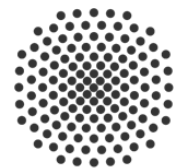


On the importance of turbulence modelling in a linear mean field analysis of turbulent flame configurations

International Conference on Numerical Combustion 2024

Kyoto, 10.05.2024

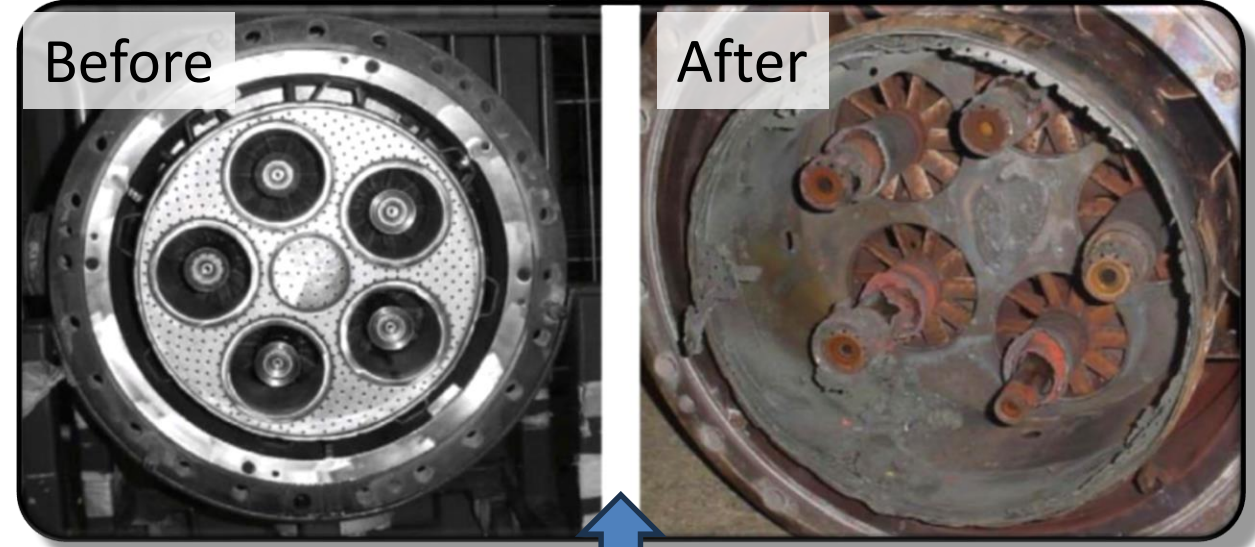
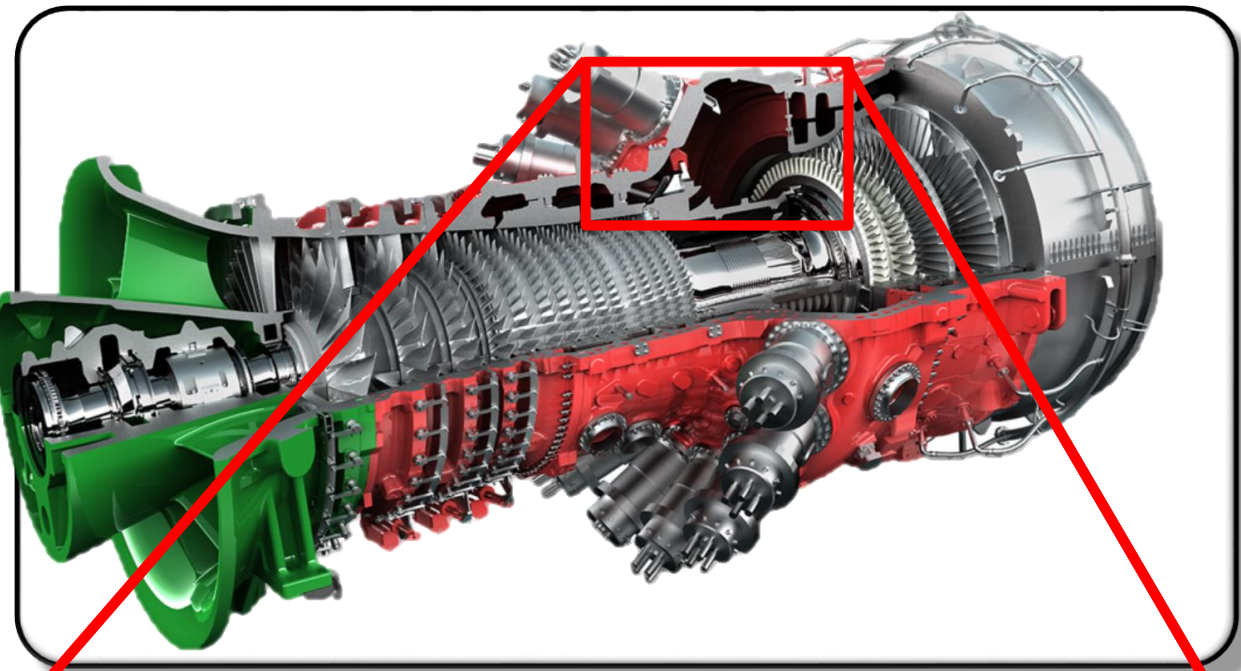
Thomas Ludwig Kaiser, Thorsten Zirwes, Feichi Zhang, Kilian Oberleithner



**University of
Stuttgart**

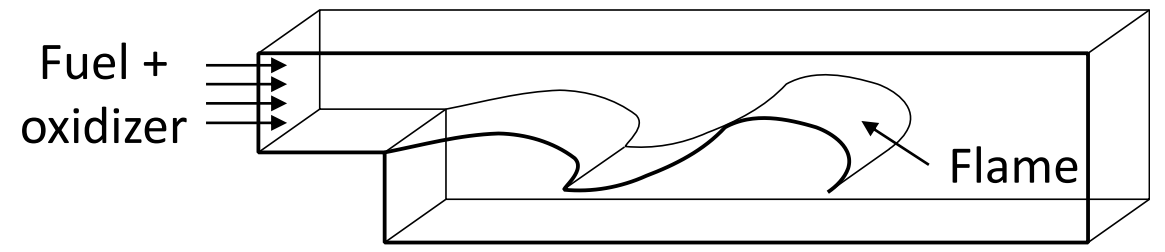


Thermoacoustic instabilities increase emissions, decrease efficiency or even destroy the engine



Lieuwen 2005

Flame configuration



Unstable operating point





Motivation: Prospect on linearized turbulent reacting flows on thermoacoustics

Current methods to study thermoacoustics

Brute force
LES

Helmholtz
solvers

1-D network
models

Precision/accuracy

Simplifications

Numerical costs

Immediate insight into physics

Flame
dynamics
resolved

Flame dynamics
needed as input (FTF)

Turbulent flame modelling in linearized equations of a turbulent reacting flow



- Holistic model including flame dynamics
- Fewer simplifications
- Immediate insight into physics (optimization)
- Shape/adjoint optimization

- 1 Thomas Ludwig Kaiser
TU Berlin, Germany
**Challenges in modelling
turbulent transport of
density varying flows in
the linear framework**



- 2 Parth Patki
Georgia Tech, USA
**Tri-global linear
stability analysis in
combustion systems**



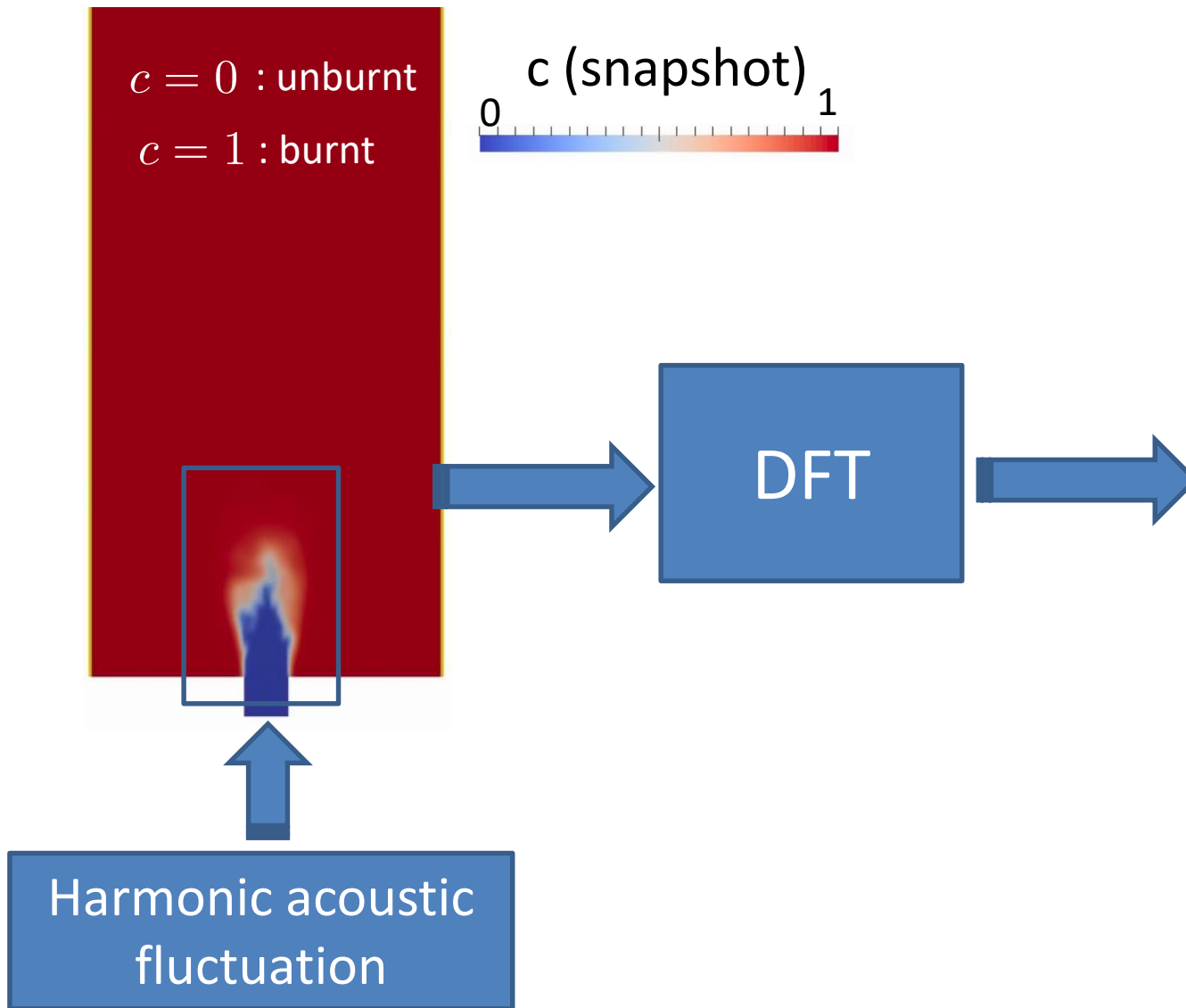
- 3 Philipp Brokof
TU Munich, Germany
**Investigations on the
feedback mechanism of
intrinsic thermoacoustic
modes**



- 4 Chuhan Wang
Tsinghua University, China
**Linear modelling of
thermoacoustic
feedback in a laminar
flame combustor**



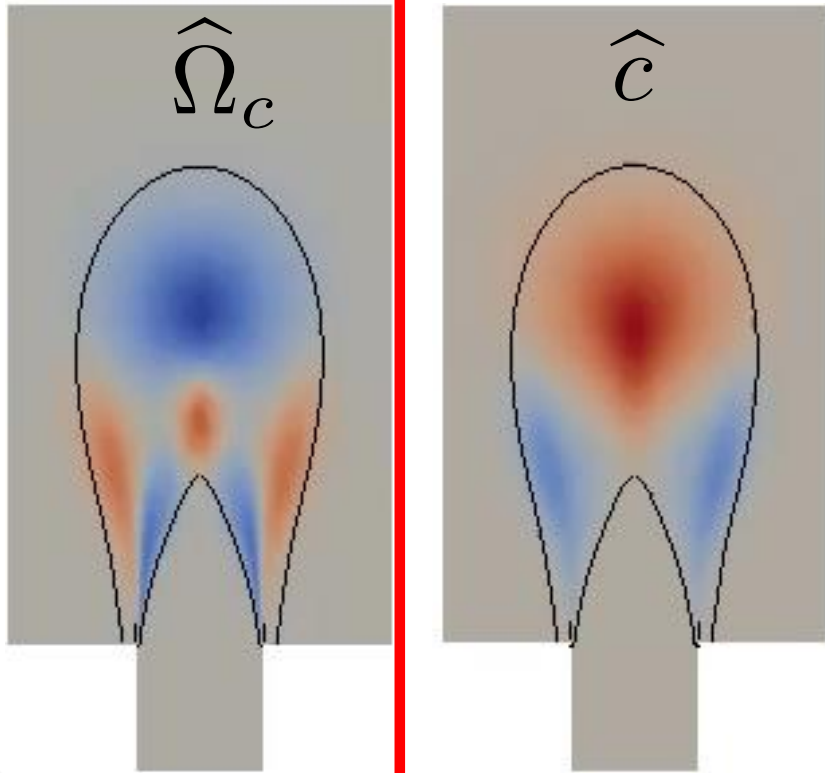
Turbulent flame is acoustically forced and its response measured





A priori analysis of reaction rate fluctuations

LES: Response to acoustic actuation



A priori analysis

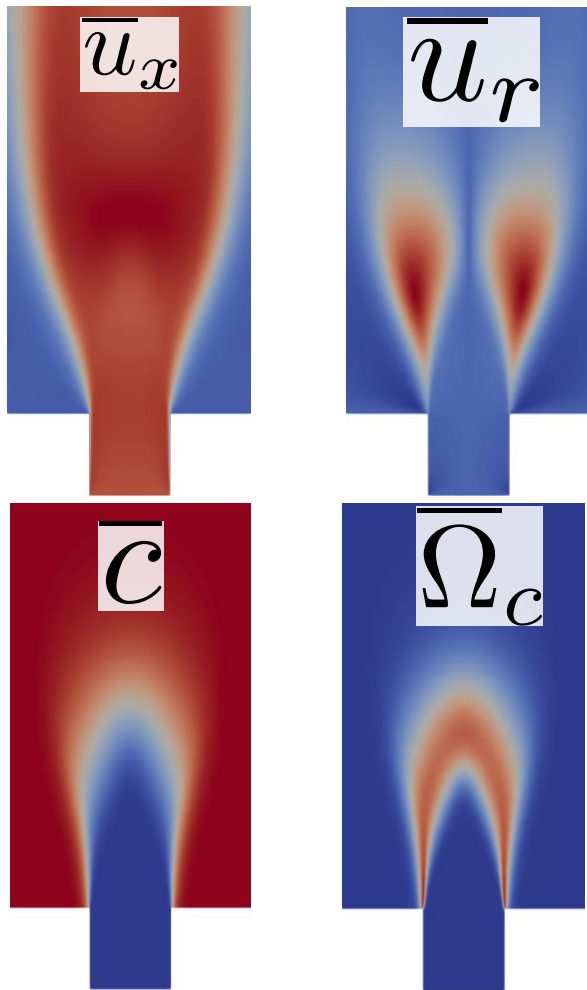
Linearized turbulent flame model

$$\hat{\Omega}_c = f(\bar{c}, \hat{c})$$

Kaiser et al., Comb. Flame, 2023

A posteriori analysis of the turbulent flame linearization using the linearized flow solver FELiCS

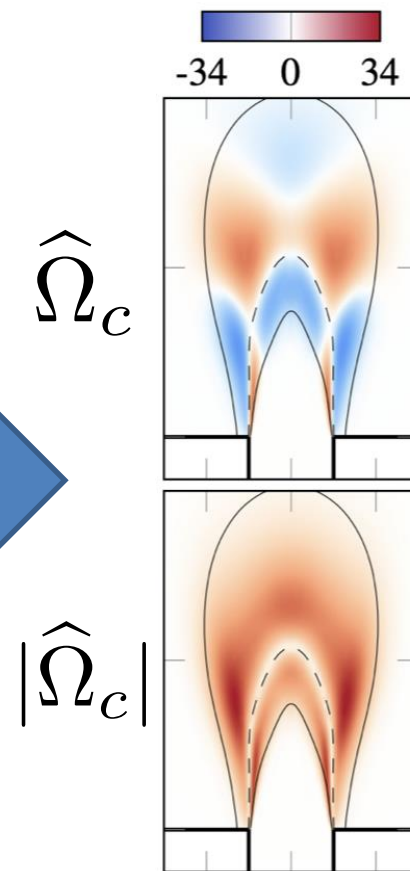
Temporal mean flow



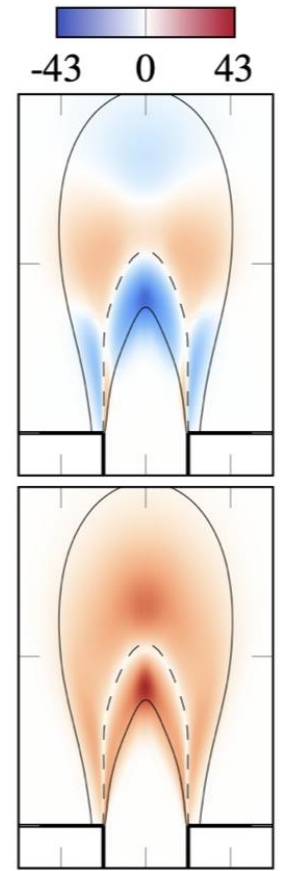
A posteriori analysis
of entire set of
equations
(including linearized
reaction model)

FELiCS

Response to acoustic forcing



Validation (LES)

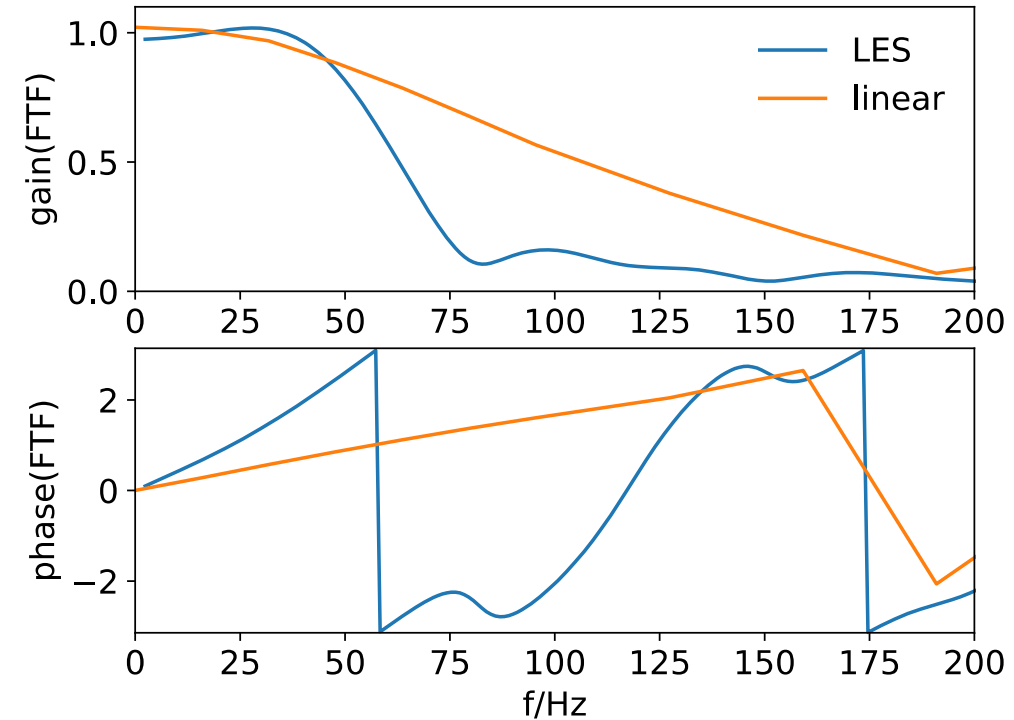


Flame transfer function, currently, incorrect. Probable cause: Insufficient turbulence modelling

Heat release fluctuation

$$\text{FTF} = \frac{\widehat{q}/\bar{q}}{\widehat{u}_{\text{ref}}/\bar{u}_{\text{ref}}}$$

Acoustic velocity fluctuation



State of the Art for **Linearization of Turbulent Reacting Flow**



Non-linear governing equations (Example: mass transport)

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0$$

$$\Phi = \bar{\Phi} + \Phi'$$

RANS

$$\nabla \cdot (\bar{\rho} \bar{\mathbf{u}}) + \nabla \cdot (\overline{\rho \mathbf{u}'}) = 0$$

unclosed term

Non-linear governing equations (Example: mass transport)

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0$$

$$\Phi = \tilde{\Phi} + \Phi''$$

RANS

$$\tilde{\Phi} = \frac{\overline{\rho \Phi}}{\bar{\rho}}$$

$$\nabla \cdot (\bar{\rho} \tilde{\mathbf{u}}) + \nabla \cdot \overline{(\rho \mathbf{u}'')} = 0$$

No additional closure problem
in continuity equation

Non-linear governing equations (Example: mass transport)

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0$$

$$\Phi = \bar{\Phi} + \hat{\Phi} + \Phi'$$

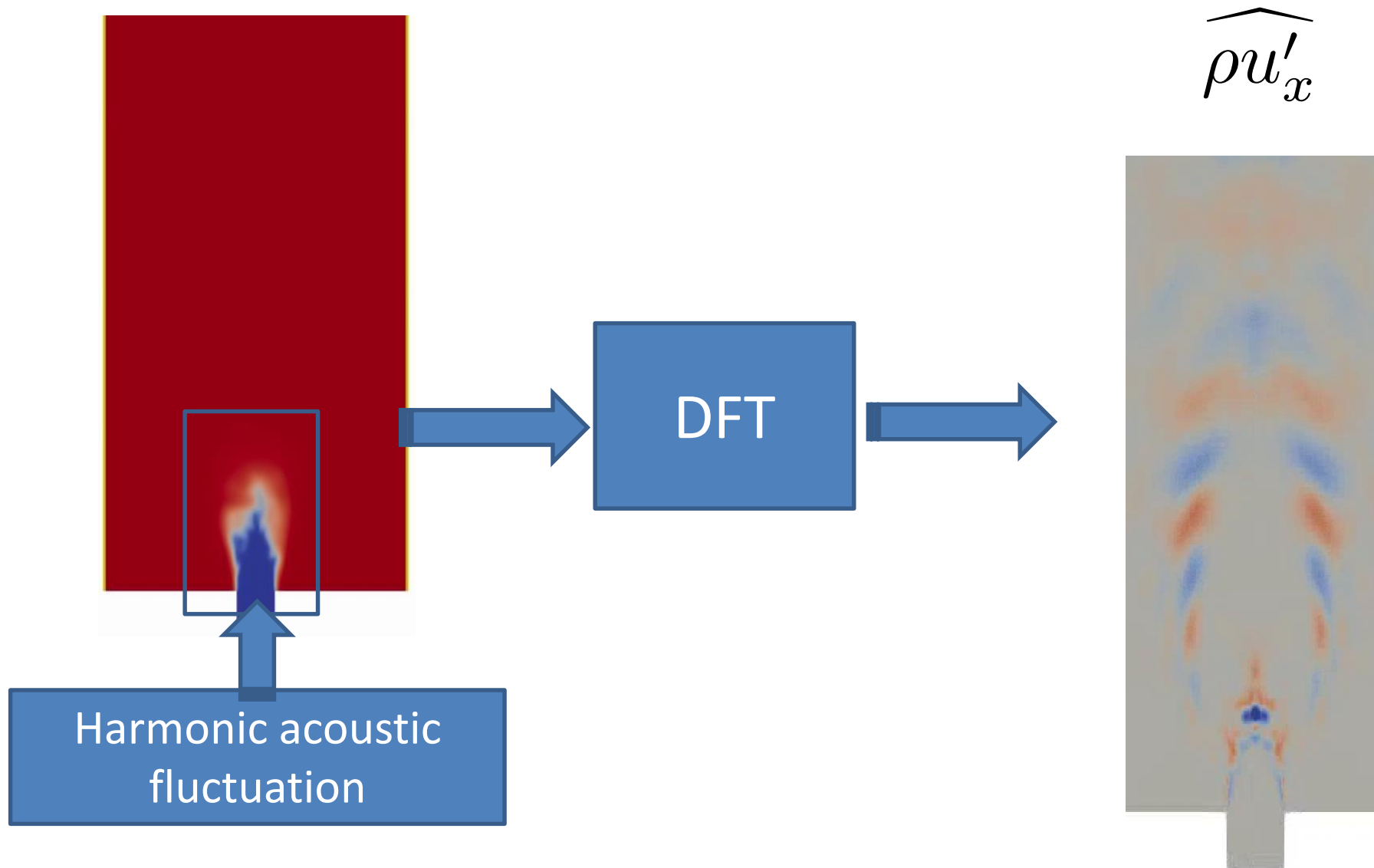
Linear framework

$$\frac{\partial \hat{\rho}}{\partial t} + \nabla \cdot (\hat{\rho} \bar{\mathbf{u}}) + \nabla \cdot (\bar{\rho} \hat{\mathbf{u}}) + \nabla \cdot (\widehat{\rho \mathbf{u}'}) = 0$$

unclosed term



Unclosed term in continuity equation



Solution: New tripple decomposition technique

RANS framework

$$\overline{\rho\Phi''} \stackrel{!}{=} 0$$



$$\Phi = \tilde{\Phi} + \Phi''$$



$$\tilde{\Phi} = \frac{\overline{\rho\Phi}}{\bar{\rho}}$$

Linear framework

$$\widehat{\rho\Phi''} \stackrel{!}{=} 0$$

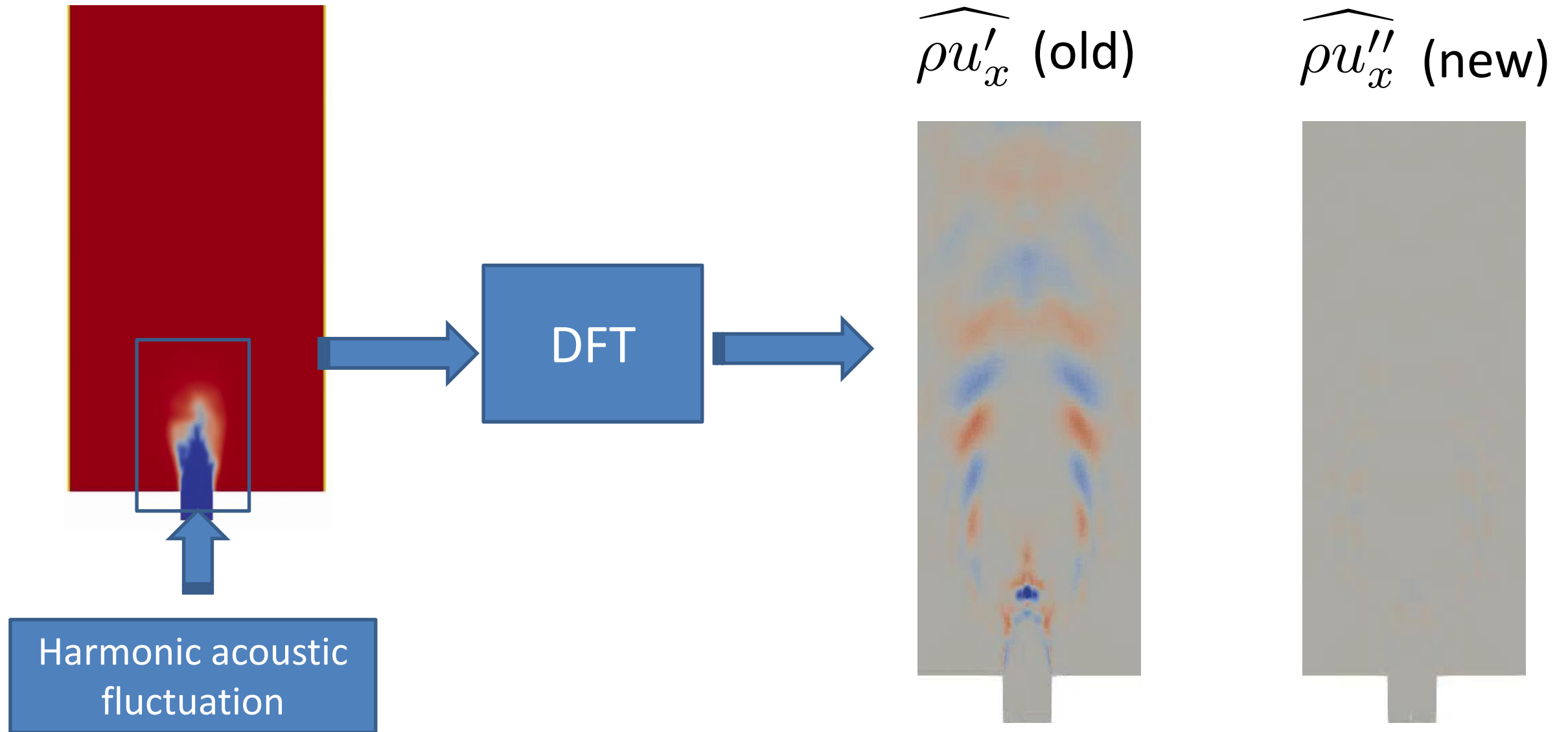


$$\Phi = \tilde{\Phi} + \overset{\circ}{\Phi} + \Phi''$$



$$\overset{\circ}{\Phi} = \frac{\widehat{\rho\Phi} - \hat{\rho}\tilde{\Phi}}{\bar{\rho}}$$

Applying the new tripple decomposition technique leads to a closed continuity equation



Closure problem is even more significant in remaining transport equations

Non-linear governing equations (Example: passive scalar transport)

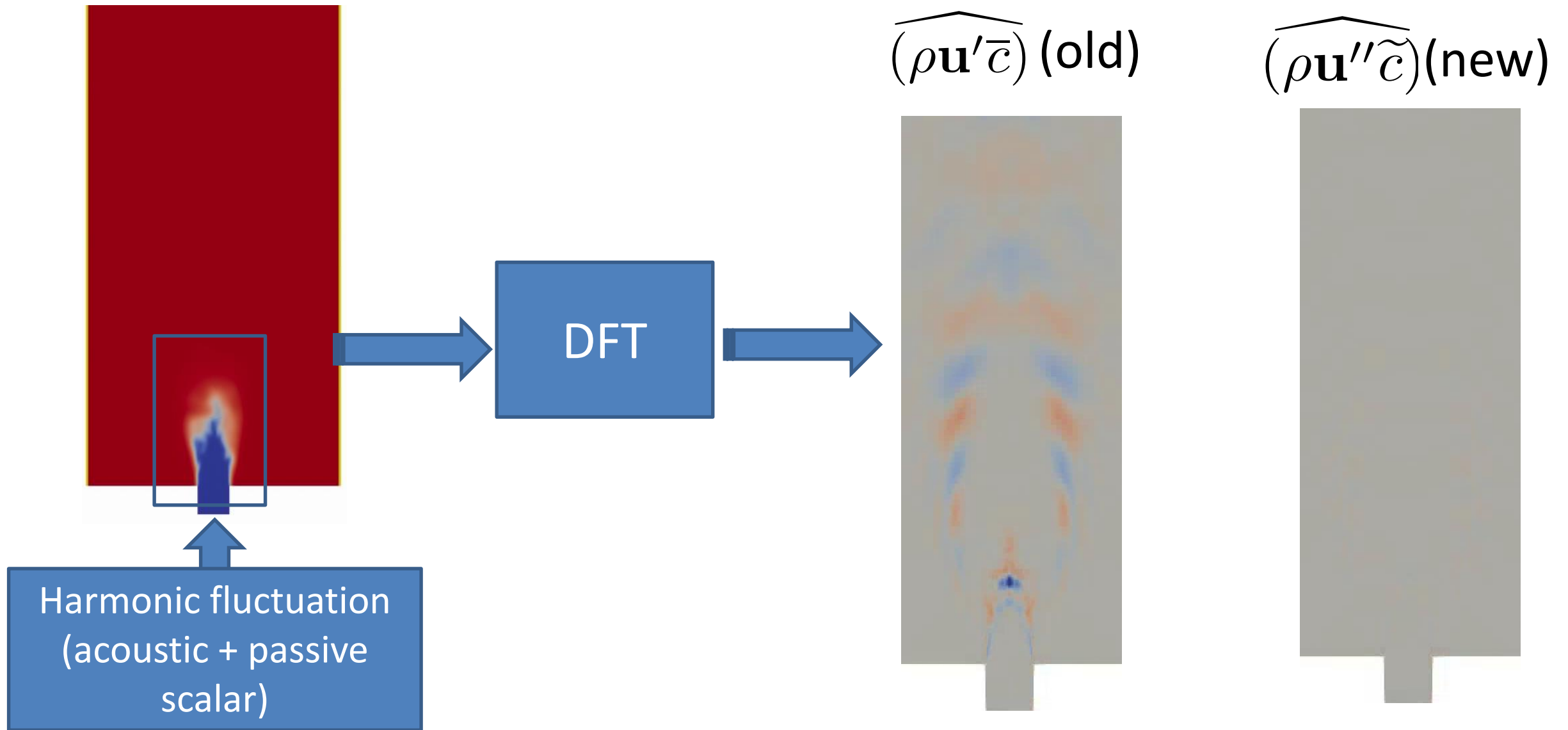
$$\frac{\partial \rho c}{\partial t} + \nabla \cdot (\rho \mathbf{u} c) - \nabla \cdot (D \nabla c) = 0$$

$$\Phi = \bar{\Phi} + \hat{\Phi} + \Phi'$$

Linear framework

$$\begin{aligned} & \nabla \cdot (\overline{\rho \mathbf{u} c}) + \nabla \cdot (\overline{\rho \mathbf{u} \hat{c}}) + \nabla \cdot (\overline{\rho \mathbf{u} c'}) \\ & + \nabla \cdot (\overline{\rho \hat{\mathbf{u}} c}) + \cancel{\nabla \cdot (\overline{\rho \hat{\mathbf{u}} \hat{c}})} + \nabla \cdot (\overline{\rho \hat{\mathbf{u}} c'}) \\ & + \nabla \cdot (\overline{\rho \mathbf{u}' \bar{c}}) + \nabla \cdot (\overline{\rho \mathbf{u}' \hat{c}}) + \nabla \cdot (\overline{\rho \mathbf{u}' c'}) \end{aligned}$$

Applying the new tripple decomposition technique reduces closure problem in the remaining transport equations





New tripple decomposition technique significantly reduces the closure problem

Non-linear governing equations (Example: passive scalar transport)

$$\frac{\partial \rho c}{\partial t} + \nabla \cdot (\rho \mathbf{u} c) - \nabla \cdot (D \nabla c) = 0$$

$$\Phi = \tilde{\Phi} + \overset{\circ}{\Phi} + \Phi''$$

Linear framework

$$\begin{aligned} & \nabla \cdot \widehat{(\rho \tilde{\mathbf{u}} \tilde{c})} + \nabla \cdot \widehat{(\rho \tilde{\mathbf{u}} \overset{\circ}{c})} + \nabla \cdot \widehat{(\rho \tilde{\mathbf{u}} c'')} \\ & + \nabla \cdot \widehat{(\rho \overset{\circ}{\mathbf{u}} \tilde{c})} + \nabla \cdot \widehat{(\rho \overset{\circ}{\mathbf{u}} c)} + \nabla \cdot \widehat{(\rho \overset{\circ}{\mathbf{u}} c'')} \\ & + \nabla \cdot \widehat{(\rho \mathbf{u}'' c)} + \nabla \cdot \widehat{(\rho \mathbf{u}'' c')} + \nabla \cdot \widehat{(\rho \mathbf{u}'' c'')} \end{aligned}$$

- State of the Art for **Linearization of Turbulent Reacting Flow**



- Reynolds averaging leads to non-negligible additional closure problems in linear transport equations (analogous to RANS)

$$\Phi = \bar{\Phi} + \hat{\Phi} + \Phi'$$



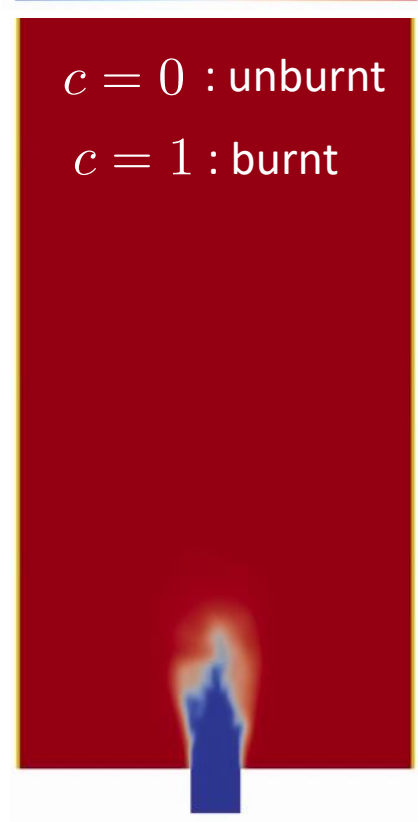
- Closure problem can be simplified by performing new decomposition technique i.e. linearize around Favre mean state

$$\Phi = \tilde{\Phi} + \overset{\circ}{\Phi} + \Phi''$$

$$\tilde{\Phi} = \frac{\overline{\rho\Phi}}{\bar{\rho}} \quad \overset{\circ}{\Phi} = \frac{\widehat{\rho\Phi} - \hat{\rho}\tilde{\Phi}}{\bar{\rho}}$$

Case of choice: LES of Bunsen flame in cylindrical confinement (Kobayashi configuration)

c (snapshot)



$c = 0$: unburnt
 $c = 1$: burnt

Fresh gas
(air+CH4)

hr (mean)



Reference solution is a LES conducted at KIT

- Very low Mach number ($Ma < 0.015$)
- $Re = 8000$
- openFoam
- Smagorinsky turbulence model
- Progress variable, c , in combination with a look-up table
- Turbulence Flamespeed Closure

$$\overline{\Omega}_c \propto \rho_0 \frac{S_t^2}{D_l + D_t} \tilde{c}(1 - \tilde{c})$$

$$\frac{S_t}{S_l} = 1 + \frac{u'}{S_l} (1 + Da^{-2})^{-1/4}$$