
335  $\mathbb{A}$  was implemented to compute fiber orientations instead of the bundle motion.  
336 In this conventional approach, no two-way coupling was included. The initial  
337 fiber orientation is described by a planar isotropic fiber orientation tensor and  
338 all other conditions remain unchanged.

#### 339 4. Results

340 Figure 7 provides an overview on the compression molding process simulation  
341 for the split stack configuration ”S”. The initial mold gap at  $t = 0 \text{ s}$  is 20 mm

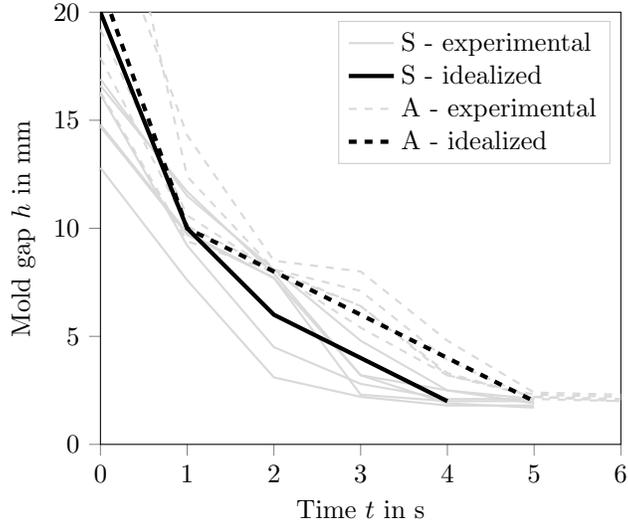


Figure 6: Distance between upper and lower mold during the flowing phase of SMC. Six parts of the split configuration "S" were produced and are shown with solid gray lines. Four parts of the asymmetric configuration "A" were produced and are shown with dashed gray lines. Additionally, the idealized mold profiles for simulations are shown in solid black and dashed black for the "S" and "A" configuration, respectively.

342 and the upper mold is just not touching the SMC stacks. Closing the mold with  
 343 the high initial closing speed deforms the stacks, but does not start material  
 344 flow. During forming, the two-way coupled approach pulls the stack sideways  
 345 in the hat-shaped mold. This can be observed by the lateral deformation of the  
 346 stack tips depicted at  $t = 2$  s in Figure 7. The mold gap is reduced to the initial  
 347 stack height of 5.3 mm after approximately two seconds. From there on, flow  
 348 dominates the mold filling process and fiber bundles are carried with the SMC  
 349 until the final part thickness of 2 mm is reached.

#### 350 4.1. Orientation and separation effects

351 Figure 8 shows slices through the midplane of the upper and lower planar re-  
 352 gions of the scanned part in split stack configuration. Additional slices through  
 353 thickness are provided in Figure A.14 in the appendix. The white strands repre-  
 354 sent fiber bundles, which remain in their bundled structure even for the applied

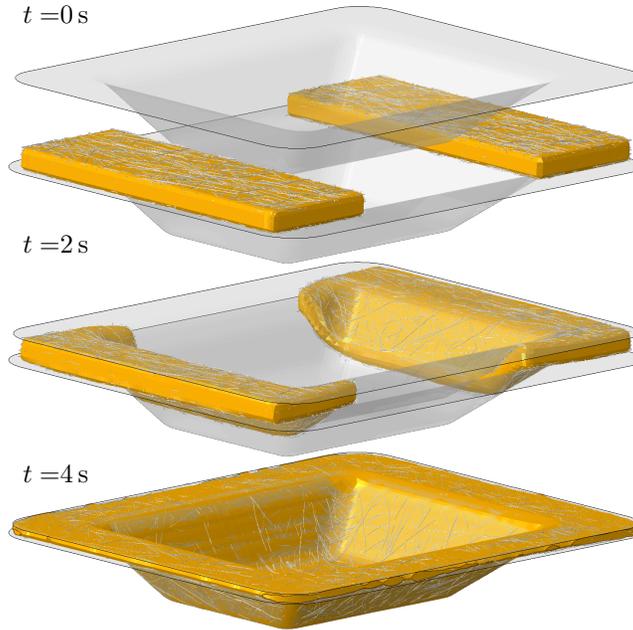


Figure 7: Snapshots of the molding process for the split stack configuration "S". The compression molding process starts with a deformation of the two initial stacks. Subsequently, the SMC is forced to flow until the part reaches its final thickness of 2 mm. The Direct Bundle Simulation approach lets bundles deform and flow with the matrix material while enforcing two-way coupling. Therefore, the flow is naturally anisotropic and depends on the current bundle configuration.

355 high degree of deformation. The weld line features a severe fiber-matrix separation and only a small amount of fiber bundles bridges the gap in this zone.  
 356  
 357 The inner slice in Figure 8 even shows some pores. Regions close to the mold boundaries and the weld line show a bundle alignment parallel to the boundary.  
 358  
 359 Bundles perpendicular to the boundary are likely pulled out of this region by forces acting over the entire length of the bundle and parallel bundles remain  
 360 close to the boundaries. Regions farther away from boundaries show a regular  
 361 random in-plane orientation.  
 362

363 The Direct Bundle Simulation result is sliced in the same planes and the result  
 364 is depicted in Figure 9. The simulation results show a slightly larger area of

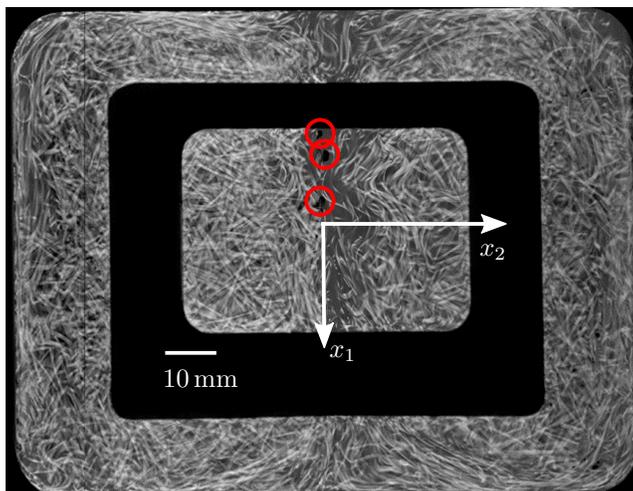


Figure 8: Slices through the upper and lower planar regions of the CT Scan. Fiber bundles stay intact during molding and fiber-matrix separation can be observed at the weld line. The weld line region includes pores (marked with red circles) close to the origin of the coordinate system.

365 fiber-matrix separation and no bundles bridge the resin-rich weld line. Similar  
 366 to the CT-scan, boundary regions show a predominant orientation parallel to  
 367 the boundaries.

368 For a quantitative comparison of the Direct Bundle Simulation to a simulation  
 369 based on fiber orientation tensors and the CT scans, bundle orientations are  
 370 evaluated on a uniform  $12 \times 16 \times 4$  grid of sub-volumes. The discrete second-  
 371 order fiber orientation tensor for each of the sub-volumes is computed as

$$\mathbf{A} = \frac{1}{N} \sum_{i=1}^N \mathbf{p}_i \otimes \mathbf{p}_i \quad (32)$$

372 with  $N$  being the number of truss elements in the sub-volume.

373 The slices of the CT scan shown in Figure 8 are analyzed in 2D using Ori-  
 374 entationJ [50], such that a major direction is assigned to each  $10 \times 10$  pixel area.  
 375 Then the same discrete fiber orientation tensor definition given in Eq. (32)  
 376 and the same  $12 \times 16$  grid is used to represent the orientation state as tensor  
 377 components.

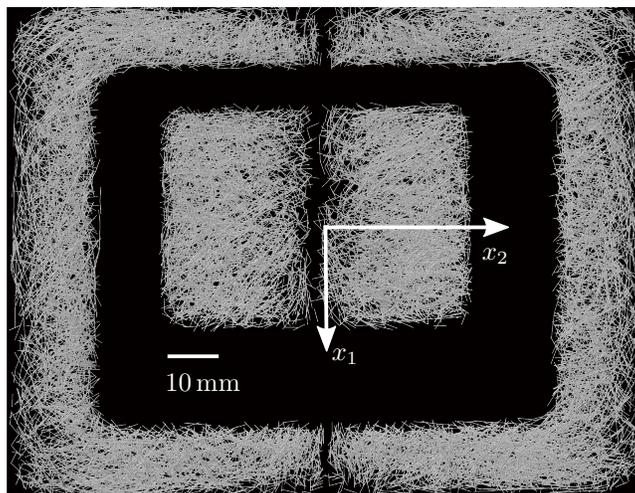


Figure 9: Slices through the planar regions of the Direct Bundle Simulation result. Each gray cylinder represents a bundle segment consisting of 200 individual fibers. The weld line at the center is matrix rich and no bundles gap the this region. Bundles close to the boundaries show a reduced fiber volume fraction and more bundles oriented parallel to the boundary.

378 A comparison of the Direct Bundle Simulation approach, CT scan and the con-  
 379 ventional fiber orientation model is depicted in Figure 10 for the split stack  
 380 configuration. The  $A_{11}$ -component of the CT-analysis features three signifi-  
 381 cantly higher oriented vertical stripes at both ends of the mold and the weld  
 382 line. Conversely, the  $A_{22}$ -component of the CT-analysis indicates a dominant  
 383 orientation in horizontal direction at the top and bottom mold boundaries with  
 384 lower values at the vertical mold boundaries to the left and right of the figure.  
 385 The corresponding Direct Bundle Simulation is able to reproduce these three  
 386 stripes of higher vertical orientation at the correct positions. Characteristic gra-  
 387 dients and the level of orientation is predicted well. The conventional approach  
 388 using fiber orientation tensors and Jeffery's equation does not account for the  
 389 constraints at mold walls and shows a homogeneous orientation distribution. In  
 390 homogeneous regions, such as the inner slice with some distance to the weld  
 391 line, Jeffery's equation leads to a reasonable prediction of the orientation state.  
 392 The Direct Bundle Simulation limits any bundle orientation normal to the

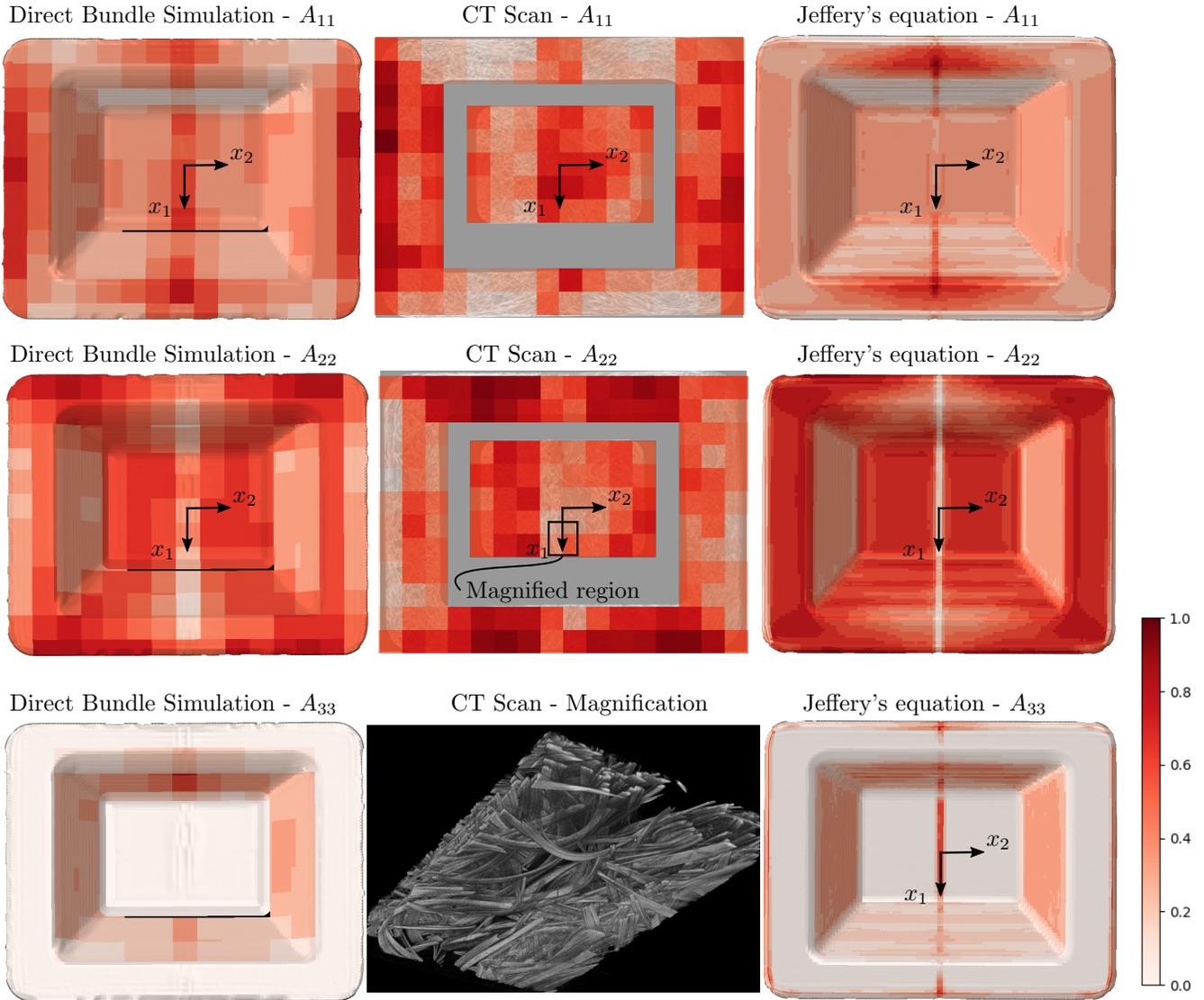


Figure 10: Comparison of Direct Bundle Simulation results with CT Analysis and fiber orientation tensor based computation utilizing Jeffery's equation for the split stack configuration. The first row shows orientation tensor component  $A_{11}$  which indicates vertical fiber orientation in this representation. The second row shows the  $A_{22}$ -component representing horizontal fiber orientation. The third row shows the  $A_{33}$ -component representing fiber orientation normal to the observation plane. The orientation analysis of the CT image slices is limited to two dimensions. Thus, the central image in the third row shows a high resolution CT scan of the region indicated in the illustration above. The magnified view reveals a dominant in-plane orientation of bundles.

393 molds, because bundle segments cannot be physically arranged in normal di-  
 394 rection in the constrained mold gap. Thus, the  $A_{33}$ -component is small in the  
 395 planar regions of the part. An investigation of a magnified CT Scan with higher  
 396 resolution confirms that fiber bundles at the weld line are primarily oriented  
 397 in-plane. The computation based on fiber orientation tensors shows a dominant  
 398 normal component of fiber orientation at the weld line.

399 Figure 11 is analogous to Figure 10, but describes the evaluation of the asym-  
 400 metric stack configuration with a maximum flow path of 60 mm in  $x_2$ -direction.  
 401 This configuration confirms observations of the previous case with significantly  
 402 higher orientations parallel to mold walls that are not described by tensor based  
 403 theory. Despite a longer flow path, the magnitude of re-orientation is similar to  
 404 the split stack configuration due to a similar stretch in  $x_2$ -direction (50% initial  
 405 mold coverage each).

#### 406 4.2. Bundle curvature

407 The curvature of bundles is evaluated as

$$\kappa = \frac{2}{L} \tan \left( \frac{1}{2} \arccos \left( \llbracket \mathbf{p}^{(j)} \rrbracket \cdot \llbracket \mathbf{p}^{(k)} \rrbracket \right) \right) \quad (33)$$

408 at each node connecting two neighboring bundles  $j$  and  $k$ . A contour plot of the  
 409 curvature for the split stack configuration is plotted in Figure 12. It shows that  
 410 the largest curvatures occur at corners and close to the weld line. The curvature  
 411 at the weld line originates probably from a flow in  $x_1$  direction compressing  
 412 bundles to a zig-zag shape. The curvature in the CT scan is obtained only  
 413 for the central region in order to have sufficient resolution for tracking bundle  
 414 curvature [51].

415 The projection of curvature values on the  $x_1$  direction is plotted in Figure 13.  
 416 The maximal values of the CT scan agree well with the maximal curvatures  
 417 computed from the direct bundle simulation. The mean curvature of the CT  
 418 scan is higher in this representation, but this is likely influenced by the lower  
 419 values outside the center region which are not taken into account for the CT

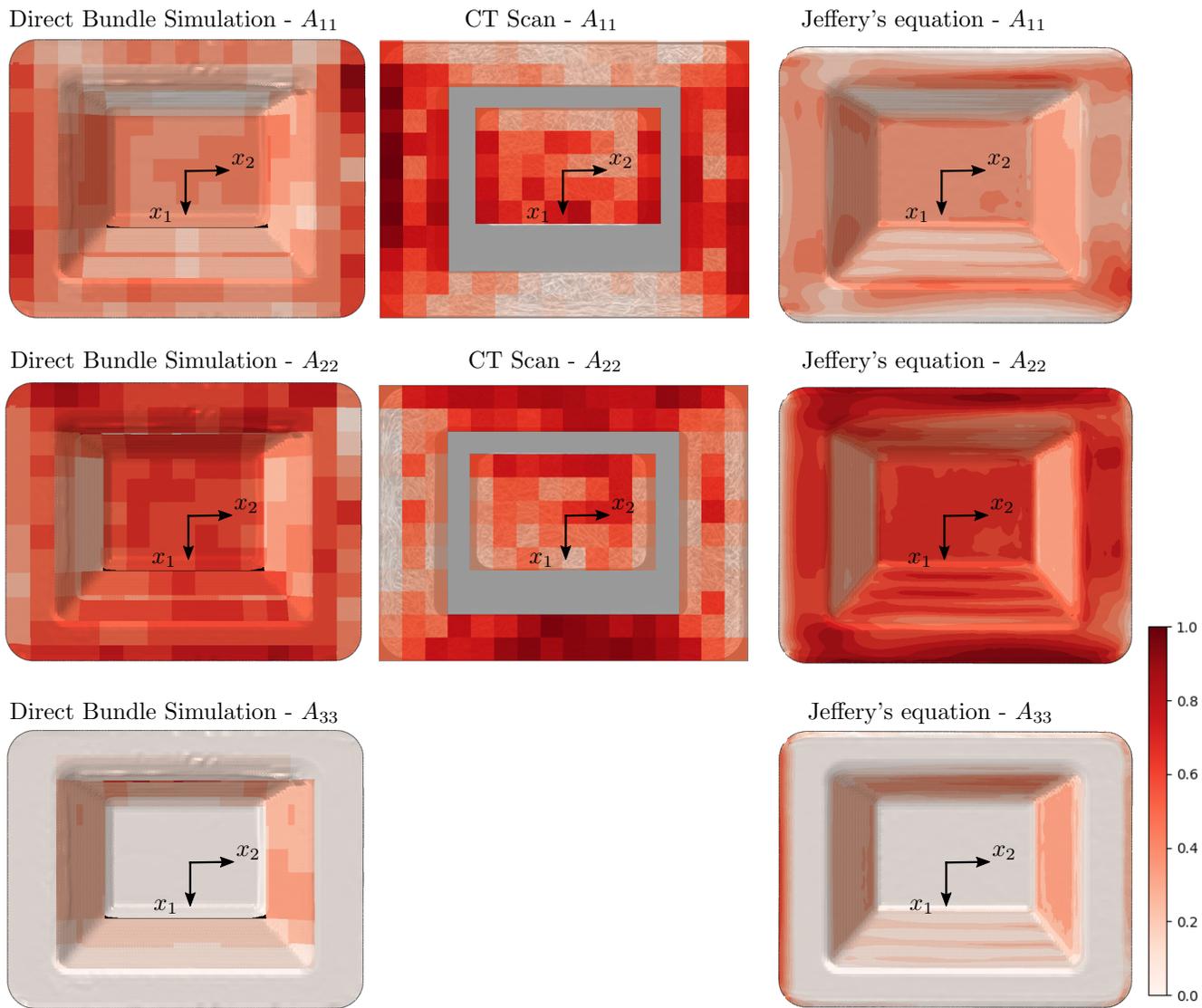


Figure 11: Comparison of Direct Bundle Simulation results with CT Analysis and fiber orientation tensor based computation utilizing Jeffery's equation for the asymmetric stack configuration. Refer to Figure 10 for a detailed explanation of the layout.

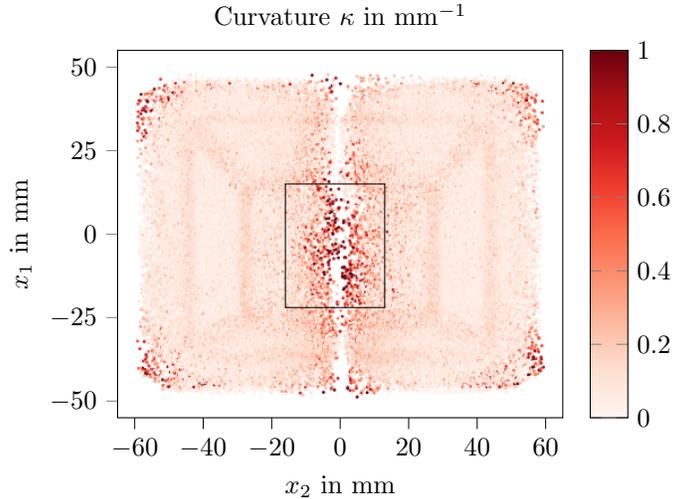


Figure 12: Simulation results of bundle curvature. The highest values occur at the corners of the mold and at the weld line. The parts three dimensional shape is visible in this plot due to the bending of bundles at curvatures of the geometry. High resolution CT data for curvatures is obtained for the central area marked with a black rectangle.

420 data. It should be mentioned that simulated curvature might depend on the  
 421 segment length of bundles.

## 422 5. Discussion

### 423 5.1. Simplifications and Limitations

424 The entire flow of material is assumed to be isothermal in this work. This  
 425 assumption is quite common for the bulk material of SMC, as the time scale of  
 426 thermal diffusivity in SMC is large compared to the time it takes the material  
 427 to flow (less than 5 s). Consequently, curing during the flow is also neglected.  
 428 The heating and curing of bulk material is a relevant process in the subsequent  
 429 holding phase though, which takes approximately 2 min.

430 The matrix is treated as a purely viscous Newtonian fluid, because shear thin-  
 431 ning behavior of the matrix system is currently not available. Typically, SMC  
 432 matrix is described with a non-Newtonian power law model [6, 7, 52], which

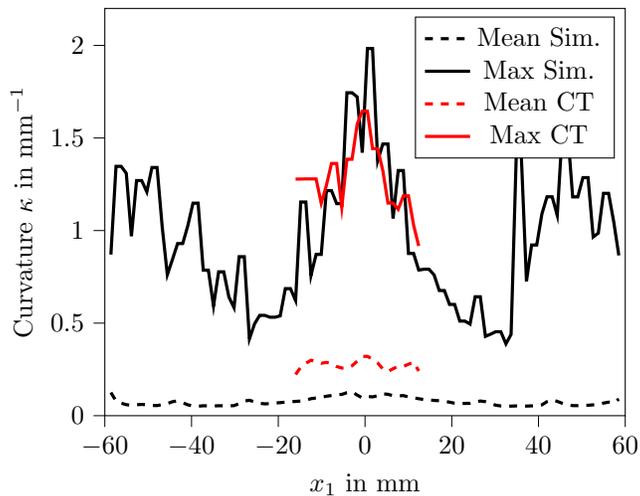


Figure 13: Curvatures projected to the  $x_2$  axis.

433 has certainly an influence on the necessary compression force. However, the  
 434 method is by no means limited to Newtonian viscosity. The characterization  
 435 can be performed in a standard rheometer without fibers and does not require  
 436 complex in-mold measurements.

437 Fiber bundles are represented with truss elements which neglect bending stiff-  
 438 ness and transfer tension only. This is based on the assumption that bundles  
 439 have much higher bending compliance compared to a homogeneous cylinder.  
 440 Bending and tension are likely decoupled at the meso-scale, as individual fila-  
 441 ments may slide in relative motion. However, modeling the complex mechanics  
 442 of a bundle and its sizing as a truss is a simplification in the present model. Truss  
 443 elements imply a cylindrical shape for collisions in the current implementation.  
 444 This is a simplification, because bundles in the actual process are mostly flat.  
 445 Further work is required to investigate the effect of bundle shape on resulting  
 446 micro structures. Additionally, short range hydrodynamic interactions (lubri-  
 447 cation forces) between bundles are neglected. These interaction forces occur if  
 448 bundles come in close contact and matrix material is sheared in the small gap  
 449 between them.

450 An a priori estimate for the number of contacts per bundle segment is given as

$$N_c = 4f \left( \frac{2}{\pi} \frac{L}{2R} \Phi_1 + \Phi_2 + 1 \right) \quad (34)$$

451 with the orientation functions  $\Phi_1 = \Phi_2 = 2/\pi$  for a 2D random fiber distribu-  
452 tion and volume fraction  $f$  [53]. This estimate predicts about 6.2 contacts per  
453 bundle segment, which makes the incorporation of short-range hydrodynamics  
454 necessary for the correct prediction of compression forces. An evaluation of the  
455 direct bundle simulation leads to an average of 4.6 to 5.0 contacts per bundle  
456 (see appendix Appendix B). This evaluation is in good agreement with the es-  
457 timate given in equation (34). An additional challenge in modeling lubrication  
458 is the increasing sheared area due to flattening bundles [35]. The introduction of  
459 lubrication effects and corresponding experimental investigations with pressure  
460 sensors will be addressed by the authors in future work.

### 461 *5.2. Comparison of Direct Bundle Simulation to the State of the Art*

462 The Direct Bundle Simulation is able to predict fiber-matrix separation effects  
463 at the weld line and thus enables a better description of structural weak spots  
464 in such areas. The simulated matrix-rich region is slightly larger than in the  
465 investigated sample. This might be caused either by the experimental setup,  
466 because the part was compressed further than the nominal thickness, or by the  
467 simplifications of the model (bundle shape and friction).

468 The presented approach is a natural access to modeling anisotropic flow. Other  
469 simulations based on fiber orientation tensors may incorporate the fourth order  
470 fiber orientation tensor to describe the fourth order viscosity tensor. However,  
471 the fourth order orientation tensor must be approximated by a closure, which  
472 becomes increasingly inaccurate, if only a few bundled directions are dominant.

473 At regions close to the mold walls and the weld line, Direct Bundle Simulation  
474 accounts for spatial constraints of the fiber orientation due to mold boundaries  
475 and leads to more accurate fiber orientation results. This is expected to be useful  
476 for narrow features such as ribs or beads. Nonetheless, Jeffery's equation leads

477 to reasonable results in planar, homogeneous regions and has an approximately  
478 ten times faster computational time. Therefore, a hybrid approach with bundles  
479 in critical regions might be a solution to improve computational efficiency for  
480 large SMC parts.

481 Finally, simulation and experiment represent only single realizations of random  
482 processes. The ability to run multiple simulations with different initial mi-  
483 crostructures may help estimating process reliability and statistical deviations  
484 in future.

## 485 **6. Conclusion**

486 The Direct Bundle Simulation approach treats fiber bundles in SMC as one-  
487 dimensional instances that move independent of the matrix material and inter-  
488 act through hydrodynamic forces as well as contact forces. The computational  
489 effort is greatly reduced compared to a simulation of all fibers by utilizing the  
490 observation that most bundles stay in a bundled configuration during SMC com-  
491 pression molding. The approach reproduces Jeffery's equation for a single fiber  
492 bundle in shear flow. A part with double-curved geometry was molded using two  
493 initial charges in order to force formation of a weld line and with a single initial  
494 charge to provide a long flow path. CT analysis of the parts shows that the  
495 Direct Bundle Simulation is able to predict a resin rich weld line and accounts  
496 for long fiber orientation constraints. Predicting such manufacturing defects in  
497 SMC compression molding enables the optimization of process parameters and  
498 molds early in the development process.

499 Compared to statistical descriptors of fiber orientation, such as commonly used  
500 second order fiber orientation tensors, the direct simulation approach offers sev-  
501 eral advantages: Regions, where fiber lengths are comparable to local dimensions  
502 of the mold and thus where scale separation does not apply, can be described.  
503 This leads to an improved accuracy of computed fiber orientation data at weld  
504 lines and close to the mold boundaries. The distribution of fiber volume fraction

505 and fiber-matrix separation effects can be simulated, as bundles move indepen-  
506 dent of the matrix material. Flow anisotropy is treated naturally by imposing  
507 opposing forces to the fluid phase and does not rely on a closure approximation  
508 of the fourth order fiber orientation tensor. Additionally, the number of con-  
509 tacts and bundle curvature can be computed and shows good agreement with  
510 analytical estimates or evaluation of CT data.

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### 517 **Contributions**

518 NM developed and implemented the Direct Bundle Simulation approach and  
519 wrote the first draft of the manuscript. LS performed CT scans and recon-  
520 structed the 3D images. LB suggested the mold configuration and performed  
521 molding trials together with NM. AH supervised the work in terms of compos-  
522 ite process knowledge and relevance of the addressed subjects. LK suggested to  
523 include two-way coupling in the Direct Bundle Simulation and supervised the  
524 presented work.

### 525 **Conflict of interest**

526 We wish to confirm that there are no known conflicts of interest associated with  
527 this publication and there has been no significant financial support for this work  
528 that could have influenced its outcome.

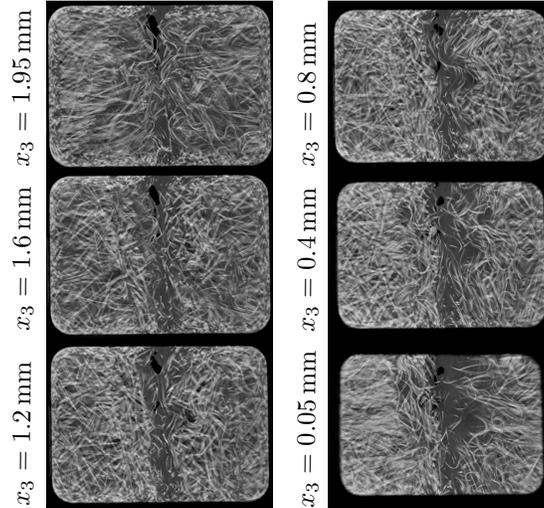


Figure A.14: Equidistant slices through the center area of the CT scan. Bundles are spread close to the mold walls, which can be seen as blurry distribution at  $x_3 = 0.05$  mm and  $x_3 = 1.95$  mm at this resolution. Most bundles in the core stay intact. There is no other pronounced difference between core and shell, which is in agreement with the plug-flow assumption for SMC [3].

## 531 **Appendix B. Contacts**

532 The total number of contacts is evaluated for each frame of the simulation  
 533 results and is plotted in Figure B.15. This averages to approximately  $1.8 \times 10^5$   
 534 contact *pairs* for the split stack configuration and  $2.2 \times 10^5$  contact *pairs* for the  
 535 asymmetric flow, which has a slightly increased fiber volume fraction compared  
 536 to the nominal value. Considering the total amount of 77438 and 87950 bundle  
 537 segments, this evaluates to 4.6 and 5.0 contacts per bundle segment, respectively.

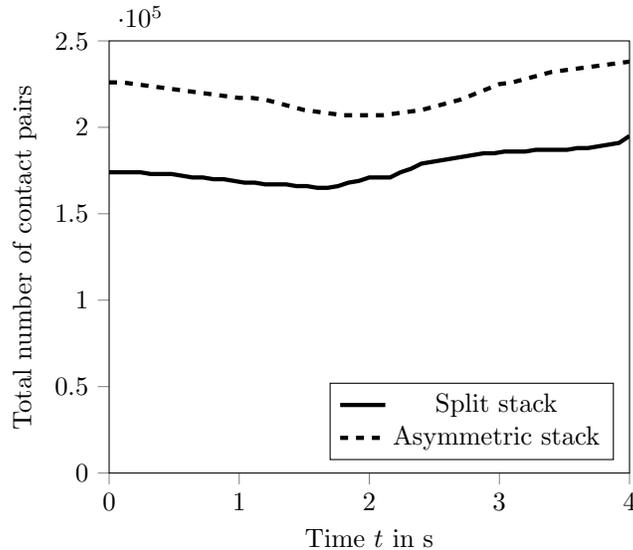


Figure B.15: Number of bundle-bundle contacts pairs during the molding process. The number of contact pairs decreases during the forming phase of the stack and increases during flow, when the entire stack is compressed.

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