An improved Coupled Level Set and Volume of Fluid (i-CLSVoF) framework for sessile droplet evaporation

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A B S T R A C T

Surface-tension-dominant droplet evaporation is ubiquitous and of importance to many applications. We present an improved Coupled Level Set and Volume of Fluid (i-CLSVoF) framework without explicit interface reconstruction for modelling micro-sized droplets with and without evaporation. In the i-CLSVoF framework, an improved surface tension force model with additional filtering steps to filter un-physical spurious velocities is developed and implemented. A simple, yet efficient, velocity-potential based approach is proposed to reconstruct a divergence-free velocity field for the advection of the free surface during droplet evaporation. This approach fixes the numerical issues resulting from the evaporation-induced velocity jump at the interface. The smeared mass source term approach incorporated in this work guarantees greater numerical stability than the non-smeared approach. Three different evaporation models (constant mass flux, thermally driven evaporation and droplet evaporation at room temperature) are implemented in the i-CLSVoF. Corresponding numerical benchmark cases (dam break, droplet relaxation and droplet evaporation subjected to different evaporation models) are conducted to validate the surface tension and the evaporation models. Good agreement between the numerical and corresponding analytical solutions is found. The model developed in this work shows convincing performance in modelling surface-tension-dominant flow with and without evaporation.

1. Introduction

Modelling droplet evaporation is of great importance for many applications, such as inkjet printing [1], spray coating [2], and combus tion of fuel droplets [3]. The key issues in the computational modelling of droplet evaporation are three fold: free surface tracking or capturing, the phase change from liquid to vapour, and accurate calculations of the surface tension force [4]. We begin this paper with a review of previous work to address each issue.

1.1. Free surface tracking/capturing

For interface tracking, the front tracking method is widely used; here, the basic idea is to use so called marker points to identify the interface's location [5,6]. Alternatively, the phase field method identi fies the interface through the value of an order parameter: the order parameter is 1 in the liquid phase, -1 in the gas phase, with a value in between representing the diffuse interface [7]. The Volume of Fluid (VoF) method is another popular interface capturing approach in which the volume fraction field is computed and the interface inferred. The Level Set (LS) method captures the free surface by the signed LS function, with the zero LS at the interface, positive in the liquid and

Corresponding author. *E-mail address:* huihuang.xia@kit.edu (H. Xia). negative in the gas phase. Furthermore, the LS method can guarantee a very sharp interface without interface diffusion [8].

1.2. The surface tension force

The surface tension force plays a significant role in droplet wetting and evaporation [9,10]. The water droplet on a leaf is a simple case to demonstrate the role of the surface tension force in forming a given contact angle between the droplet and the leaf surface. The surface tension force is also crucial to maintain the droplet shape under the influence of gravity and other external forces acting on the droplet [11]. The Continuum Surface tension Force (CSF) model was proposed to model the surface tension force as a volumetric body force [12]. However, this conventional surface tension model suffers from spurious currents or velocities which appear around the interface. Spurious cur rents destabilize the simulations and even influence the internal flow inside the droplets when studying droplets numerically [13]. Spurious currents partially result from numerical errors when calculating the interface curvature. Some numerical models have been developed to improve the calculation of the interface curvature and thus suppress the un physical velocities. The geometric VoF represents the interface

Nomenclature

Acronyms

Greek symbols

Others

by a reconstructed thin interface inside each of the interface cells explicitly and is reported to have better performance in interface representations as well as reducing spurious velocities [14]. Some open source codes or libraries incorporate the geometric VoF, such as PARIS [15], Basilisk [16], isoAdvector [17], interPlicFoam [18] and VoFLibrary [19].

In contrast to the geometric VoF approach, the algebraic VoF method is relatively simple and easy to implement as the interface is represented implicitly and without explicit interface reconstruc tion [20]. The algebraic VoF method is mass conserving but was reported to suffer from some interface diffusion [7]. Several methods have been proposed to address the interface diffusion problem, for instance, incorporation of a surface compression term [21], adaptive interface compression [22], and coupling VoF to some other numerical methods, such as the LS method [23]. Concerning the advantages and shortcomings of both the VoF and LS method, the so called Coupled Level Set and Volume of Fluid (CLSVoF) method was proposed to combine sharp interface representation and mass conservation [24]. The coupled approach improves the suppression of the spurious cur rents. However, according to our experience, relatively large spurious velocities still exist around the interface, especially for micro sized droplets. A short summary of methods used to track/capture the free surface and their applications are outlined in Table 1.

1.3. Phase change from liquid to vapour

Several phase change models have been developed to model the phase change from liquid to vapour. These include the constant mass flux model [29,30], the thermally driven model [31 33], and the vapour mass fraction gradient model [5,29,34]. The challenging part in modelling phase change or evaporation is to address the velocity jump at the interface, which results in some numerical difficulties. Kunkelmann developed an approach that removes the source terms at the interface cells and defines positive and negative mass sources in the most adjacent liquid and gas cells, respectively [35]. This approach was demonstrated to have good performance in modelling boiling [35]. A similar method is also implemented into the open source code Gerris for modelling droplet evaporation subject to a large mass transfer rate [36]. Both methods are highly dependent on the mesh resolution at the inter face. Normally, Adaptive Mesh Refinement (AMR) is needed to cut the interface region into two different regions with negative and positive mass sources accurately. These numerical models are implemented in either in house or commercial codes. Thus, we propose to develop a comprehensive solver in the open source C++ toolbox OpenFOAM to incorporate the simple yet efficient evaporation models to model the micro sized sessile droplet evaporation with negligible influence of un physical spurious currents.

This paper addresses the issues mentioned above related to sup pressing un physical spurious currents and the velocity jump due to phase change at the interface. It presents a simple yet efficient nu merical framework to model the evaporation of micro sized sessile droplets with sharp interface representation and suppressed spurious velocities. The main contribution of the work is coupling algebraic VoF to LS with an improved surface tension force model incorporating filtering steps. (All these contributions are referred to here as the i CLSVoF framework.) Furthermore, improved evaporation models are implemented into i CLSVoF for studying sessile droplet evaporation with three different evaporation models. The model predicts the evap oration rates accurately with less influence of spurious velocities. This work consists of the following main sections: First, the mathematical description of the incompressible Newtonian flow is given in Section 2. The corresponding numerical method and the improved numerical framework i CLSVoF are detailed in Section 3. Section 4 presents the numerical benchmark cases to demonstrate the performance of the numerical framework developed in this work.

Table 1

2. Mathematical formulation

This section presents the equations governing the physics behind incompressible Newtonian flow based on the one field formulation. The one field formulation solves only one set of governing equations for both liquid and gas phases [37]. The indicator function $I(x, t)$

$$
I(\mathbf{x}, t) = \begin{cases} 1 & \mathbf{x} \text{ in liquid,} \\ 0 & \text{otherwise,} \end{cases}
$$
 (1)

is used to identify the tracked phase (liquid phase in this work) at time t and position x. The fundamental fluid quantities such as fluid density ρ and dynamic viscosity μ can be represented in the complete multiphase domain by the indicator function $I(x, t)$ using expressions like

$$
\rho = I(\mathbf{x}, t)\rho_1 + [1 - I(\mathbf{x}, t)]\rho_2, \n\mu = I(\mathbf{x}, t)\mu_1 + [1 - I(\mathbf{x}, t)]\mu_2.
$$
\n(2)

2.1. Governing equations for incompressible flow without phase change

The physics behind the incompressible Newtonian fluid without phase change is governed by the Navier Stokes equations

$$
\nabla \cdot \mathbf{U} = 0,\tag{3}
$$

$$
\frac{\partial(\rho \mathbf{U})}{\partial t} + \nabla \cdot (\rho \mathbf{U} \mathbf{U}) = -\nabla p + \nabla \cdot [\mu (\nabla \mathbf{U} + (\nabla \mathbf{U})^T)] + \rho \mathbf{g} + \mathbf{F}_{\text{st}},\tag{4}
$$

where U is the velocity field. Its divergence being equal to zero means that the flow is incompressible. In the momentum equation (Eq. (4)), p is the pressure field and ρ g the gravity force term. The surface tension force newly introduced in this work is incorporated into the momentum equation as the last term on the RHS (the surface tension force model is detailed in Section 3). The VoF transport equation is solved to capture the free surface, by which the volume fraction field can be continuously updated. The material derivative of the volume fraction field α_l defines the VoF transport equation

$$
\frac{D\alpha_l}{Dt} = \frac{\partial \alpha_l}{\partial t} + \mathbf{U} \cdot \nabla \alpha_l = 0. \tag{5}
$$

In OpenFOAM, the conservative VoF equation

$$
\frac{\partial \alpha_l}{\partial t} + \nabla \cdot (\alpha_l \mathbf{U}) = 0,\tag{6}
$$

is solved where the divergence free condition (Eq. (3)) must be satisfied simultaneously to guarantee the incompressibility. The additional so called interface compression term $\nabla \cdot [\alpha_l(1-\alpha_l)U_r]$ with U_r being the artificial compression velocity is generally incorporated into the LHS of the VoF equation (Eq. (6)) to suppress the interface diffusion [38]. As demonstrated in our benchmark study and the literature, however, the compression term enhances spurious currents [39,40].

2.2. Governing equations for incompressible flow with phase change

The primary governing equations for the incompressible flow with out phase change are modified to account for the phase change from the liquid phase to the vapour phase occurring at the liquid surface. The velocity field for the incompressible flow with phase change is not divergence free anymore. We introduce a volumetric mass source term \dot{m} (mass change rate per unit volume) on the RHS of the continuity equation giving

$$
\nabla \cdot \mathbf{U} = m\left(\frac{1}{\rho_g} - \frac{1}{\rho_l}\right). \tag{7}
$$

Here, *in* means the mass loss of the liquid phase which reappears at the vapour phase with the same amount. Correspondingly, ρ_{ϱ} and ρ_l are the density of the gas and liquid phases, respectively. The momentum equation is the same as for the incompressible flow without phase change.

However, we need to modify the velocity field in the VoF transport equation (Eq. (6)) by the new interface velocity field U_r as

$$
\frac{\partial \alpha_l}{\partial t} + \nabla \cdot (\alpha_l \mathbf{U}_\Gamma) = \alpha_l \nabla \cdot \mathbf{U}_\Gamma.
$$
 (8)

The reason behind that lies in our preliminary numerical simulations which confirmed that using the default one field velocity U to solve the VoF transport equation tends to overestimate the evaporation rate, and we can find the same conclusion in the literature [30].

The calculations of U_{Γ} can be derived from the interface mass flux balance. The interface mass flux per unit area *J* [kg/(m² s)] when the phase changes is derived from the mass flux balance across the interface,

$$
J = \rho_I (\mathbf{U}_e - \mathbf{U}_\Gamma) \cdot \mathbf{n} = \rho_g (\mathbf{U}_g - \mathbf{U}_\Gamma) \cdot \mathbf{n},\tag{9}
$$

where **n** is the interface normal vector, and U_e and U_g are the fluid velocity in the liquid and gas phase, respectively. The interface velocity U_{Γ} can be accordingly derived as

$$
\mathbf{U}_{\Gamma} = \mathbf{U}_e - \frac{J}{\rho_l} \mathbf{n},\tag{10}
$$

where the second term $J/\rho_l \cdot \mathbf{n}$ is the interface recession velocity and accounts for the interface shrinking during the evaporation process. The first term U_e is known as the extended divergence free liquid velocity.

A simple method based on the algebraic VoF is developed in this work to reconstruct the new divergence free velocity field U_e . The basic idea behind the reconstruction of U_e is to solve the velocity potential equation

$$
\begin{cases} a\phi + \nabla^2 \phi = \dot{m}(\frac{1}{\rho_g} - \frac{1}{\rho_l}), \\ U_s = \nabla \phi \end{cases}
$$
 (11)

Fig. 1. The sub-domains for solving the velocity potential equation.

in the whole computational domain with the homogeneous Dirichlet boundary condition applied on the boundary to guarantee that the ve locity potential at the domain boundary is zero. Here, ϕ is the velocity potential, and U_s denotes the evaporation induced Stefan flow velocity (equal to the velocity potential gradient). The critical parameter a is used to divide the whole computation domain into two sub domains (refer to Fig. 1), where a is zero in the liquid phase (blue circle) and within the three most adjacent cells around the interface (the circle with solid line). For the rest of the computational domain, a can be any arbitrary non zero value (the square of the time step size is used in the current study). As the RHS of the velocity potential equation (Eq. (11)) has the same source term as the one of the continuity equation (Eq. (7)) for the incompressible flow with phase change, the new divergence free velocity field U_e is defined by subtracting the evaporation induced Stefan flow velocity \mathbf{U}_s from the one field velocity field U as

$$
\mathbf{U}_e = \mathbf{U} - \mathbf{U}_s. \tag{12}
$$

Typically, the divergence of U_e should approximate 10^{-8} or even smaller values of 10^{-10} , which can be regarded as zero numerically. As an alternative to solve and update the liquid volume fraction field α_l with the implicit source term (as shown in Eq. (8)), the divergence free velocity field \mathbf{U}_e can also be used to advect the free surface with either explicit

$$
\frac{\partial \alpha_l}{\partial t} + \nabla \cdot (\alpha_l \mathbf{U}_e) = -\frac{\dot{m}}{\rho_l} \tag{13}
$$

or implicit

$$
\frac{\partial \alpha_l}{\partial t} + \nabla \cdot (\alpha_l \mathbf{U}_e) = \alpha_l \frac{-\dot{m}}{(\alpha_l + \delta_s) \rho_l} \tag{14}
$$

source term¹ accounting for the mass loss in the liquid phase due to the evaporation where δ_s in Eq. (14) is a small number to guarantee a non zero denominator.

Tracking the temperature distribution around an evaporating droplet is crucial to model droplet evaporation subjected to the ambient temperature gradient. Conservation of thermal energy is given by the temperature equation

$$
\frac{\partial(\rho c_p T)}{\partial t} + \nabla \cdot (\rho c_p \mathbf{U} T) = \nabla \cdot (k \nabla T) - \dot{m} h_{\text{ev}} + \left[\frac{\partial(\rho c_p)}{\partial t} + \nabla \cdot (\rho c_p \mathbf{U}) \right] T, (15)
$$

¹ When solving the VoF equation in OpenFOAM, the implicit source term is recommended [41].

where T is the temperature field, c_p the specific heat capacity, k the thermal conductivity and *h*_{ev} the enthalpy or latent heat of evaporation. The second term on the RHS of the temperature equation is due to the evaporation induced cooling, and the last term couples to the mass source term.

For droplet evaporation at room temperature, the vapour concentra tion gradient around the evaporating droplet drives the phase change from liquid to vapour. Accordingly, the vapour concentration is solved directly to model the phase change from liquid to vapour [42]. To simplify the characterization of the numerical model, in the current work, a dimensionless quantity called the vapour mass fraction field Y is solved and updated by the convection diffusion equation given as

$$
\frac{\partial Y}{\partial t} + \nabla \cdot (YU) = D_v \nabla^2 Y,\tag{16}
$$

where D_v is the vapour diffusivity. The vapour convection diffusion equation is solved on the whole computational domain with the pre scribed so called internal boundary conditions, i.e., all the liquid cells are assigned with the saturation mass fraction (detailed in Section 3). Accordingly, this internal boundary condition guarantees that vapour diffuses only from the liquid surface to the gas domain without un physical diffusion going back to the droplet [43].

3. Numerical method

We solve the aforementioned equations using the Finite Volume Method (FVM). The improved Coupled Level Set and Volume of Fluid (i CLSVoF) framework is proposed in this work to suppress un physical spurious velocities and improve numerical stability, especially when phase changes. The corresponding in house solvers interDyMFoamX (without phase change) and interDyMEvapFoamX (with phase change) are accordingly developed.

3.1. Interface capturing approaches

In the VoF method, the volume fraction field α is defined as the volume averaged volume integral of the phase indicator function $I(x, t)$, denoted as

$$
\alpha = \frac{1}{V} \int_{V} I(\mathbf{x}, t) dV.
$$
 (17)

The basic idea behind the VoF interface capturing approach is to track the evolution of the volume fraction field for a given phase. Typically, the liquid is selected as the tracked phase, and accordingly, the liquid volume fraction field α_l is tracked. As shown in Fig. 2, the computational domain has two sub domains Ω_1 (liquid) and Ω_2 (gas) in the whole computational domain. When the liquid cells are full of liquid, the volume fraction of all liquid cells in Ω_1 is 1. Furthermore, the interface cells are partially filled with the liquid, so that the intermediate value between 0 and 1 are given there. The VoF method is mass conserving, but as mentioned in the literature, the VoF method suffers from some interface diffusion, which can diffuse over several cells around the interface depending on the mesh resolution [44].

The LS method is another interface capturing method, and the quan tity used in LS is the so called signed LS function $\psi(x, t)$. The interface can be identified as zero level set, and in the liquid phase (Ω_1) , we have positive LS function and negative values in the gas phase (Ω_2) [8]. The signed LS function is continuous and has a smooth transition from liquid to gas phase. It is obvious that the LS method can guarantee a more sharp interface compared to the VoF approach, but the LS method is reported to be not mass conserving in the literature [45].

Both the VoF and LS methods suffer from un physical spurious velocities, which destabilize the numerical simulations. Spurious ve locities occur due to inaccurate interface curvature calculations. One method that attempted to improve the curvature calculation was to refine the mesh. However, we found that a finer mesh could not reduce the un physical velocity but even enhanced it (see Fig. 3); indeed, the same conclusion can also be found in the literature [46].

Fig. 2. Interface capturing approaches: (a) the VoF method, (b) the LS method.

Fig. 3. Spurious velocities around the interface (represents with the solid white circle) with two different mesh sizes : (a) coarse mesh (cell number: 40 × 40), (b) fine mesh (cell number: 80×80

3.2. The i CLSVoF framework

In this work, we combine the advantages of VoF and LS by using the so called Coupled LS and VoF (CLSVoF) approach to capture the free surface. The CLSVoF approach improves the mass conserving issues of LS while also guaranteeing a sharp interface [24]. However, although it improves the calculation of the interface curvature [23,45], the conventional CLSVoF, also known as the simple CLSVoF (s CLSVoF) approach, still suffers from un physical spurious velocities. The filtering surface tension model based on the VoF approach is reported to sup press un physical spurious currents, especially for droplets interacting with substrates [27]. Accordingly, we propose the improved CLSVoF (i CLSVoF) method to suppress the spurious currents and improve the numerical stability by extending the filtering method to filter un physical spurious velocities further. In contrast to the s CLSVoF method, our i CLSVoF framework incorporates an improved surface tension force model to calculate surface tension forces more accurately. This framework filters and reduces spurious velocities by additional filtering steps, which are discussed further in the following part.

The basic idea behind the i CLSVoF framework is that initializing the initial signed distance function ψ_0 from the liquid volume fraction field α_l . ψ_0 is initialized by the initialization function

$$
\psi_0 = (2\alpha_l - 1)\Gamma \tag{18}
$$

with the dimensionless quantity $\Gamma = 0.75 \Delta x$, where Δx is the minimum mesh size around the interface. Δx is dimensionless as well because

 Δx is artificially divided by a dimensional quantity with a value of 1 and the dimension of metre. Normally, the LS function gradually loses its property to have a value of zero at the interface and cannot be sufficiently sharp after moving with the convection velocity. There fore, a re initialization step is adopted to recover its sharpness. The Hamilton Jacobi equation

$$
\frac{\partial \psi}{\partial \tau} - S(\psi_0)(1 - |\nabla \psi|) = 0,\tag{19}
$$

is used to re initialize the LS function with the initial condition $\psi =$ ψ_0 [8]. Here, τ is an artificial time step, and the smoothed out sign function $S(\psi_0)$ is defined as

$$
S(\psi_0) = \frac{\psi_0}{\sqrt{\psi_0^2 + 4x^2}}.
$$
\n(20)

Similar to the conclusion in [25], our benchmarking case study showed that the smoothed out sign function $S(\psi_0)$ can further reduce the un physical velocity and guarantee more numerical stability than the conventional sign function. The Hamilton Jacobi equation needs to be solved with continuous numerical iterations until $|\nabla \psi| = 1$, and the iteration time N_{iter} around 15 can be enough [45]. The re initialization scheme is outlined in Fig. 4.

Before introducing the new surface tension force model imple mented in the i CLSVoF framework, two widely used surface tension models are also listed below for completeness. The Continuum Surface tension Force (CSF) model approximates the surface tension with the

Fig. 4. Flowchart for solving the Hamilton–Jacobi equation.

help of the α_l gradient [9,12]. This surface tension force model is given by

$$
\sigma K(\alpha_l) \nabla \alpha_l, \tag{21}
$$

where σ is the surface tension coefficient, and $K(\alpha_l)$ the interface mean curvature defined as

$$
K(\alpha_l) = -\nabla \cdot \frac{\nabla \alpha_l}{|\nabla \alpha_l| + \delta_n}.
$$
\n(22)

Here, δ_n is a stabilization factor that is used to guarantee a non zero denominator, and δ_n can be calculated by

$$
\delta_n = \frac{10^{-8}}{\left(\sum_{i=1}^N V_i/N\right)^{\frac{1}{3}}},\tag{23}
$$

where N is the number of the cells in the computational domain, and V_i is the volume of the *i*th cell.

As the liquid volume fraction field α_l is not continuous, the cal culation of its gradient is another source of numerical error. The LS method uses the signed LS function ψ to calculate the interface curvature, which is more accurate as ψ ensures continuity along the interface normal [8]. The interface curvature with the LS method can be calculated as

$$
K(\psi) = -\nabla \cdot \frac{\nabla \psi}{|\nabla \psi| + \delta_n}.
$$
\n(24)

The improved curvature calculation method is incorporated into the conventional CLSVoF approach to improve the surface tension calcula tions $[23,45]$. As an alternative to Eq. (21) , the surface tension force is then given as

$$
\sigma K(\psi)\delta_{\psi}\nabla\psi,\tag{25}
$$

where the smoothed delta function δ_{ψ} is given by

$$
\delta_{\psi} = \begin{cases} \frac{1}{2\epsilon} \left(1 + \cos(\frac{\pi \psi}{\epsilon}) \right) & |\psi| < \epsilon, \\ 0 & \text{otherwise,} \end{cases}
$$
 (26)

and ϵ is the interface thickness; usually taken as $\epsilon = 1.5 \Delta x$ [47]. As discussed in the literature, ϵ can also range from 1.0 Δx to 1.5 Δx depending on the mesh type [48].

The conventional CLSVoF approach is reported to suppress un physical spurious currents [45]; however, some further work can be done to refine the model and reduce the un physical velocities (some improvements are detailed below). Therefore, the non symmetrical Heaviside function H_ψ is incorporated into our new surface tension force model. The reason is that the non symmetrical Heaviside func tion is reported to improve the numerical stability compared to the symmetrical Heaviside function H_s [28,45]. The symmetrical Heaviside function H_s is obtained by smoothing out the Heaviside step function (purple line in Fig. 5) and denoted as

$$
H_s = \begin{cases} 0 & \text{if } \psi < -\epsilon, \\ \frac{1}{2} [1 + \frac{\psi}{\epsilon} + \frac{1}{\pi} \sin(\frac{\pi \psi}{\epsilon})] & |\psi| \le \epsilon, \\ 1 & \text{if } \psi > \epsilon. \end{cases} \tag{27}
$$

Fig. 5. Three different Heaviside functions (H : the Heaviside step function, H_s : the symmetrical Heaviside function, H_{ψ} : the non-symmetrical Heaviside function).

The non symmetrical Heaviside function H_{ψ} (yellow dotted line in Fig. 5) is given by

$$
H_{\psi} = \begin{cases} 0 & \text{if } \psi < -\epsilon, \\ \frac{1}{2} \left[\frac{1}{2} + \frac{\psi}{\epsilon} + \frac{\psi^2}{2\epsilon^2} - \frac{1}{4\pi^2} \left(\cos(\frac{2\pi\psi}{\epsilon}) - 1 \right) + \frac{\epsilon + \psi}{\epsilon \pi} \sin(\frac{\pi\psi}{\epsilon}) \right] & |\psi| \le \epsilon, \\ 1 & \text{if } \psi > \epsilon. \end{cases} \tag{28}
$$

The difference among the three different Heaviside functions is shown in Fig. 5.

The new and improved surface tension force model developed for our i CLSVoF framework is given accordingly as

$$
\mathbf{F}_{\text{st}} = \sigma K(\psi) \nabla H_{\psi}.
$$
 (29)

As inspired by the VoF based surface tension model, the surface tension related pressure term p_{st} is separated from the total pressure to avoid difficulties in the discretization of the pressure jump [27]. The pressure equation is given by

$$
\nabla \cdot \nabla p_{\rm st} = \nabla \cdot \mathbf{F}_{\rm st},\tag{30}
$$

and the pressure equation can be solved with the prescribed boundary condition

$$
\frac{\partial p_{\text{st}}}{\partial n} = 0. \tag{31}
$$

To filter spurious currents, the modified indicator function $\alpha_{\rm pc}$ is defined for calculating the new Delta function given as Eq. (35), and $\alpha_{\rm pc}$ is calculated with

$$
\alpha_{\rm pc} = \frac{1}{1 - C_{\rm pc}} \left[\min \left(\max(\alpha_l, \frac{C_{\rm pc}}{2}), 1 - \frac{C_{\rm pc}}{2} \right) - \frac{C_{\rm pc}}{2} \right],\tag{32}
$$

where C_{pc} is the sharpening coefficient. C_{pc} equal to 0 yields the original indicator function α_l , which is the liquid volume field and defined by Eq. (17). Increasing C_{pc} leads to a sharp representation of the interface and can suppress the spurious velocity but also brings numerical instabilities [27].

Finally, employing p_{st} and α_{nc} , the filtering surface tension force model is introduced to filter un physical spurious currents parallel to the free surface and is defined as

$$
F_{\text{st},f}^f = F_{\text{st},f} - F_{\text{st},f}^{\text{filt}}.
$$
\n(33)

Here, $F_{\text{st.f}}$ is the surface tension force calculated at face centre by $F_{\text{st},f} = \mathbf{F}_{\text{st}} \mathbf{n}_f$ with \mathbf{n}_f being the normal vector defined at face centre, and $F_{\text{st},f}^{\text{filt}}$ is a time related term also defined at the face centre. It is calculated from

$$
F_{\text{st},f}^{\text{filt}} = \frac{\delta_{\text{st}}}{|\delta_{\text{st}}| + \delta_n} \left(R_f (F_{\text{st},f}^{\text{filt}})_{i-1} + C_{\text{fc}} \left\langle \nabla p_{\text{st}} - (\nabla p_{\text{st}} \cdot \mathbf{n}_s) \mathbf{n}_s \right\rangle_f \cdot \mathbf{n}_f \right), \tag{34}
$$

where R_f is a relaxation factor, and $(F_{st,f}^{filt})_{i-1}$ the value of $F_{st,f}^{filt}$ in the previous time step, and $\langle \rangle_f$ denotes the interpolation from cell centre to cell face in OpenFOAM. C_{fc} is the coefficient determining how fast the spurious velocity is filtered, and \mathbf{n}_s is the normal vector defined at cell centre ($\mathbf{n}_s = \nabla \alpha_l / |\nabla \alpha_l|$). δ_{st} is a newly defined Delta function based on the previously introduced sharpening indicator function $\alpha_{\rm nc}$, and its definition is

$$
\delta_{\rm st} = \nabla_f^T \alpha_{\rm pc}.\tag{35}
$$

Here, ∇_f^{Γ} denotes the gradient normal to the interface.

The final step is to define the threshold for filtering the surface tension flux, also called the capillary flux ($\phi_{cf} = F_{st,f} |S_f|$ with $|S_f|$ being the magnitude of face area). We artificially set the capillary flux as zero when the capillary flux is smaller than the threshold, where the filtering capillary flux is defined as

$$
\phi_{\rm cf}^{\rm filt} = \phi_{\rm cf} - \min\left(\max(\phi_{\rm cf}, -\phi_{\rm cf}^{\text{thre}}), \phi_{\rm cf}^{\text{thre}}\right). \tag{36}
$$

Here, the ϕ_{cf}^{thre} is the threshold value below which the capillary flux is regarded as zero, and it can be calculated by

$$
\phi_{\rm cf}^{\rm thre} = C_{\rm filt} |F_{\rm st,f}||S_f|,\tag{37}
$$

where C_{filt} is the filtering coefficient. It is normally set as 0.01, which means that the capillary flux can be regarded as zero when its mag nitude is less than 1% of the average capillary flux. $|F_{st,f}^-|$ is the magnitude of the average surface tension force.

3.3. The i CLSVoF framework with evaporation

Modelling micro sized droplet evaporation suffers from un physical velocities, and the i CLSVoF framework proposed for suppressing un physical velocities employs an extension to model droplet evaporation. In this section, three different evaporation models are introduced. These models share the same governing equations with the exception of the calculation of the source term per unit area J .

We start with the most simple evaporation model, where the only parameter needing to be defined is the constant mass flux per unit area J . In contrast to calculating the mass flux J with complex equations, for instance, by temperature difference (Eq. (38)), a given constant, e.g. $J = 1.25 \times 10^{-2}$ kg/(m² s) is specified in this work. The constant mass flux evaporation model can be used to validate the implementations of the governing equations before implementing complex approaches to conduct calculations of the source terms.

The thermally driven evaporation drives the phase change from liquid to vapour when the temperature around the liquid interface is higher than its saturation temperature T_{sat} , and the mass flux depends on the temperature difference at the interface [31,49]. The mass flux J is given by

$$
J = \frac{T - T_{\text{sat}}}{R_{\text{int}} h_{\text{ev}}},\tag{38}
$$

Fig. 6. The evaporation coefficient χ_e calculated by Eq. (39) versus the density ratio λ in the range up to 1000.

where h_{ev} is the latent heat of evaporation and R_{int} the heat resistance of the liquid–vapour interface. R_{int} is calculated by

$$
R_{\rm int} = \frac{2 - \chi_e}{2\chi_e} \frac{\sqrt{2\pi R_{\rm gas}}}{h_{\rm ev}^2} \frac{T_{\rm sat}^{3/2}}{\rho_g},\tag{39}
$$

where R_{gas} is the gas constant and χ_e the evaporation coefficient which depends on the density ratio $\lambda = \rho_l/\rho_g$ between the liquid and gas phase [50]. The density ratio dependent evaporation coefficient

$$
\chi_e = \left\{ 1 - \left(\frac{1}{\lambda}\right)^{\frac{1}{3}} \right\} \exp\left(-\frac{1}{2\lambda^{\frac{1}{3}} - 2}\right),\tag{40}
$$

is adopted in this work instead of determining χ_e empirically [51]. The evaporation coefficient χ_e increases gradually with the density ratio λ and tends to reach a plateau at high density ratios (shown in Fig. 6).

The third model implemented in this work is to model droplet evaporation at room temperature. The mass flux per unit area J can be calculated by the mass balance across the interface for a single component liquid as $J(1 - Y^{\Gamma}) = \rho_g D_v \nabla_{\Gamma} Y \cdot \mathbf{n}^{\Gamma}$, which leads to the formula

$$
J = \frac{\rho_g D_v \nabla^T Y \cdot \mathbf{n}^T}{1 - Y^T}
$$
\n(41)

for the mass flux per unit area, where \mathbf{n}^T is the unit interface normal and ∇^{Γ} denotes the gradient at the interface. Furthermore, Y^{Γ} is the saturation vapour mass fraction, and it is given by

$$
Y^{\Gamma} = \frac{X_v M_l}{(1 - X_v) M_g + X_v M_l},\tag{42}
$$

where X_v is further given by the Clausius Clapeyron relation

$$
X_v = \exp\left[\frac{-h_{\text{ev}}M_l}{R_{\text{gas}}}\left(\frac{1}{T} - \frac{1}{T_{\text{sat}}}\right)\right]
$$
(43)

with M_l and M_g being the molar mass of liquid and gas respec tively [34].

The mass source term per unit volume *in* can be calculated by

$$
\dot{m} = J|\nabla \alpha_l| \tag{44}
$$

once the mass source term per unit area J is calculated. Normally, the mass source term \dot{m} is only non zero at a thin layer around the droplet interface (see Fig. 7(a)), and our preliminary numerical study showed that it leads to numerical instability, especially for evaporation with large mass flux.

Our improved solution in this work is to extend the distribution of *̇* to a wide band by smearing *̇* over several adjacent cells near

Fig. 7. The mass source distributions for two different cases: (a) without smearing, (b) with smearing.

Fig. 8. Evolution of the numerical residual for two different cases with and without smearing the mass source term.

the interface (see Fig. $7(b)$). The basic idea is to solve the Helmholtz equation

$$
\dot{m}_s = \dot{m} + (\Delta x N)^2 \Delta \dot{m}_s \tag{45}
$$

with the homogeneous Neumann boundary conditions [31]. Here, \dot{m} , is the smeared mass source term, Δx the minimal mesh size and N is the number of cells over which the mass source term is smeared along the interface normal. Employing smearing of the mass source term, the numerical residual for simple 2D static droplet evaporation cases have been recorded and it turned out that the smeared approach can guarantee a smaller numerical residual (as can be seen in Fig. 8).

3.4. The semi discretized form of the equations

The momentum equation after numerical discretization and lin earization with the FVM in OpenFOAM is outlined in this section. The detailed derivations and numerical issues can be found in the literature [52]. The velocity field can be predicted by

$$
\mathbf{U} = \mathbf{U}^{\star} - \frac{1}{A_p} \nabla p_{\text{rgb}},\tag{46}
$$

where $p_{\text{rgh}} = p - \rho \mathbf{g} \cdot \mathbf{h}$ is the modified pressure field with **h** being the position vector. U^* is given as

$$
\mathbf{U}^{\star} = \frac{\mathbf{H}(\mathbf{u}) + \mathbf{F}_b}{A_p},\tag{47}
$$

where A_p is a scalar field corresponding to the linear algebraic equa tions after discretizing the momentum equation. $H(u)$ accounts for the residual that is left after extracting the diagonal from the coefficient matrix except pressure gradient and body forces, and \mathbf{F}_b is defined by $\mathbf{F}_b = \rho \mathbf{g} + \mathbf{F}_{\text{st}} - \nabla p_{\text{st}}.$

The pressure equation is then derived by substituting Eq. (46) into the continuity Eq. (7), which yields

$$
\nabla \cdot \left(\frac{1}{\langle A_p \rangle_f} \nabla p_{\text{rgb}}\right) = \nabla \cdot \mathbf{U}_f^{\star} - \dot{m} \left(\frac{1}{\rho_g} - \frac{1}{\rho_l}\right),\tag{48}
$$

where \mathbf{U}_f^{\star} is the predicted velocity on the cell face. It is defined as

$$
\mathbf{U}_{f}^{\star} = \frac{\langle \mathbf{H}(\mathbf{u}) \rangle_{f} + \langle \mathbf{F}_{b} \rangle_{f}}{\langle A_{p} \rangle_{f}},\tag{49}
$$

where the last term on the RHS of Eq. (48) accounts for the phase change from liquid to vapour, and this term vanishes for the incom pressible flow without phase change. The face flux ϕ is then corrected as

$$
\phi = \mathbf{U}_f^{\star} \cdot \mathbf{S}_f - \frac{|S_f|}{\langle A_p \rangle_f} \nabla_f^T p_{\text{rgb}},\tag{50}
$$

where S_f is the surface vector and then the corresponding corrected velocity field is reconstructed from the corrected face flux ϕ .

3.5. Time step constraint for stable simulations

The minimum time step for solving the governing equation and ensuring spurious currents do not enhance over time is estimated from two constraints. The first constraint is

$$
A_{t\sigma} < \sqrt{\frac{\rho_{\text{avg}} \Delta x^3}{2\pi\sigma}},\tag{51}
$$

where ρ_{avg} is the average density of the phases. It is proposed for the explicit treatment of the surface tension force term [12]. Another more comprehensive time step constraint is given by

$$
\Delta_{tc} < \frac{1}{2} \left(C_2 \tau_{\mu} + \sqrt{(C_2 \tau_{\mu})^2 + 4C_1 \tau_{\rho}^2} \right),\tag{52}
$$

which involves the density and the viscosity of the multiphase system. which involves the density and the viscosity of the multipliase system.
 τ_{μ} and τ_{ρ} are given as $\mu_{avg}\Delta x/\sigma$ and $\sqrt{\rho_{avg}\Delta x^3/\sigma}$ respectively with μ_{avg} being the average dynamic viscosity of the phases [53]. Accordingly, the minimum time step size for stable numerical simulations is given as

$$
\Delta_t < \min(\Delta_{t\sigma}, \Delta_{tc}) C_{\Delta t} \tag{53}
$$

with $C_{\Delta t}$ being the stabilization factor. A range of $C_{\Delta t}$ between 0.3 and 0.7 is recommended for more stable constraints, especially for cases with phase change [29].

Fig. 9. Flowchart of the i-CLSVoF framework with evaporation.

3.6. The overall solution procedure

The overall solution procedure is outlined in Fig. 9. After initializing the essential fields, such as liquid volume fraction, pressure, velocity, etc., the free surface is advected by solving the VoF equation and then the LS field can also be calculated. Accordingly, the interface curvature and the surface tension can be updated then. To calculate the mass flux, the temperature distribution or the vapour mass fraction needs to be updated by solving the corresponding equations. The velocity and the pressure fields are calculated by solving the continuity and the pressure correction equations. The new divergence free velocity and the interface velocity are reconstructed then. The reconstructed velocity field is used for solving the VoF equation in the next cycle until the pre defined total simulation time t_{end} is reached.

4. Results and discussion

Representing a sharp interface by the i CLSVoF framework is demonstrated by the well known 2D dam break benchmark case in Ap pendix. Numerical benchmark cases are conducted to validate the i CLSVoF framework proposed in this work. We first study the 2D droplet cases in static equilibrium to compare the suppression of un physical spurious velocities with three different surface tension force models (the VoF based surface tension force (Eq. (21)), the CLSVoF based surface tension force (Eq. (25)) and the i CLSVoF based surface tension force (Eq. (29))). Then, validations and discussion regarding the evaporation models implemented in the i CLSVoF framework are presented.

4.1. Suppression of un physical spurious velocities

The simple yet widely used case to demonstrate the suppression of spurious velocities is to study the surface tension induced relaxation of

Fig. 10. Evolution of the un-physical velocities with three different surface tension force models.

a 2D quadratical droplet (density: 1000 kg/m^3 , viscosity: 10^{-3} Pa s) im mersed in a base fluid (density: 1000 kg/m^3 , viscosity: 10^{-3} Pa s) [27]. A constant surface tension coefficient is assumed as 0.07 N∕m. The initial configuration is a 2D square droplet (side length: 40 μm) sitting at the centre of a square 2D computational domain (side length: 100 μm). Gravity is absent, and the surface tension force is the only external force acting on the droplet. Accordingly, the surface tension gradually deforms the droplet from its initial square shape to its equilibrium shape, i.e. a 2D circle.

The maximum velocity of the system is recorded, and we compare the numerical results of our i CLSVoF framework to simulation results with the conventional VoF and CLSVoF methods. For the simulation with the i CLSVoF framework shown in Fig. 10, a sharpening coefficient of 0.95 was adopted. In our manuscript, Fig. 13 shows the effect of the sharpening coefficient on the evolution of the spurious velocity. The best suppression of spurious velocity was found when the sharpening coefficient was 0.8. As will be discussed in the following contents, it may lead to numerical instability issues when the sharpening coefficient is too large. Accordingly, we adopted a very large coefficient, namely, 0.95, to test the stability of our model. As an alternative to track the maximum velocity $|U_{max}|$, the dimensionless Capillary number Ca = $\frac{\mu|U_{\text{max}}|}{\sigma}$ is used to quantify the evolution of spurious velocities in the literature [54]. The convergence of the Capillary number with our i CSLVoF framework is also promising as the ratio between liquid dynamic viscosity $\mu = 10^{-3}$ and surface tension coefficient $\sigma = 0.07$ is smaller than unity in our benchmark cases. As shown in Fig. 10, the velocity with the i CLSVoF framework decreases to 10−¹⁰ m∕s at time $t = 0.001$ s. Thus, the Capillary number decreases to 0.014×10^{-10} m/s at time $t = 0.001$ s. The evolution of the Capillary number is not presented in this section because the evolution of the Capillary number and the velocity have the same tendency. Concerning the total simulation time for droplet relaxation, 0.001 s is enough to guarantee that the maximum velocity converges to zero numerically (less than 1*.*0 × 10−8 in our model). As shown in Fig. 10, the conventional CLSVoF approach can improve the suppression of un physical spurious velocity better than the VoF approach; however, the un physical velocities are still too large to guarantee that the influence of un physical spurious velocity on numerical stability is minimized. The dashed line represents the convergence of the velocity within one millisecond by the i CLSVoF method, and the velocity converges to 10^{-10} m/s which is small enough to eliminate the influence of un physical spurious currents on the numerical stabilities. In principle, as can also be seen from Fig. 10, the velocity for the droplet relaxation case oscillates and decreases gradually [6,55]. This velocity finally reaches a plateau which can be regarded as converging to a constant velocity. For our simulation

Fig. 11. Velocity vector contours at 0.001 s with three different surface tension models (white circle represents the 0.5 iso-surface for α_{l} field): (a) VoF, (b) CLSVoF, (c) i-CLSVoF.

results shown in Fig. 10, the velocities with the VoF and CLSVoF methods are still large at time $t = 0.001$ s, while the velocity with the i CSLVoF method decreases to a very small value at time $t = 0.001$ s. The velocity with the i CLSVoF method will decrease continuously if we conduct our simulation with a longer simulation time. Accordingly, the simulation time of 0.001 s is large enough for evaluating the spurious velocities. It is sufficiently large for the velocity to converge to zero numerically, as discussed in the literature [22,27].

The corresponding velocity vector contours at 0*.*001 s with three different surface tension models the VoF based surface tension force (Eq. (21)), the CLSVoF based surface tension force (Eq. (25)) and the i CLSVoF based surface tension force (Eq. (29)) are given in Fig. 11, and the white circles represent the 0.5 iso surface for the liquid volume fraction field α_l . The distributions of spurious vortices for the three different surface tension models are different. For the normal VoF approach, four large spurious vortices appear around the free surface and point toward four different directions, which lead to the strong spurious velocities that deform the free surface of the droplet and then move the droplet randomly away from its centre. Concerning the spurious vortices of the CLSVoF approach, we can also see four main vortices pointing inward but the overall distribution is symmetrical along with the horizontal and vertical directions. For the simulation with the i CLSVoF method, the maximum velocity is located inside the droplet, however, no large spurious vertices are found around the free surface. The velocity distribution is symmetrical along with the diagonal of the computational domain, and the magnitude of the maximum velocity is tiny enough to avoid the influence of spurious currents on the numerical stability.

The analytical solution of the capillary pressure jump across droplets is given by the Young Laplace equation, and $\Delta p_c = \frac{\sigma}{R}$ $\frac{0.07 \text{ N/m}}{20.07 \text{ N/m}}$ = 3101*.8* Pa is the theoretical solution for 2D droplets (where R is the droplet radius) [54]. The relative error for the capillary

pressure jump is given by

$$
E(\Delta p_c) = \frac{|p_c^n - p_c^n|}{p_c^n},
$$
\n(54)

where p_c^a and p_c^n are the analytical and numerical capillary pressure jump, respectively. The comparison among three different surface tension force models in predicting the capillary pressure is shown in Fig. 12(a). The i CLSVoF developed in this work demonstrates the best agreement between numerical prediction and the analytical solution. The quantitative study on the relative error in predicting capillary pressure jump with the i CLSVoF framework is shown in Fig. 12(b), and first order convergence is found with our model.

The sharpening coefficient C_{pc} is a key parameter governing the sup pression of un physical spurious currents within the i CLSVoF frame work. An additional parameter study demonstrates the effect of the sharpening coefficient on the suppression of the spurious currents. As shown in Fig. 13, the suppression of un physical velocities is im proved with increasing sharpening coefficient C_{pc} , especially for rel atively smaller sharpening coefficient values (0.1 0.4). However, for a relatively large sharpening coefficient (0.5 0.95), the corresponding results give no major improvement. Note that a too large sharpening coefficient may lead to numerical instability.

4.2. Numerical validation of the i CLSVoF framework with evaporation

Several benchmark cases are conducted to validate the i CLSVoF framework with evaporation. For saving computational costs, a sym metrical model is adopted in the current study. As shown in Fig. 14, only a quarter of a 2D droplet with an initial radius of 125 μm is simulated, and symmetry boundary conditions on the left and bottom sides are applied. Outflow boundary conditions (a Dirichlet boundary condition for the pressure field and a Newmann boundary condition for

Fig. 12. (a) Capillary pressure fields of droplets with three different surface tension force models (dashed purple line represents the analytical solution), (b) Relative errors for capillary pressure jump with different mesh resolutions.

Fig. 13. Effect of the sharpening coefficient C_{pc} on the suppression of spurious velocity.

Fig. 14. The numerical setup for 2D droplet evaporation.

velocity field) are applied on the rest two sides so as to let the newly generated vapour from the liquid surface leave the domain freely.

The parameters used in the evaporation cases are listed in Table 2. In order to save the computational cost, the scaled liquid density 10 is used in the current study, and this density scaling approach was also used in several previous studies [5,14,29,56]. A constant surface tension coefficient is used here, meaning that the effect of temperature on the surface tension coefficient is not considered in this work.

4.2.1. Sessile droplet evaporation with the constant mass flux evaporation model

First, we study the evaporation of 2D sessile droplets with the pre defined constant mass flux J , as we can validate the implemen tation of the governing equations in simple manner without taking the calculation of source terms into account. The analytical solution for 2D sessile droplet evaporation with constant mass flux is derived in this section. Let R and R_0 be the shrinking and the initial droplet radius, respectively. The droplet shrinks when the evaporation moves the interface inwards with the interface velocity U_{Γ} . For the 2D sessile droplet cases, the interface velocity U_{Γ} equals the interface recession velocity $\mathbf{U}_{\mathrm{re}},$ leading to

$$
R = R_0 - |\mathbf{U}_\Gamma| t = R_0 - \frac{J}{\rho_l} t. \tag{55}
$$

Let D and D_0 denote the shrinking and the initial droplet diameter, re spectively, and t^* the total evaporation time. The dimensionless droplet diameter changes with the dimensionless time during the evaporation process which is given as

$$
\frac{D}{D_0} = 1 - \frac{2J}{\rho_l D_0} t = 1 - \frac{2Jt^*}{\rho_l D_0} \frac{t}{t^*}.
$$
\n(56)

This formula is the analytical solution for 2D static droplet evapora tion subject to the constant mass flux and is also valid for 3D static evaporation cases.

The crucial aspect of modelling droplet evaporation with the i CLSVoF framework lies in reconstructing the new divergence free ve locity field U_e . After implementing the velocity potential approach (refer to Section 2.2, Eqs. (11) and (12)) proposed in this work for the reconstruction of the new velocity field, the three different ve locity fields are obtained by the simulations: the one field velocity U, the evaporation induced Stefan flow velocity U_s and the newly reconstructed velocity U_e for a 2D droplet subject to the constant mass flux are shown in Fig. 15. The maximum magnitudes of U and U_s are the same. We can also see the contour of the divergence of the velocity

Fig. 15. Different velocities fields of an evaporating droplet (white curve represents the interface): (a) the one-field velocity field U, (b) the evaporation-induced Stefan flow velocity field \mathbf{U}_s , (c) the divergence-free velocity field \mathbf{U}_e , (d) the divergence of \mathbf{U}_e .

field U_e (as shown in Fig. 15(d)), demonstrating that its magnitude is around 10−⁸ which can be regarded as numerically zero. This means that the divergence free velocity field \mathbf{U}_e is successfully reconstructed. The divergence free velocity field \mathbf{U}_{e} for the sessile droplet evaporation case should be zero, but some velocity vertices can be seen around the interface (Fig. $15(c)$). The reason lies in both evaporation and surface tension deforming the interface during the evaporation process, where the interplay between evaporation and surface tension produces additional spurious velocities. These additional spurious currents are challenging to reduce, especially with the algebraic VoF approach of OpenFOAM on which our i CLSVoF framework is based. Furthermore, the distribution of the spurious velocities is symmetrical along the diagonal of the computational domain (as shown in Fig. $15(c)$). A symmetrical distribution of spurious velocities around an evaporating droplet is more stable than the case with random distribution.

In Fig. 16, we present the velocity contour of the one field velocity field U. All the vectors are perpendicular to the interface (represented by the white solid curve) and point from the liquid phase to the vapour phase. Additionally, a velocity jump can also be found around the interface. This demonstrates that the influence of spurious currents on the droplet evaporation is negligible. Our further numerical validations (as shown in Fig. $19(a)$) also confirm that these symmetrical spurious velocities never deform the interface in an un physical way and perfect interface shapes are predicted with our i CLSVoF framework.

Three different mesh sizes are conducted to study the effect of mesh resolution on the numerical results. The mesh resolution (number of cells in x and y directions) are 25×25 , 50×50 and 100×100 for coarse, medium and fine mesh, respectively. Fig. 17 shows the temporal evolution of the dimensionless droplet diameter. t/t^* on the x axis denotes the dimensionless time. The agreement between the numerical solution and the corresponding analytical solution is getting better with finer mesh. For the fine mesh, the analytical curve perfectly

Fig. 16. Vector contour of the one-field velocity field U around an evaporating sessile droplet (the white solid curve represents the interface).

goes through all the numerical data points (as shown in the locally magnified plot in Fig. 17(b)).

To validate the evaporation model quantitatively, the relative error of the predictions for the dimensionless droplet diameter $\hat{D} = D/D_0$ is defined by

$$
E(\hat{D}) = \frac{|\hat{D}^n - \hat{D}^a|}{\hat{D}^a},\tag{57}
$$

Fig. 17. The mesh convergence study: (a) the global plot, (b) the local magnified plot.

Fig. 18. The numerical error in predicting dimensionless droplet diameter for different mesh resolutions (larger cell number corresponds to finer mesh).

where \hat{D}^a and \hat{D}^n are the analytical and numerical dimensionless droplet diameters, respectively. $E(\hat{D})$ is calculated with different mesh resolutions are compared in Fig. 18 (when $t/t^* = 0.8$). Four mesh resolutions, 10×10 (10 cells in each direction of the computational domain as shown in Fig. 14), 25×25 , 50×50 and 100×100 are adopted in the numerical simulations. The relative error $E(\hat{D})$ is lower than 0.01 when the mesh resolution is 100 \times 100, and first order convergence is obtained with the evaporation model developed in this work. Additionally, the numerical error for predicting the shrinking droplet diameter is still acceptable (around 5%) even for a coarse mesh with our improved numerical model.

Interface capturing is another crucial aspect in modelling droplet evaporation, and the i CLSVoF approach developed in this work can capture the free surface in a sharp manner. The corresponding vali dation is also conducted. The solid yellow curves are the analytical solution in different evaporation stages (as can be seen in Fig. 19), while the blue dots are the numerical data collected on the 0.5 iso surface for the eight stages. We can see from Fig. 19(a) that the agreement between the numerical data and the corresponding analyti cal solution is perfect. It is also promising to see that the circular droplet shape is maintained, meaning that the influence of spurious currents on destroying the droplet shape is suppressed throughout the whole evaporation process. For a given stage, the mesh convergence study indicates no major difference when refining the mesh (as shown in

Fig. 19(b)), because the numerical data are only collected at the 0.5 iso surface, which does not depend on mesh resolution.

4.2.2. Sessile droplet evaporation with the thermally driven evaporation model

After validating the sessile droplet evaporation with constant mass flux model, the i CLSVoF framework is extended to incorporate evap oration with more complex evaporation mass flux calculations. The mass flux of the thermally driven evaporation model depends on the interfacial temperature difference, which drives the phase change from liquid to vapour. The analytical solution for the shrinking droplet di ameter square D^2 during the evaporation process was derived with the interface energy balance in the literature [57]. The analytical solution is given as the ordinary differential equation

$$
\frac{dD^2}{dt} = -\frac{8k_g}{\rho_l c_{p_g}} \ln(1 + B_q),\tag{58}
$$

where B_q is the Spalding mass transfer number defined by

$$
B_q = \frac{c_{p_g}(T^{\infty} - T_{\text{sat}})}{h_{\text{ev}}},\tag{59}
$$

and T^{∞} is the temperature on the boundary. This is known as the D^2 law, and it is derived for the droplet evaporation in an infinite domain. The modified D^2 law takes into account the effect of computational do main size on the evolution of droplet diameter, and this model is more suitable for validating droplet evaporation in a finite computational domain. The modified analytical solution is

$$
\frac{dD^2}{dt} = -\frac{8k_g}{\rho_l c_{pg}} \frac{\ln(1 + B_q)}{\ln(D_s/\sqrt{D^2})},\tag{60}
$$

where D_s is the diameter of the inscribed circle for the computational domain [5]. However, the unknown D^2 appears in the denominator due to the correction, and consequently we can only solve Eq. (60) numerically.

The numerical set up for the 2D simulation of the thermally driven evaporation model is similar to the set up as shown in Fig. 14. The only difference is that in the initial configuration the temperature for the droplet is equal to its saturation temperature T_{sat} while the temperature for the rest of the domain is higher than the saturation temperature. Ad ditionally, a Dirichlet boundary condition for the temperature field on the boundaries except the symmetry boundaries must be applied. The numerical validation in this part starts with the temperature difference of 50 K, and the corresponding temperature distribution around the evaporating droplet is shown in Fig. $20(a)$. The temperature gradient around the droplet drives the phase change from liquid to vapour, and the temperature inside the droplet stays constant and equals the satu ration temperature. As shown in Fig. 20(b), the interface velocity U_T

Fig. 19. Numerical validations of the interface capturing during evaporation of a sessile droplet: (a) the global plot, (b) the mesh convergence study.

Fig. 20. (a) The temperature distribution around the droplet, (b) the interface velocity field U_r .

Fig. 21. (a) The numerical validation for thermally driven evaporation model, (b) the temperature evolution during the evaporation process (∕ [∗] = 0 corresponds to the initial temperature field).

points towards the droplet centre, which is related to the evaporation induced droplet shrinking. Additionally, the interface velocity U_\varGamma is dominant only at the droplet interface, which demonstrates that the divergence free velocity field is successfully reconstructed also for the thermally driven evaporation model.

The quantitative study of the shrinking droplet diameter is com pared against the corresponding analytical model according to Eq. (60). The dimensionless droplet diameter and the dimensionless time are adopted. It can be seen from Fig. $21(a)$, that an accurate solution is achieved for around 65% of the total evaporation time t^* with the

Fig. 22. Validations for four cases with different Stefan number St (solid curves represent the analytical solutions, and points represent the corresponding numerical data).

graded mesh shown in Fig. 20(a). As discussed in the literature [5], the simulation proceeded to $(\frac{D}{D_0})^2 = 0.5$ was considered reasonable for validating their evaporation models. For this reason, we consider going even beyond that down to $(\frac{D}{D_0})^2 < 0.3$ as sufficient for numerical vali dations. $(\frac{D}{D_0})^2 = 0.25$ corresponds to the 65% evaporation time on the x axis. For the rest 35% evaporation time, adaptive mesh refinement is needed for tracking the shrinking droplet interface as droplet size be comes very small. The corresponding temperature evolution collected from the bottom left corner to the bottom right corner can be seen in Fig. 21(b). The solid green line represents the initial configuration for the temperature field, where the transition band from the saturation temperature to the temperature value corresponding to the boundaries in the initial temperature field is relatively large. Further mesh refine ment can shorten this transition band. The other curves in Fig. 21(b) show the evolution of temperature (the purple arrow indicates time going on) during the evaporation process. The saturation temperature inside the droplet and the temperature at the domain boundary are strictly maintained constant during the evaporation process.

Additionally, droplet evaporation with different Stefan numbers is further studied to validate our model extensively. The Stefan number is defined as

$$
St = \frac{c_{p_g} \Delta T}{h_{\text{ev}}},\tag{61}
$$

where $\Delta T = T_{\infty} - T_{\text{sat}}$ is the temperature difference between the satura tion temperature T_{sat} inside the droplet and the temperature T_{∞} at the boundaries, and c_{p_g} the specific heat capacity of the gas/vapour phase. As can be seen from Fig. 22, four evaporation cases with different Stefan numbers are presented. Cases with a large Stefan number experience faster evaporation, and a good agreement between the numerical and the corresponding solution of the analytical model is observed for all cases.

4.2.3. Sessile droplet evaporation at room temperature

The numerical set up for modelling droplet evaporation at room temperature is the same as for the two aforementioned evaporation models (refer to Fig. 14). In addition to prescribing the outflow bound ary conditions, a Dirichlet boundary condition for the vapour mass fraction field is applied on the boundaries except for the symmetry boundaries.

The essential parameters used in the simulations are listed in Ta ble 2. The one field velocity field during the evaporation process is shown in Fig. 23(a) where a velocity jump can be seen around the interface region (the white curve represents the interface). The corre sponding vapour mass fraction field is shown in Fig. $23(b)$ with the vapour mass faction at the top and right boundaries given as constants prescribed by the Dirichlet boundaries conditions. The vapour mass fraction gradient at the interface drives the droplet evaporation.

The shrinking droplet diameter D is recorded to quantitatively vali date the evaporation model by comparing the diameter predicted by the numerical simulations to the analytical solution given by the so called D^2 law [57]. However, for the droplet evaporation in a finite domain, the classical D^2 law should be corrected to take the computational domain size into account, and a modified analytical model is adopted in this work [5]. This modified D^2 law is given by

$$
\frac{dD^2}{dt} = -\frac{8\rho_g D_v}{\rho_l} \frac{\ln(1 + B_y)}{\ln(D_s / \sqrt{D^2})},\tag{62}
$$

where D_s is the diameter of the inscribed circle for the computational domain, and B_y the Spalding mass transfer number defined as

$$
B_y = \frac{Y^{\Gamma} - Y^{\infty}}{1 - Y^{\Gamma}}
$$
\n(63)

with Y^{∞} being the vapour mass fraction far way from droplets [57].

A parameter study on a wide range of evaporation temperatures was conducted to validate the evaporation model. As shown in Fig. 24, the solid curves represent corresponding analytical solutions given by Eq. (62) and the points represent numerical solutions predicted by the evaporation model. Good agreement between numerical and analytical solutions are found for evaporation at relatively higher temperatures (303.15 K, 313.15 K and 323.15 K). Some minor discrepancy between the numerical and analytical solutions are found, especially for evap oration at relatively low temperatures (283.15 K and 293.15 K). The underlying reason is that evaporation at lower temperatures is more susceptible to the influence of un physical spurious velocity.

5. Conclusions

In this work, the i CLSVoF framework which combines the sharp interface LS method and the mass conserving algebraic VoF method with filtering steps to suppress un physical velocities, has been devel oped and implemented in OpenFOAM. The numerical framework was specially developed for modelling surface tension dominant flow with and without phase change in a simple yet accurate way.

The governing equation was solved based on the one field formu lation using the FVM. An improved surface tension force model and the additional filtering approach were incorporated into the i CLSVoF framework to further reduce un physical spurious currents. The superi ority of the improved numerical framework over the conventional VoF and the CLSVoF approaches was demonstrated by several benchmark cases. The sharp interface was captured, and interface diffusion was suppressed. A promising amount of suppression of un physical spurious currents was achieved with our improved numerical framework.

The i CLSVoF framework was further extended to model sessile droplet evaporation, especially in micro scale. A simple approach was proposed and implemented to reconstruct a divergence free velocity by removing the evaporation induced irrotational velocity from the one field velocity field to predict the evaporation rate accurately. The constant mass flux evaporation model was first incorporated into the i CLSVoF framework to check the implementation of the basic equations. The numerical validations proved the successful construction of the divergence free velocity field. Besides, the model accurately predicted the droplet evaporation as shown by comparing the shrinking dimen sionless droplet diameter to an analytical solution derived in this work. The interface capturing of the i CLSVoF was also demonstrated to be accurate enough by comparing the numerically captured free surface to the analytical solution; crucially, no spurious velocity induced interface deformation was found during the evaporation process. The thermally

Fig. 23. (a) The one-field velocity field U of droplet evaporation at room temperature, (b) The vapour mass fraction field of droplet evaporation at room temperature (white curve represents the interface).

Fig. 24. Validations for five cases with different evaporation temperatures (solid curves represent the analytical solutions, and points represent the corresponding numerical data).

driven evaporation model was implemented to account for the temper ature gradient induced phase change. A novel density ratio dependent evaporation coefficient was used to calculate the mass flux at the inter face rather than determining the evaporation coefficient empirically. The modified D^2 law was used to validate the numerical model, and a good agreement between the numerical solution and the analytical solution was achieved. Additionally, parameter studies were conducted to demonstrate the accuracy of the thermally driven evaporation model for a wide range of Stefan numbers. The third evaporation model incor porated in this work describes the sessile droplet evaporation at room temperature. This model can predict the evaporation rate accurately for different evaporation temperatures while some minor discrepancy between numerical and analytical solutions is found for relatively small evaporation temperature, which we consider still acceptable. Droplet evaporation with a fixed triple contact line (also known as evaporation with Constant Contact Radius (CCR) mode) is ubiquitous in case of drying droplets on rough substrates. In the future work, the i CLSVoF framework will be extended to model the evaporation of micro sized droplets under CCR conditions.

Declaration of competing interest

The authors declare that they have no known competing finan cial interests or personal relationships that could have appeared to influence the work reported in this paper.

Data availability

Data will be made available on request.

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Appendix

The purpose of the current numerical demonstration is to compare the i CLSVoF model developed in this work against the conventional VoF model in terms of free surface capturing. The interDyMFoam solver of OpenFOAM 5.x is adopted for conducting simulations with the conventional VoF model.

In the current case, the surface tension force is not dominant other than in the droplet relaxation cases. The numerical set up is that a water column is initially situated at the bottom left of the 2D compu tational domain with the top as an opening, and the other boundaries are regarded as walls (as shown in Fig. $25(a)$). The essential parameters for the simulations are outline in Table 3.

The Adaptive Mesh Refinement (AMR) is used to track a sharp interface with a finer mesh around the interface region and a coarser mesh elsewhere to save computational costs (as shown in Fig. 25(b)).

Fig. 25. (a) Geometry and boundaries of the 2D dam break simulation, (b) Numerical simulations with 2D AMR (the white curve represents the free-surface).

Fig. 26. Evolutions of the free-surface in the dam break simulations: (a) with the VoF method (t = 0.2 s), (b) with the i-CLSVoF method (t = 0.2 s), (c) with the VoF method (t $= 0.5$ s), (d) with the i-CLSVoF method (t = 0.5 s).

The total simulation time is 0.5 s which is enough for water to reach both the bottom obstacle and the right wall. As shown in Fig. 26, Figs. 26(a) and $26(c)$ are the simulation results with the VoF approach, and Figs. 26(b) and 26(d) are with the i CLSVoF model. When water hits the obstacle and forms a wave over the obstacle, the results with the VoF approach show some interface diffusion (see Fig. 26(a)) while the i CLSVoF captures the sharp interface accurately (see Fig. 26(b)). As the simulation progresses, water hits the right wall and some water

is bounced back to the container. Fig. $26(c)$ shows large interface diffusion like rising gas around centre of the right wall for VoF, while Fig. 26(d) presents a sharp interface captured by the i CLSVoF without interface diffusion.

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