

Helium-based cooling concept of the ET-LF interferometer

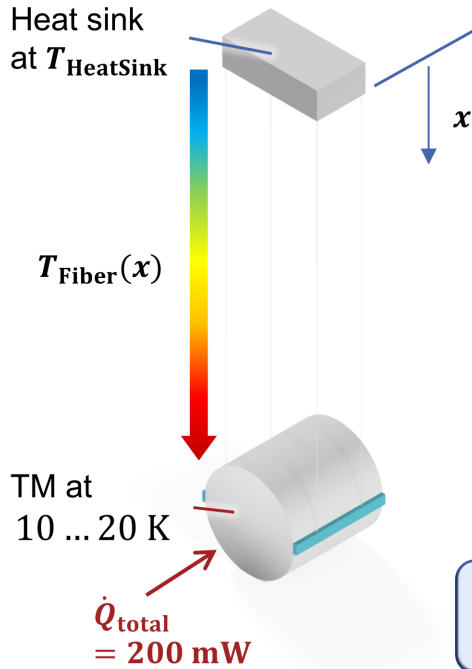
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Gravitational Wave
Advanced Detector Workshop
17-21 May 2021



Test mass temperature limitation

Last-stage suspension scheme:

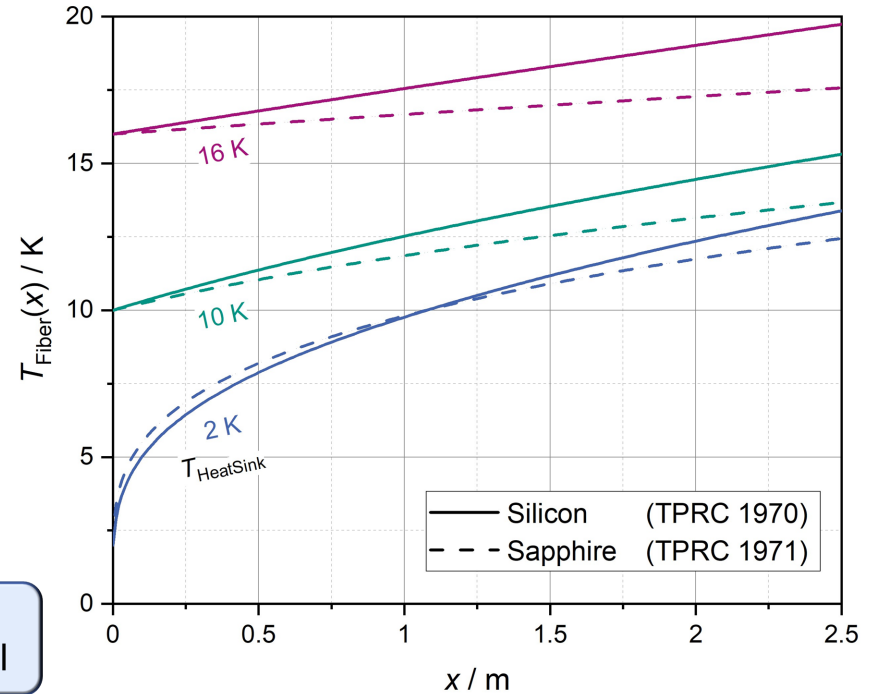


Analytical calculations



$T_{\text{HeatSink}} \rightarrow 2 \text{ K}$ offers
 $T_{\text{Mirror}} \leq 10 \text{ K}$ potential

4x High-purity fibers ($\varnothing = 3 \text{ mm}$)



He-II: payload heat extraction

Two liquid phases of ^4He :

■ He-I (classical liquid helium)

➤ Behaviour: ~ideal gas

----- $T_\lambda(1 \text{ atm}) \approx 2.17 \text{ K}$ -----

$T > T_\lambda$
 $T < T_\lambda$

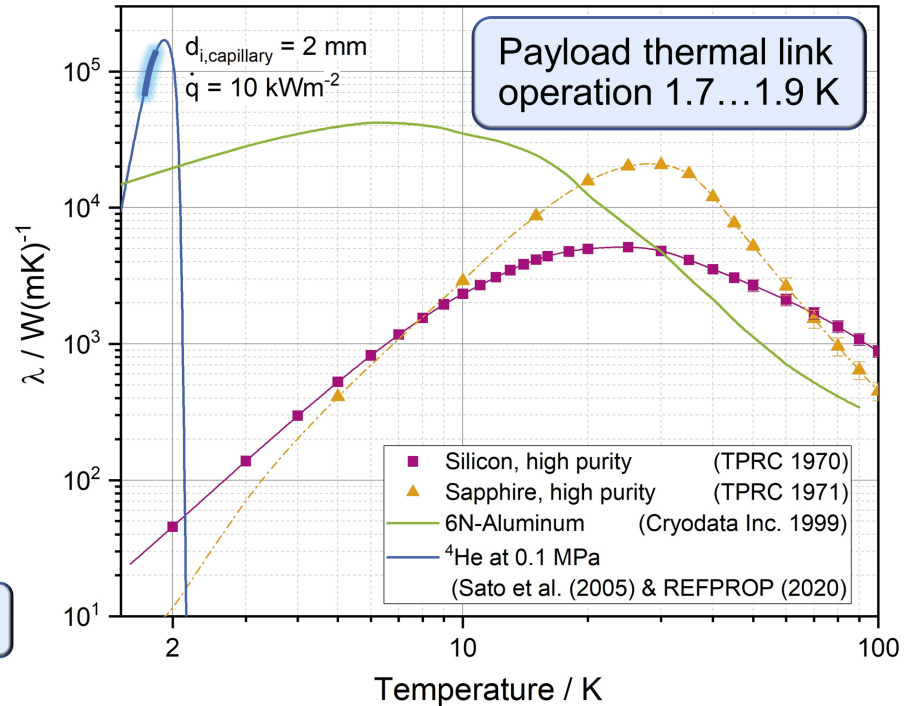
■ He-II (“two fluid model” [1][2])

■ Normal component

■ Superfluid component

➤ Bose-Einstein condensate

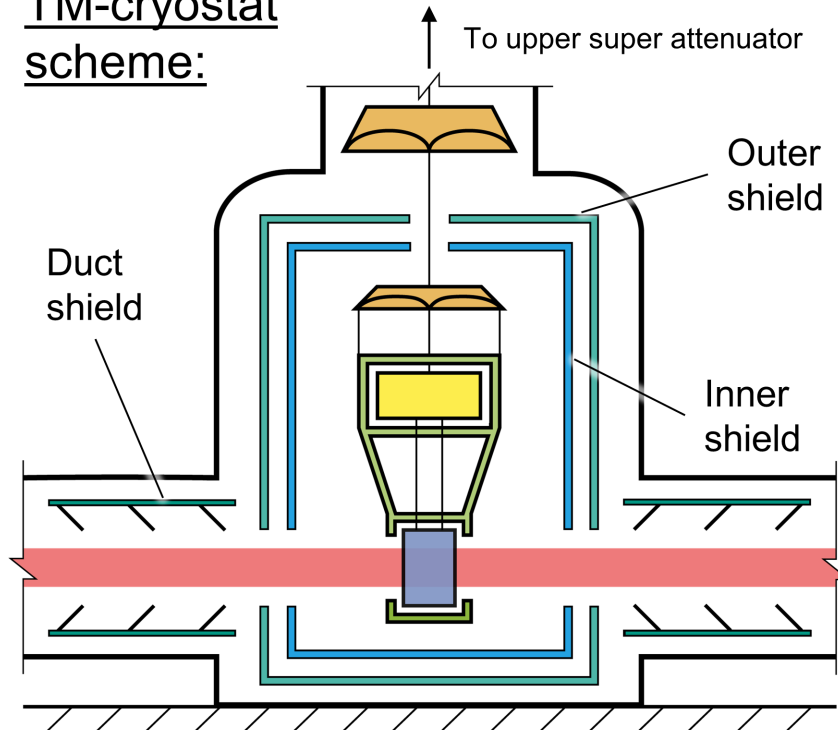
He-II: enhanced heat transfer properties



Sources: [1] Tisza, L. Transport Phenomena in Helium II. Nature 141, 913 (1938).
 [2] Landau, L. Theory of the Superfluidity of Helium II. Phys. Rev. 60, 356-358 (1941).

TM cryostat cooling: temperature levels

TM-cryostat scheme:



Three temperature stages:

Part(s)	Temp. level	Estimated cooling power
Outer thermal shield	50...80 K	$x \dots 10^3$ W
Inner thermal shield	5 K	$x \dots 10^2$ W
Payload heat sink	2 K	$x \dots 1$ W

$$T_{\text{InnerShield}} < T_{\text{Mirror}} (\sim 10 \text{ K}) \text{ possible}$$

Helium-based cooling power provision

Basis: He-refrigerator + subcooler

Example: Linde L-Series

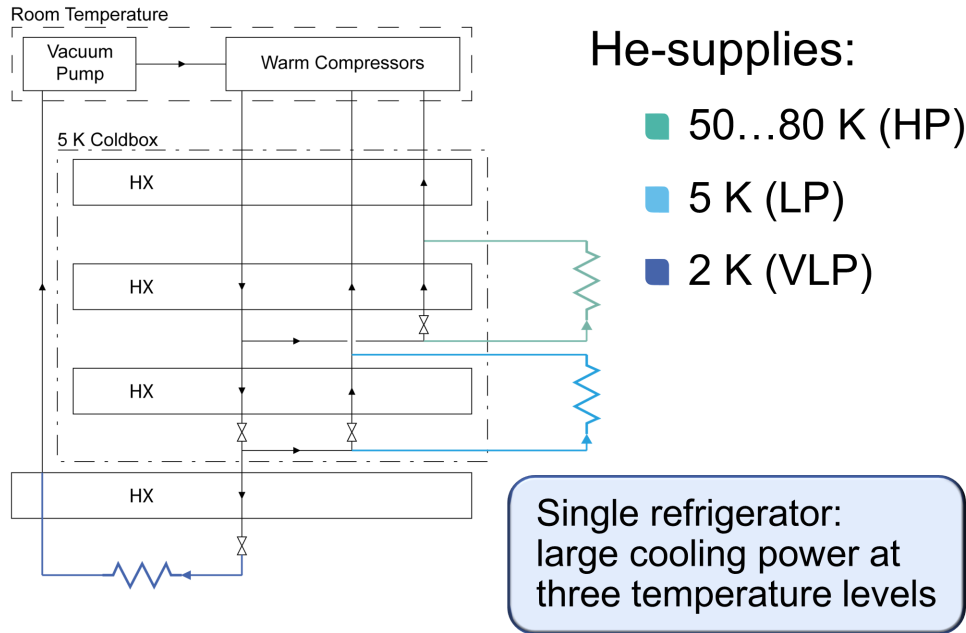
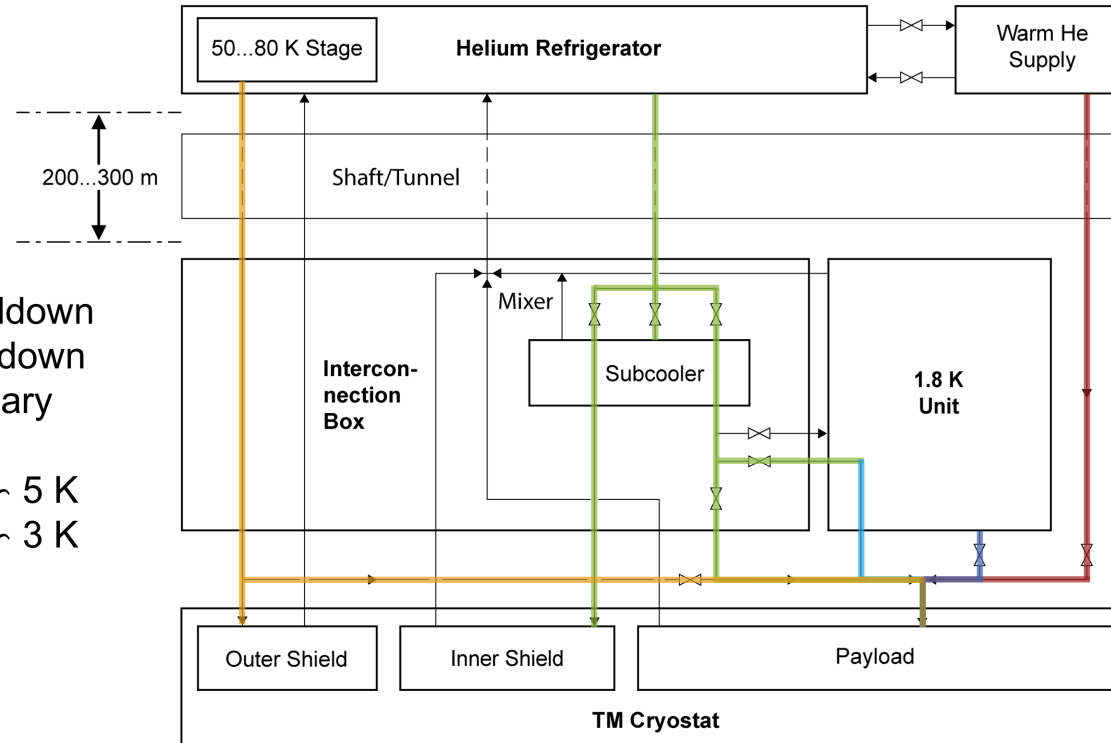


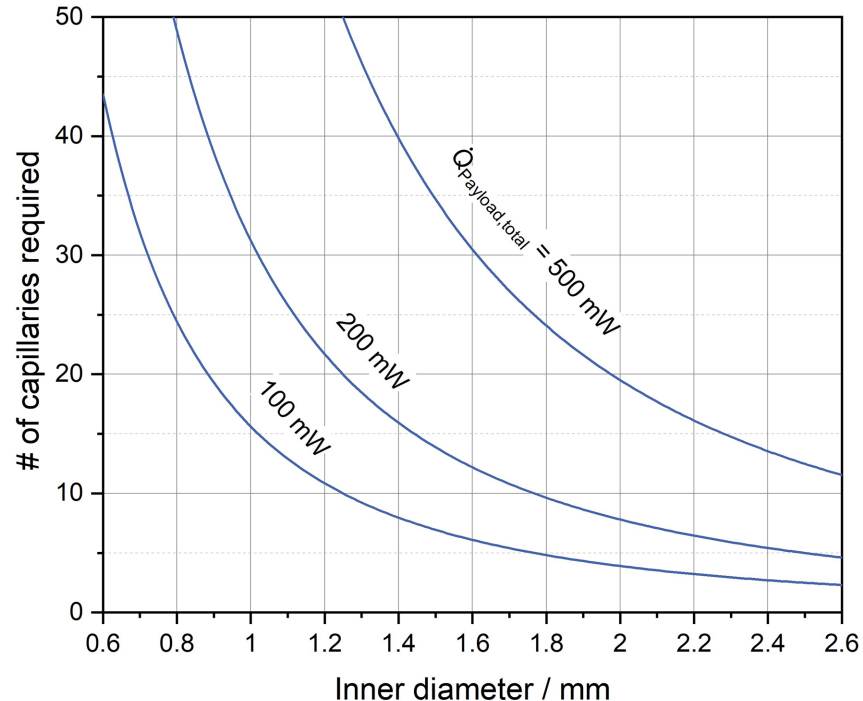
Image: L-Series - Standard Helium Liquefiers / Refrigerators, Linde Kryotechnik AG, 2021.

Basic layout: complete cryostat cooling



1. Outer shield cooldown
2. Inner shield cooldown
3. Payload preliminary cooldown
4. Payload link → ~ 5 K
5. Payload link → ~ 3 K
6. He-II formation
7. Steady-state (no flow)

Payload thermal link: He-II capillaries



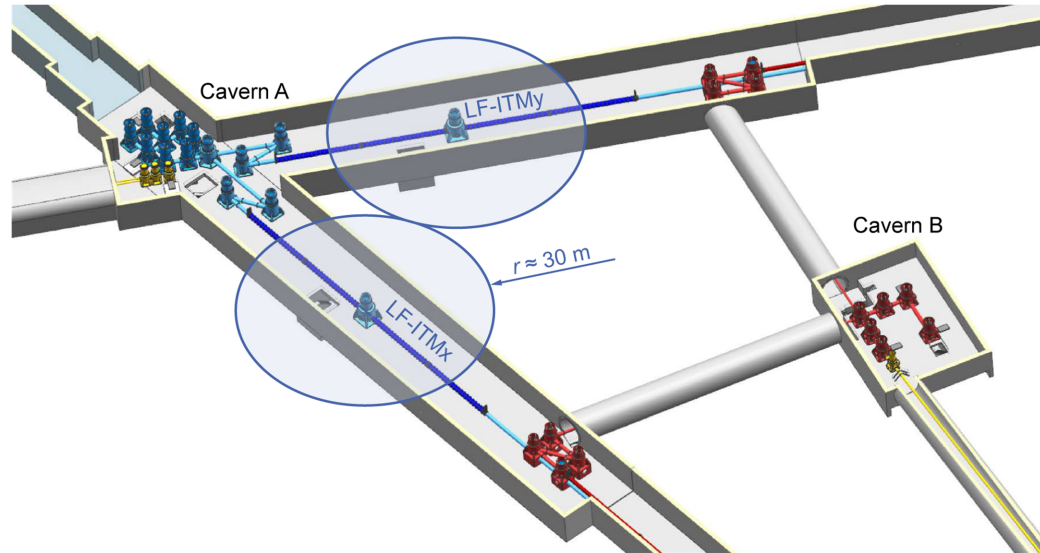
Key boundary conditions:

- Operating pressure: 0.1 MPa
- Capillary length: 30 m
- Capillary cold end temp. (He-II): 1.80 K
- Capillary warm end temp. (He-II): 1.90 K

Dimensioning example (approx.):
 10 capillaries with $d_i = 1.8$ mm can extract 200 mW from a payload at 1.9 K over 30 m distance with a ΔT of only 0.1 K.

1.8 K unit positioning possibilities

Corner cavern scheme:



1.8 K Unit main components:

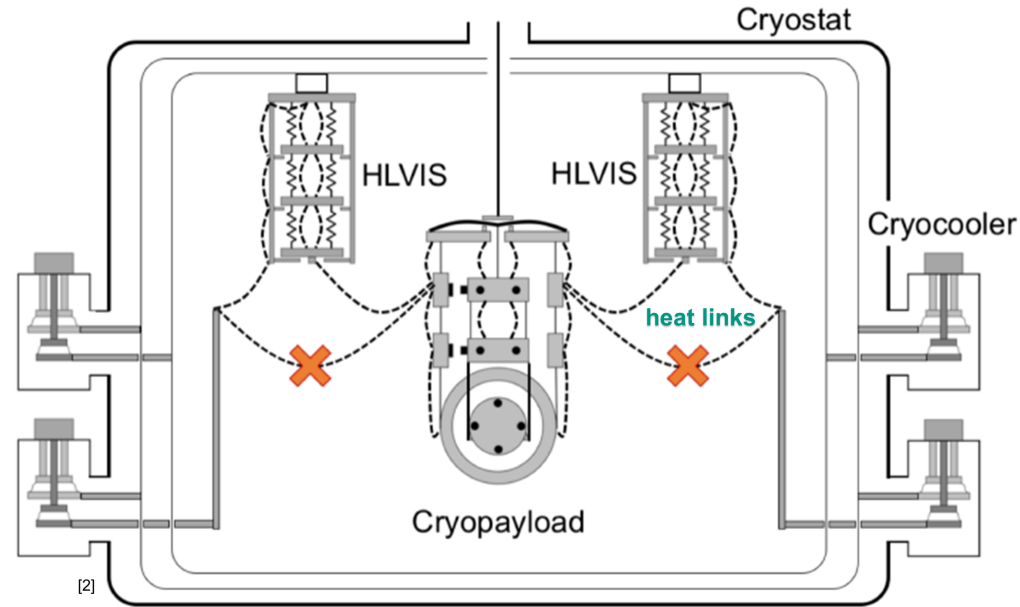
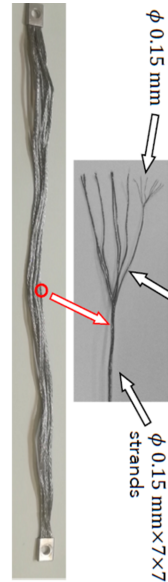
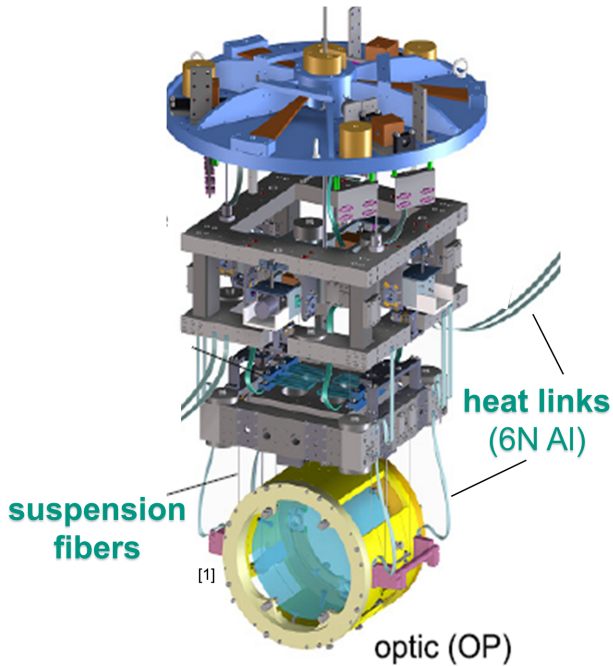
- Vacuum vessel
- Low-pressure heat exchanger
- Vacuum pumps

Long capillaries offer low-noise cooling potential and flexible positioning of the 1.8 K units.

Image: ET Steering Committee Editorial Team. Design Report Update for the Einstein Telescope. Technical report, ET-0007B-20, 2020 (altered)

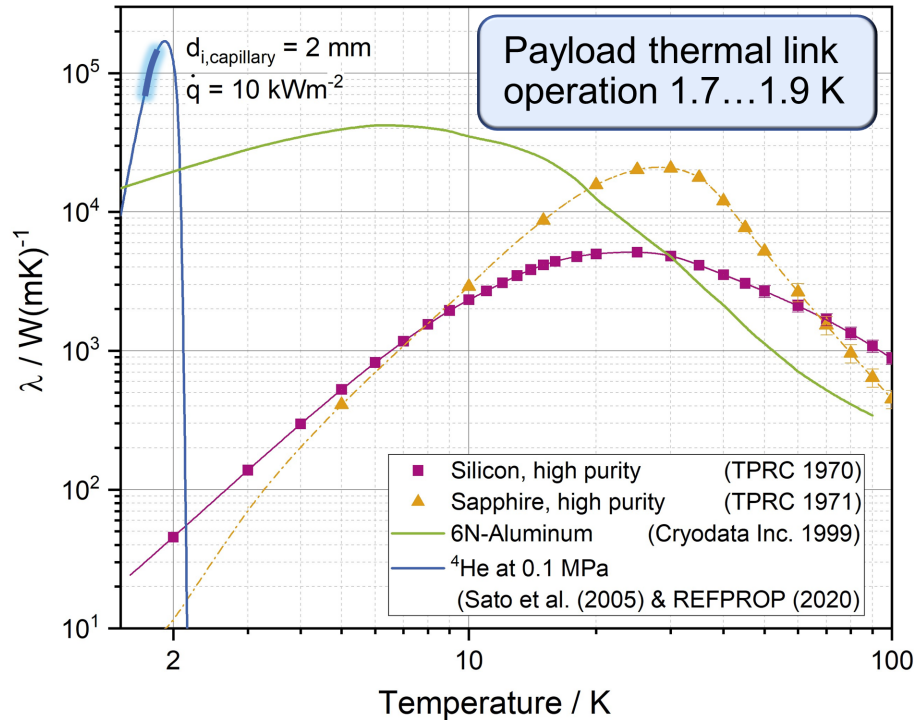
Vibration propagation via He-II hollow heat links

KAGRA experience on vibration transmission into payload



Sources: [1] Akutsu et al (2019) – First cryogenic test operation of underground km-scale GW Observatory KAGRA
 [2] T.Yamada (2020) – Low-Vibration Conductive Cooling of KAGRA Cryogenic Mirror Suspension

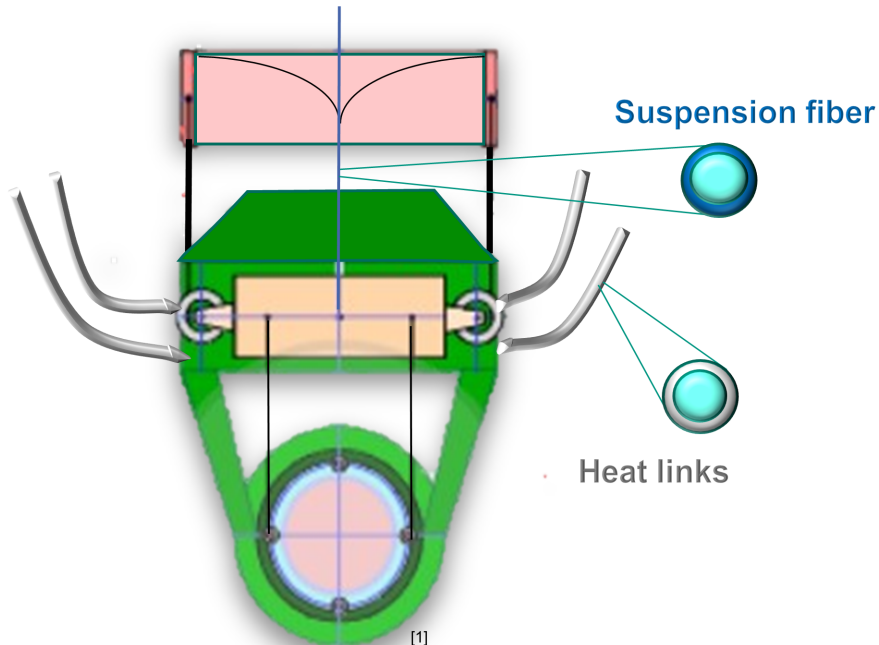
He-II hollow capillaries as heatlinks of ET-LF payload



Lower thermal dissipations

- Cryogenic temperatures
- Dissipation-free superfluid component in He-II

Integration of He-II filled capillary into payload

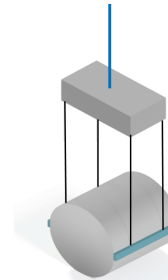


- He-II filled capillary ($\varnothing \leq 3 \text{ mm}$)

- Hollow capillary as **suspension fiber**

- Marionette

- Mirror (?)



- Hollow capillary as external **heat links**

Source: [1] Payload design from P.Rapagnani: ET-LF-Main Features and Constrains (26.04.21)

Theoretical description of thermal dissipation in He-II capillaries

Vibrational Noise into payload

- He-II capillaries → suspension fibers and/or heat links
- Cooling system noise

Experimental Proof of Concept

- Ultra-low noise He-cooling system

Thank you for your attention!

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