

# Creep behavior of a precipitation-strengthened A2-B2 refractory high entropy alloy

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## Highlights

- 1 • Ta-Mo-Ti-Cr-Al with two-phase A2+B2 microstructure exhibits competitive creep resistance
- 2 • Precipitation strengthening by B2 precipitates in A2 matrix is effective even close to solvus
- 3 • N-type rafting is obtained for grains with close to  $\langle 100 \rangle$  parallel to the loading direction

## Abstract

4 Refractory high entropy alloys (RHEA) consisting of a disordered A2 matrix and ordered B2 precipitates  
5 mimic the microstructures of state-of-the-art, Ni-based superalloys with A1 matrix and L1<sub>2</sub> precipitates.  
6 They are promising candidates for high-temperature applications because of their high melting points.  
7 27.3Ta-27.3Mo-27.3Ti-8Cr-10Al (in at. %) is a relevant example where this A2-B2, two-phase  
8 microstructure is formed by a precipitation reaction and remains stable even at temperatures close to the  
9 precipitate solvus temperature ( $T_S$ ). The present study systematically addresses the creep response of  
10 poly-crystalline 27.3Ta-27.3Mo-27.3Ti-8Cr-10Al at temperatures of 1000 °C and above. Compared to  
11 poly-crystalline, single-phase A2 and B2 RHEA, a substantially higher creep resistance is observed for  
12 the two-phase alloy while minimum creep rates comparable to those of state-of-the-art single-crystalline  
13 A1-L1<sub>2</sub> CMSX-4 are found. This is specifically remarkable considering its poly-crystalline condition,  
14 the open matrix A2 crystal structure and extremely high homologous temperature of 0.98  $T_S$  compared  
15 to 0.85  $T_S$  for CMSX-4. Consistent with a positive lattice misfit, directional coarsening of precipitates  
16 is revealed in 27.3Ta-27.3Mo-27.3Ti-8Cr-10Al perpendicular to the compression direction after creep  
17 in grains with  $\langle 100 \rangle$  close to the compression direction.

## Keywords

18 Refractory high entropy alloys, compositionally complex alloys, super alloys, creep, high temperature

# 1. Introduction

19 The concept of high entropy alloys (HEA) has attracted the attention of many researchers since its  
20 introduction in the early 2000s by Yeh et al. [1,2] and Cantor et al. [3]. While HEA synthesized from 3d  
21 transition metals like Fe, Cr, Ni, Co and others have been of particular interest with respect to mechanical  
22 properties at ambient and low temperatures, the concept was soon extended to refractory high entropy  
23 alloys (RHEA) by the seminal work of Senkov et al. [4–7]. RHEA are composed of refractory elements  
24 and exhibit usually high melting points beyond 2000°C leading to excellent strength even up to 1600 °C  
25 [5]. As all refractory elements suffer from catastrophic oxidation at intermediate and high temperatures  
26 [8], concepts to improve their oxidation resistance were developed and introduced by adding potentially  
27 passivating elements like Al, Cr, etc. [9–12]. Particularly, RHEA containing Ta, Cr and Ti were  
28 identified to provide a strategy to achieve oxidation resistance by the formation of a slowly growing  
29 rutile type oxide [8,12–15]. Even though this protective oxide does not contain Al, Al promotes its  
30 formation and is thus added as a required element in the alloys [14]. Furthermore, Al and Ti lead to a  
31 significant reduction of density [16–18] as compared to RHEA composed of only refractory elements.  
32 When synthesized in close to equimolar composition and subjected to high-temperature homogenization  
33 treatments, most of the alloys from the Ta-Mo-Nb-Cr-Ti-Al system form almost single-phase material  
34 with disordered, body centered cubic (BCC) A2 (Strukturbericht designation) or ordered B2 crystal  
35 structure and low amounts of secondary phases [16–20].

36 Recently, considerable effort has been placed on advanced alloying strategies in non-equimolar,  
37 complex alloys (RCCA) to mimic the microstructures of state-of-the-art, Ni-based superalloys with  
38 Strukturbericht A1 (face centered cubic, FCC) matrix and L<sub>12</sub> precipitates. The analogous  
39 microstructure in RCCA consists of a disordered A2 matrix and ordered B2 precipitates and the class of  
40 alloys is often referred to as refractory (or BCC) superalloys [21]. A huge variety of experimental [22–  
41 25] and theoretical considerations [26] were made to deliberately synthesize the A2-B2 microstructure.  
42 Particularly, within the Ta-Mo-Cr-Ti-Al system, alloys were identified that form the desired A2-B2  
43 microstructures by a nucleation and growth process and the relevant reaction sequence of phase  
44 separation and ordering [27,28].

45 In view of recent systematic creep studies by Gadelmeier et al. [29] and Liu et al. [30], this type of  
46 microstructural tailoring aimed at efficient precipitation strengthening seems imperative to achieve  
47 competitive creep performance over established high temperature materials. In these works, the single-  
48 phase A2, equimolar TiHfNbTaZr was investigated with respect to its tensile creep behavior at 980 °C  
49 (50 – 120 MPa) and 1100 °C (10 – 30 MPa) [29,30]. The minimum creep rate at both temperatures was  
50 higher than 10<sup>-7</sup> s<sup>-1</sup> and thus creep resistance is notably lower than that of a state-of-the-art, precipitation-  
51 strengthened, single-crystalline Ni-based superalloy (e.g., CMSX-4 with a L<sub>12</sub> solvus temperature of  
52 approximately 1280 °C [31]). This discrepancy is not only attributed to the poly-crystalline  
53 microstructure of HfNbTaTiZr, but also to the presence of a single-phase microstructure with solid  
54 solution strengthening only (absence of precipitation strengthening) as well as the commonly accepted  
55 faster diffusion in the non-closed packed A2 crystal structure. Furthermore, the microstructure was  
56 found to undergo a decomposition into A2 and A3 during creep [29].

57 Apart from the fundamental consideration of precipitation strengthening of an A2 matrix alloy by B2  
58 precipitates, the detailed analysis of the microstructures after creep of the benchmark A1-L<sub>12</sub> Ni-based  
59 superalloys has revealed significant microstructural changes induced by the creep deformation, the most  
60 prominent of which is rafting [32]. Rafting is a phenomenon of directional coarsening of the precipitates

61 during creep deformation. A rafted microstructure is essential to maintain high creep resistance at high-  
 62 temperature and low stress condition, which can increase the dislocation climb distance at the interface  
 63 between  $L1_2$  and A1 [33]. The lattice misfit at the interface between precipitate and matrix not only  
 64 plays an important role in the formation of the rafted microstructure, but also affects the directional  
 65 relationship between the rafted microstructure and the external stresses [34]. This lattice misfit for the  
 66 discussed system (in analogy to the Ni-based superalloy case) can be quantified by the following  
 67 equation:

$$\delta = \frac{2(a_{B2} - a_{A2})}{a_{B2} + a_{A2}} \quad (1)$$

68 where,  $\delta$  represents the lattice misfit, while  $a_{A2}$  and  $a_{B2}$  represent the lattice parameters of A2 and B2,  
 69 respectively.

70 Table 1 shows the correlation between the sign of lattice misfit  $\delta$  (+/-), loading direction  
 71 (tensile/compression) and the type of rafting (perpendicular/parallel to the applied uniaxial stress)  
 72 observed during creep of A1- $L1_2$  Ni- and Co-based superalloys. When an alloy with a negative  $\delta$  (-) is  
 73 subjected to tensile creep (+) (i.e. the sign of the lattice misfit and the loading type are opposite to each  
 74 other), the rafting direction of the  $L1_2$  precipitates is reported to be perpendicular to the loading direction,  
 75 which is known as N-type rafting [35]. Similarly, when an alloy with negative  $\delta$  (-) is subjected to  
 76 compression creep stress (-) (i.e. the sign of the lattice misfit and the loading type are same as each  
 77 other), the rafting direction of the precipitates is parallel to the loading direction, which is known as P-  
 78 type rafting [35]. It has to be noted that most of the studies focused on single crystalline (SX) superalloys  
 79 and the creep was performed with uniaxial stress along the  $\langle 001 \rangle$  direction. Even in the studies involving  
 80 poly-crystals (PX), the N-type rafting was observed only for grains with close to  $\langle 001 \rangle$  direction parallel  
 81 to the loading direction [36]. A schematic for this relationship between the sign of  $\delta$  and the type of  
 82 uniaxial loading as reported for Ni- and Co- based superalloys is shown in Fig. 1.

Table 1. Dependence of the directional relationship between the loading direction and the rafted microstructure in Ni- and Co-based A1- $L1_2$  superalloys on the sign of lattice misfit  $\delta$  (+/-) and uniaxial loading type (+, tension/-, compression).

Base	Designation	SX/PX	Misfit	Loading Type	Rafting Type	Ref.
Ni	ERBO 1				Perpendicular (N)	[34]
	TMS-75	SX	-	+	Perpendicular (N)	[35]
	LEK 94				Perpendicular (N)	[36]
	M951G		-	+	Perpendicular (N)	[40]
	IN738LC	PX	+	-	Perpendicular (N) at high stress positions for a conical compression sample	[41]
Co	Co-Al-W				Parallel (P)	[39]
	Co-Al-W-Ta-Ti	SX	+	+	Parallel (P)	[40]
	Co-Al-W-B			-	Perpendicular (N)	[44]
	Co-Al-W-B	PX	+	-	Perpendicular (N) only in $\langle 100 \rangle$ parallel to loading axis	[42], [43]

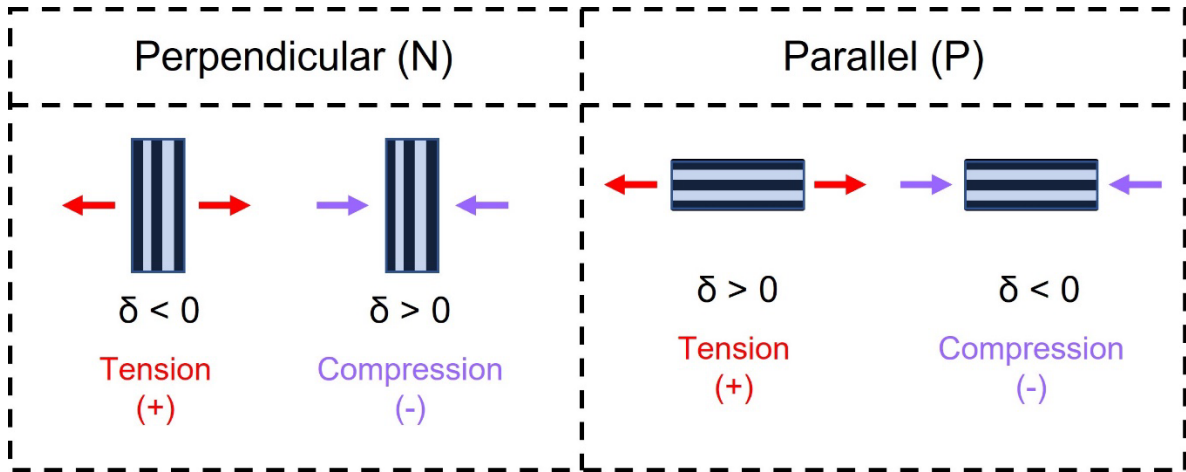


Fig. 1. The schematic of the dependence of rafting direction on the sign of lattice parameter  $\delta$  and loading direction in Ni- and Co-based superalloys.

83 Despite the considerable efforts to develop A2-B2 RCCA, there is currently no systematic assessment  
 84 of the creep behavior and mechanisms of such alloys as well as the impact of creep deformation on the  
 85 A2-B2 microstructure available in literature. Thus, we present a first systematic study on the creep  
 86 behavior of 27.3Ta-27.3Mo-27.3Ti-8Cr-10Al (in at. %, abbreviated by TMT-8Cr-10Al in what follows)  
 87 as a model alloy introduced by Laube et al. [27,28]. It has been confirmed via differential scanning  
 88 calorimetry (DSC) that B2 precipitates form by a diffusion-controlled precipitation below temperatures  
 89 of around 1055 °C [27]. Additionally, a stable two-phase microstructure was found even after prolonged  
 90 exposure at temperatures close to the solvus temperature [28]. These findings suggest promising  
 91 prospect for achieving high-temperature creep resistance in this alloy.

92 In detail, we focus on the following fundamental research questions:

- 93 1. How does the creep behavior of a two-phase, precipitation-strengthened A2-B2 TMT-8Cr-10Al  
 94 compares to that of state-of-the-art, single-crystalline Ni-based superalloys with A1-L1<sub>2</sub>  
 95 microstructure as well as to poly-crystalline, single-phase A2 or B2 RHEA?
- 96 2. How does the microstructure of TMT-8Cr-10Al evolves during the creep loading?
- 97 3. How do the microstructural changes depend on the coherency and lattice misfit between the B2  
 98 precipitates and the A2 matrix?

## 2. Experiments and Materials

99 TMT-8Cr-10Al was synthesized by repetitive arc melting under Ar atmosphere using an AM/0.5 furnace  
 100 supplied by Edmund Bühler GmbH (Germany). The purities of the elements Ta, Mo, Ti, Cr, and Al  
 101 provided by chemPUR GmbH (Germany), were 99.9%, 99.95%, 99.8%, 99.99%, and 99.99%,  
 102 respectively. One button, weighing approximately 150 g, was synthesized by five steps of flipping and  
 103 re-melting. The button was then cast into a cuboid shape using a water-cooled Cu mold. Subsequently,  
 104 the as-cast TMT-8Cr-10Al alloy was wrapped in Mo foil (to prevent any forms of oxidation) for  
 105 homogenization at 1600 °C for 20 h to remove the as-cast dendritic microstructure at a heating and  
 106 cooling rate of 100 K/h in a HTRT 70-600/18 resistance tube furnace (Carbolite Gero GmbH & Co. KG,  
 107 Germany) with three evacuation cycles and backfilling with continuous Ar flow in order to prevent  
 108 oxidation. Further details on the synthesis process can be found in previous studies [27,28].

109 To investigate the microstructure, samples were cut from the button by electrical discharge machining  
110 (EDM) were first ground with SiC abrasive paper to P4000 grit. Afterwards, 3  $\mu\text{m}$  and 1  $\mu\text{m}$  poly-  
111 crystalline diamond suspension were used for polishing for 5 min each. Then, the samples were polished  
112 for 10 min using a colloidal oxide polishing suspension provided by Buehler ITW (Germany) on a semi-  
113 automatic machine. The final preparation step was chemo-mechanical vibratory polishing using a non-  
114 crystallizing oxide polishing suspension provided by Struers (Germany) for at least 16 h. Scanning  
115 electron microscopy (SEM) was performed using on a Zeiss LEO 1530 (Zeiss, Germany) at 20 kV  
116 acceleration voltage. Electron backscatter diffraction (SEM-EBSD) with an EDAX Digiview system  
117 (Ametek, USA) in a Zeiss Auriga 60 (Zeiss, Germany) was used to image the local crystallographic  
118 orientation and to identify secondary phases. Composition was determined by inductively coupled  
119 plasma-optical emission spectroscopy (ICP-OES) of the homogenized TMT-8Cr-10Al alloy and  
120 scanning electron microscopy-based energy dispersive X-ray spectroscopy (SEM-EDS) in the Zeiss  
121 Auriga 60 system with Octane Super-A detector (Ametek, USA). The levels of O and N impurities were  
122 detected in five samples of the TMT-8Cr-10Al by hot carrier gas extraction (HCGE) utilizing a TC500  
123 system supplied by Leco Instrumente GmbH, Germany. The density of the alloy was calculated  
124 according to Archimedes' principle by selecting five samples to calculate the average density. The  
125 density measurements were conducted with the weighing machine ME204T/00 with a density measuring  
126 kit ME-DNY-43 from the company Mettler-Toledo GmbH (Germany).

127 For transmission electron microscopy (TEM), electron-transparent foils were prepared using a lift-out  
128 technique with a ThermoFisher Scios-2 (USA) system equipped with a dual beam focused ion beam  
129 (FIB). Initially, a protective platinum layer was deposited, followed by FIB sectioning at 30 kV with  
130 probe currents of 5 nA and 1 nA, respectively. The TEM lamellae were then lifted out and attached to a  
131 molybdenum half-grid through platinum deposition. Subsequently, the lamellae were thinned to  
132 approximately 80 nm using various voltage and current combinations in the FIB. Final cleaning was  
133 performed at 2 kV and 10 pA for an extended period.

134 Diffraction contrast imaging was performed using a ThermoFisher Tecnai T20-ST (USA) TEM operated  
135 at 200 kV in parallel illumination mode. The atomic structure of the phases was investigated using  
136 scanning transmission electron microscopy (STEM) in a probe-corrected ThermoFisher Titan Themis  
137 (USA) TEM operated at 300 kV. The TEM foil was tilted to a low-index zone axis for diffraction contrast  
138 and atomic resolution imaging. Atomic resolution micrographs were acquired with a high-angle annular  
139 dark field (HAADF) detector and a camera length of 160 mm. Specimen drift during atomic resolution  
140 imaging was corrected using the drift-corrected frame integration (DCFI) function of Velox software.  
141 Lattice parameters of the phases were estimated by filtering a specific reflection in the fast Fourier  
142 transformation (FFT) of the high-resolution HAADF-STEM micrographs and measuring interplanar  
143 spacing with the line profile tool in Gatan Microscopy Suite software.

144 The samples for compression creep tests with dimensions of  $(5 \times 3 \times 3) \text{ mm}^3$  were prepared by electrical  
145 discharge machining (EDM). Before creep testing, the surface of each sample was ground to remove the  
146 oxidation/EDM layer. Compression creep tests were carried out under vacuum (to exclude  
147 environmental effects on creep) utilizing a Z100 electro-mechanical universal testing machine provided  
148 by ZwickRoell GmbH & Co. KG (Germany) equipped with a vacuum furnace supplied by Maytec  
149 GmbH (Germany). The test temperatures ranged from 1000  $^{\circ}\text{C}$  to 1060  $^{\circ}\text{C}$  and true stresses of 75 MPa,  
150 100 MPa and 125 MPa were applied. All compression creep tests were started after a duration of 30 min  
151 under a preload 50 N at the test temperature to ensure uniform temperature distribution. After the creep  
152 test, the samples were cooled to room temperature by furnace cooling and then prepared for

153 microstructural investigations using the metallographic steps described above. At least three tests were  
 154 performed at each test condition in order to ensure the relevance of the results. In order to track the  
 155 impact of long-term heat treatment under the same conditions without external mechanical load, stress-  
 156 free heat treatments were also performed at the same temperatures and durations compared to the  
 157 compression creep tests. The samples used for heat treatment were wrapped in Ta foil and then sealed  
 158 into fused silica tubes under Ar atmosphere to avoid oxidation during the heat treatment process. After  
 159 quenching, microstructural studies were carried out on the heat-treated samples following the above-  
 160 mentioned metallographic steps.

### 3. Results and Discussion

#### 3.1 Initial Composition and Microstructure

161 In order to assess the chemical composition of TMT-8Cr-10Al, ICP-OES and HCGE were conducted  
 162 and the results are shown in Table 1. The composition of the constituent elements observed via ICP-  
 163 OES is within the experimental accuracy range. The levels of O and N are found to be between 100 and  
 164 500 wt.-ppm and are comparable to other studies in RHEA [27,46–48]. The chemical homogeneity of  
 165 this alloy at the microscale determined by SEM-EDS was reported previously [27]. The experimental  
 166 density of the alloy determined by the Archimedes method is  $9.6 \text{ g}\cdot\text{cm}^{-3}$ .

Table 2. Composition, impurity levels and density of the homogenized TMT-8Cr-10Al.

	Composition in at. %				Impurities in wt.-ppm		Density in $\text{g}\cdot\text{cm}^{-3}$	
	Ta	Mo	Ti	Cr	Al	O		N
<b>Target</b>	27.3	27.3	27.3	8	10	–	–	
<b>Experimental</b>	26.0	29.8	27.4	7.3	9.4	$352 \pm 88$	$45 \pm 14$	9.6

167 Fig. 2 (a) shows an inverse pole figure (IPF) map of the TMT-8Cr-10Al obtained by SEM-EBSD in the  
 168 as-homogenized condition. It can be observed that the orientation of the grains is random indicating no  
 169 preferred orientation present in the initial microstructure. In the orientation-contrast SEM-BSE  
 170 micrographs, a coarse-grained microstructure as well as some pores (displayed as dark features) are  
 171 revealed, as shown in Fig. 2 (b). The appearance of pores is due to the Kirkendall effect that occurs  
 172 during homogenization of dendritic cast microstructures with significant Al enrichment in the  
 173 interdendritic regions [49]. The mean grain size of the investigated samples is above  $200 \mu\text{m}$ . At higher  
 174 magnification, two different SEM-BSE contrasts within the grains at the nanoscale can be observed as  
 175 shown in Fig. 2 (c). Those contrasts are attributed to the difference in atomic number  $Z$  between A2 and  
 176 B2. The composition of these phases was reported earlier [27], the continuous bright matrix is enriched  
 177 in Ta and Mo while the dark precipitates are enriched in Al and Ti, with Cr distributed uniformly  
 178 throughout the alloy. The bright matrix phase was identified as a disordered A2 and the discontinuous  
 179 dark phase as ordered B2 precipitates via TEM experiments in Refs. [27], [28]. Additional dark particles  
 180 (marked with white arrows in (c)) of less than 0.1 vol.% were also observed at grain boundaries, which  
 181 were previously identified to be Ti-rich nitrides resulting from N contamination during arc melting and  
 182 homogenization [28].

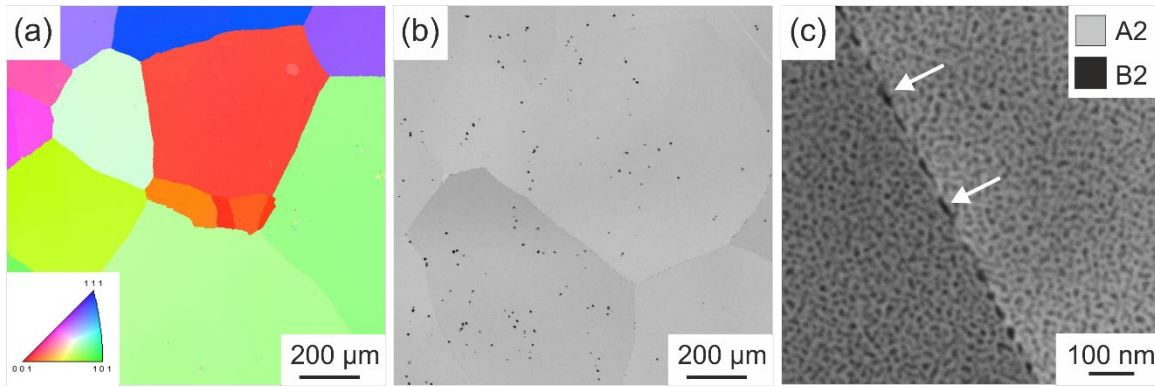


Fig. 2. Microstructure of TMT-8Cr-10Al after homogenization at 1600 °C for 20 h: (a) IPF map (color code in the inset). (b-c) SEM-BSE micrographs at different magnification.

183 In order to determine the coherency relationship between the A2 matrix and B2 precipitates, TEM  
 184 investigation was performed on a sample heat-treated at 1000 °C for 1000 h with a microstructure  
 185 coarsened under similar conditions compared to the creep investigations discussed later in this article.  
 186 Figure 3a shows a dark-field (DF) micrograph with bright-contrast plate-like precipitates uniformly  
 187 distributed throughout the matrix consistent with the SEM-BSE images shown in Fig. 2 (c). The inset  
 188 displays the corresponding indexed selected area electron diffraction (SAED) pattern acquired along the  
 189  $\langle 011 \rangle$  zone axis from this region. The DF micrograph was obtained using a (weak)  $\{100\}$  superlattice  
 190 spot at half the distance of the  $\{200\}$  A2 fundamental peaks, indicating the presence of the long-range  
 191 ordered phase. The superlattice spots correspond to the B2 phase. Fig. 3 (b) and Fig. 3 (c) present atomic  
 192 resolution STEM micrographs taken near the  $\langle 011 \rangle$  zone axis of the matrix and precipitate, respectively.  
 193 These micrographs reconfirm that the matrix and precipitates have A2 and B2 crystal structure,  
 194 respectively. The lattice parameters of the matrix  $a_{A2}$  and precipitate  $a_{B2}$  were estimated to be 0.315  
 195 and 0.322 nm, respectively. From that, the lattice misfit parameter  $\delta$  between the A2 matrix and B2  
 196 phase with  $\{100\}_{A2} \parallel \{100\}_{B2}$  and  $\langle 001 \rangle_{A2} \parallel \langle 001 \rangle_{B2}$  is determined to be positive with a value of  
 197  $+0.022$  according to Eq. 1.

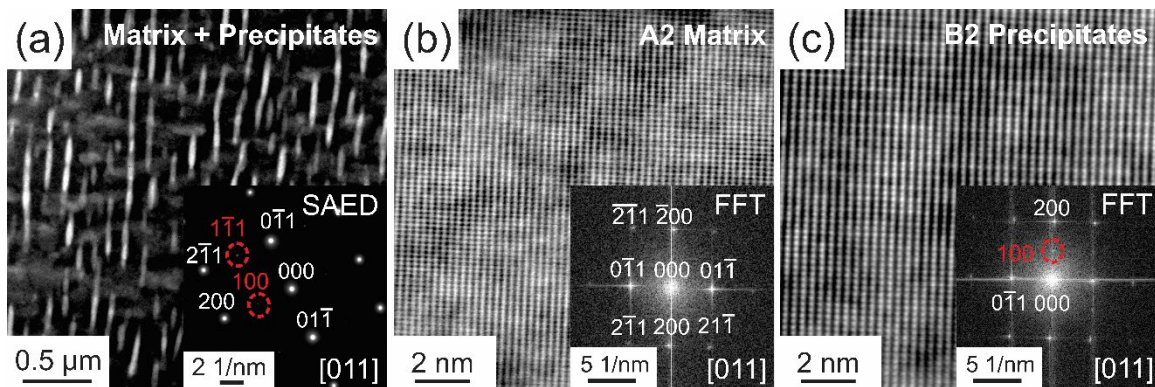


Fig. 3. TEM results of the TMT-8Cr-10Al alloy after heat treatment at 1000 °C for 1000 h: (a) DF TEM micrograph showcasing bright contrast B2 precipitates uniformly distributed throughout the matrix. The micrograph was acquired using a  $\{100\}$  superlattice spot in the corresponding SAED pattern taken along a  $\langle 011 \rangle$  zone axis from this region (inset). (b) Atomic resolution micrograph of the matrix with its corresponding FFT in the inset. (c) Atomic resolution micrograph of a B2 precipitate with its corresponding FFT in the inset.

### 3.2 Creep Behavior

198 To understand the creep behavior of TMT-8Cr-10Al, creep tests were conducted at  $(1000 \pm 5)$  °C and  
 199 125 MPa, i.e. about 50 K below the reported solvus temperature of TMT-8Cr-10Al [27]. In order to



200 study the relationship of creep strain with respect to time and to check its reproducibility for a  
 201 precipitation strengthened microstructure, three different creep tests under the same conditions were  
 202 conducted with each test being interrupted at different strains. To study the evolution of the creep rate  
 203 with increasing temperature, the creep temperature was varied between 1000 to 1060 °C, keeping the  
 204 applied true stress at 125 MPa.

205 Fig. 4 (a) shows the compression creep curves of TMT-8Cr-10Al with respect to time for total strains of  
 206 0.5%, 0.6% and 0.8%. The creep rate at the end of each test was found to be approximately  $2.5 \cdot 10^{-9} \text{ s}^{-1}$ .  
 207 In order to visualize the creep rate for these three tests, the logarithm of creep rate vs. creep strain was  
 208 plotted as shown in Fig. 4 (b) with blue lines (dashes and dotted) ending in an open circle symbol. Fig. 4  
 209 (b) also shows the variation of the true creep rate vs. creep strain with solid lines ending in an open circle  
 210 symbol for tests at different temperatures (higher than 1000 °C) and a constant stress of 125 MPa.

211 It is relevant to note here that between the curves for 1000 and 1030 °C (a difference of 30 K), the  
 212 minimum creep rate jumps by one order of magnitude. However, between 1030 °C and 1040 °C (a  
 213 difference of 10 K), the minimum creep rate changes by one order of magnitude signifying a significant  
 214 reduction in the creep resistance between 1030 and 1040 °C. This can also be observed from the values  
 215 in Table 2 that the  $\dot{\epsilon}_{\text{min}}$  substantially increases (which indicates a reduction in creep resistance) between  
 216 1030 and 1040 °C. These two regions,  $\leq 1030 \text{ °C}$  and  $\geq 1040 \text{ °C}$ , are demarcated as high creep  
 217 resistance (HCR) regime which exhibits a good creep resistance and the low creep resistance (LCR)  
 218 regime where the creep resistance is notably lower. These two regions are highlighted by blue and red  
 219 lines or symbols.

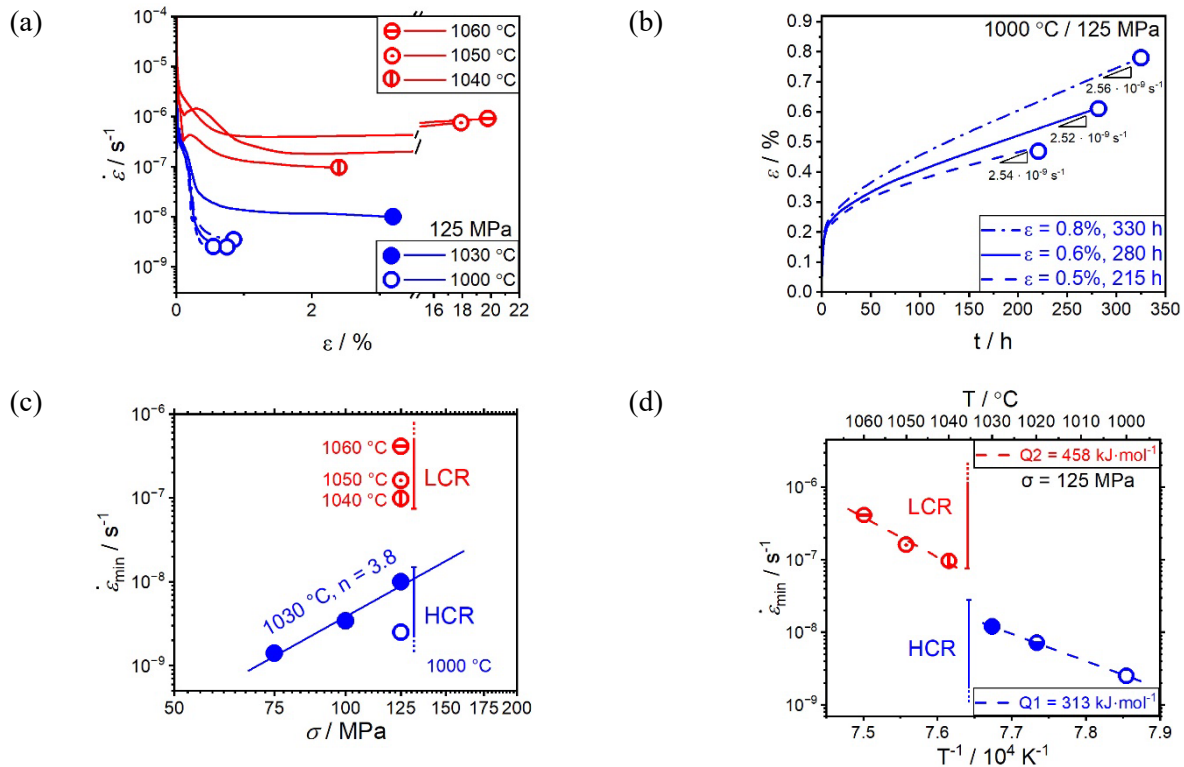


Fig. 4. Creep behavior of TMT-8Cr-10Al: (a) creep strain  $\epsilon$  as a function of time  $t$  for 125 MPa and 1000 °C condition. (b) Logarithmic plot of strain rate  $\dot{\epsilon}$  vs. strain  $\epsilon$  for 125 MPa at different temperatures. (c) Arrhenius plot for 125 MPa. (d) Norton plot. The creep regimes are demarcated as High Creep Resistance (HCR) marked in blue and Low Creep Resistance (LCR) depicted in red.



220 Here, a distinction has to be noted between steady state creep rate ( $\dot{\epsilon}_{\text{steady}}$ ) and minimum creep rate  
 221 ( $\dot{\epsilon}_{\text{min}}$ ). In order to qualify for a mechanical [50] or even microstructural steady state [51], a substantial  
 222 plastic strain release should be obtained in the period of constant strain rate. All the three tests performed  
 223 at 1000 °C do not reach a steady state according to this definition. However, the creep tests performed  
 224 at temperatures higher than 1000 °C reveal this steady state creep clearly. It must be noted that a steady  
 225 state regime at 1000 °C could, in principle, be experimentally verified but a plastic strain release of for  
 226 example 5% would take close to 200 days. Based on the creep curve exhibiting a steady state regime at  
 227 1030 °C (full blue circle in Fig. 4, where the A2-B2 microstructure is still stable), we can infer that the  
 228 creep rate of  $2.5 \cdot 10^{-9} \text{ s}^{-1}$  at 1000 °C might also be regarded as the steady state creep rate. Unlike the  
 229 creep curves for A1-L1<sub>2</sub> superalloys [33], the  $\dot{\epsilon}_{\text{steady}}$  observed for the TMT-8Cr-10Al at all  
 230 temperatures are also incidentally the  $\dot{\epsilon}_{\text{min}}$ . Hence, for the discussion that follows next, we will denote  
 231 the  $\dot{\epsilon}_{\text{steady}}$  as  $\dot{\epsilon}_{\text{min}}$ , since both are essentially the same for the TMT-8Cr-10Al alloy and it enables us to  
 232 make a direct comparison to the creep behavior of other materials.

Table 2.  $\dot{\epsilon}_{\text{min}}$  of TMT-8Cr-10Al measured at 125 MPa with different temperatures.

$\sigma$ / MPa	125				
T / °C	1000	1030	1040	1050	1060
$\dot{\epsilon}_{\text{min}}$ / s <sup>-1</sup>	$2.5 \cdot 10^{-9}$	$1.0 \cdot 10^{-8}$	$9.6 \cdot 10^{-8}$	$1.6 \cdot 10^{-7}$	$4.1 \cdot 10^{-7}$

233 The logarithm of  $\dot{\epsilon}_{\text{min}}$  for all temperatures is plotted in Fig. 4 (c) against the inverse of the temperature  
 234 according to the Mukherjee-Bird-Dorn equation [52]:

$$\dot{\epsilon}_{\text{min}} = A \frac{D_0 G b}{k T} \left(\frac{b}{d}\right)^p \left(\frac{\sigma}{G}\right)^n \exp\left(-\frac{Q}{k T}\right) \quad (2)$$

235 where,  $\dot{\epsilon}_{\text{min}}$  is the minimum creep rate,  $A$  the Dorn constant,  $D_0$  a pre-exponential factor,  $G$  the shear  
 236 modulus,  $b$  the Burgers vector,  $k$  the Boltzmann constant,  $T$  the temperature,  $d$  the grain size,  $p$  the  
 237 inverse grain size exponent,  $\sigma$  the applied stress,  $n$  the stress exponent,  $Q$  the activation energy. A linear  
 238 dependence is obtained.

239 For TMT-8Cr-10Al, two different parts of the linear fit can be observed. For each part of the Arrhenius  
 240 plot, two distinct, apparent activation enthalpy values ( $Q$ ) can be evaluated. These two regimes  
 241 incidentally also fall in the ranges of  $\leq 1030$  and  $\geq 1040$  °C, in line with the demarcation suggested for  
 242 Fig. 4 (b). Hence the color scheme of blue and red for  $\leq 1030$  and  $\geq 1040$  °C, respectively, has been  
 243 maintained for Fig. 4 (c). The apparent activation energy for creep was determined to be 313 kJ/mol in  
 244 the temperature range of HCR and 458 kJ/mol in the temperature range of LCR. For pure metals, a  
 245 possible correlation between the apparent activation enthalpy observed during creep and the activation  
 246 enthalpy for the self-diffusion of the metal atoms was reported [53]. However, this straightforward  
 247 correlation becomes difficult when multi-component, single phase or even multi-phase materials are  
 248 addressed [54]. Considering that TMT-8Cr-10Al not only has five principal elements but also two  
 249 different phases along with grain boundaries, the correlation of the obtained activation enthalpy for creep  
 250 with the activation enthalpy for the self-diffusion of the constituent elements is not trivial.

251 In order to determine the apparent stress exponent  $n$  according to Eq. (2) [52], creep tests were  
 252 conducted at 1030 °C at three different stresses of 75, 100 and 125 MPa. The results are shown in  
 253 Fig. 4 (d). The  $n$  value is determined by plotting the logarithm of minimum creep rate at constant  
 254 temperature against the different applied stresses. The apparent stress exponent  $n$  determined by linear

255 fitting is 3.8. When compared to pure metals and single-phase alloys,  $n$  values between 3 and 5 indicate  
 256 dislocation climb as the rate controlling process of creep [55]. It should be noted that similar to the  
 257 activation enthalpy discussed earlier, the evaluated  $n$  is an apparent quantity. This is due to the fact that  
 258 the observed creep behavior is an effective behavior potentially involving dislocation glide, dislocation  
 259 climb as well as grain boundary sliding in TMT-8Cr-10Al. Similar to the activation enthalpy, a simple  
 260 comparison to single-phase metals or alloys might lead to a false conclusion without proper  
 261 microstructural evidence for the active micromechanical mechanisms.

262 For a precipitation strengthened material like TMT-8Cr-10Al, it is imperative to compare the macro-  
 263 mechanical properties like creep with other high temperature alloys. Fig. 5 shows the creep results of  
 264 our poly-crystalline TMT-8Cr-10Al consisting of A2 matrix and B2 precipitates as compared to those  
 265 of single-crystal Ni-based superalloy CMSX-4 with A1 matrix and  $L1_2$  precipitates [56,57], poly-  
 266 crystalline single-phase, A2 TiZrHfNbTa [29] and single-phase B2 NbMoCrTiAl [58]. The minimum  
 267 creep rates at low stresses of 100 to 200 MPa of poly-crystalline TMT-8Cr-10Al at 1030 °C are  
 268 comparable to that of single-crystalline CMSX-4 at 1050 °C as can be seen in Fig. 5. It must be noted  
 269 that TMT-8Cr-10Al tested at 1030 °C is operated much closer to the reported solvus temperature of the  
 270 B2 precipitates ( $T_S$ ) of 1055 °C ( $0.98 T_S$ ) compared to that in the case CMSX-4 tested at 1050 °C with  
 271 a solvus temperature of  $L1_2$  of 1280 °C ( $0.85 T_S$ ). At comparable temperatures and stresses, TMT-8Cr-  
 272 10Al has a minimum creep rate about four orders of magnitude lower than both, the single-phase A2  
 273 TiZrHfNbTa alloy and the single-phase B2 NbMoCrTiAl alloy, which undergoes a microstructural  
 274 decomposition to a multi-phase intermetallic microstructure during creep [58]. This reaffirms the  
 275 substantial impact of precipitation strengthening achieved in TMT-8Cr-10Al and provides a pathway  
 276 for developing RHEA with a two-phase microstructure in high-temperature creep applications.

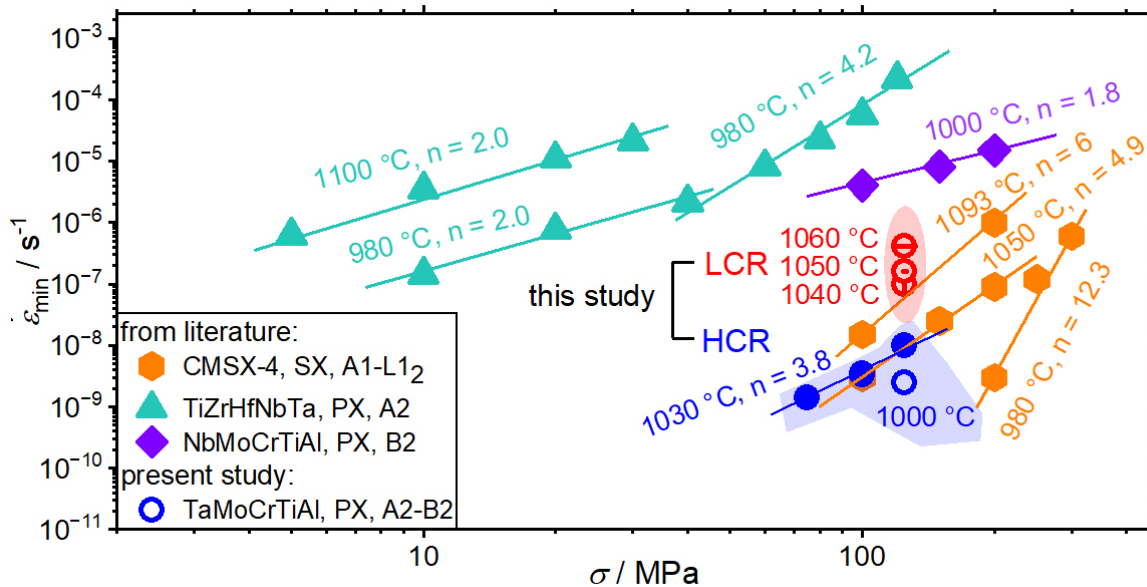


Fig. 5. Comparison of strain rate vs. applied stress for investigated poly-crystalline TMT-8Cr-10Al alloy and poly-crystalline B2 NbMoCrTiAl alloy [58], poly-crystalline A2 TiZrHfNbTa [29] as well as for a single-crystalline Ni-based superalloy CMSX-4.

### 3.3 Microstructure evolution after creep deformation

277 Fig. 6 (a-c) shows the SEM-BSE micrographs and the evolution of TMT-8Cr-10Al after creep  
278 deformation at 1000 °C/125 MPa for strains of 0.5, 0.6, and 0.8%, respectively. The compression  
279 direction in the micrographs is horizontal. In the absence of creep deformation, the size of the B2  
280 precipitates were previously reported to increase with increasing heat treatment times while the phase  
281 fraction of the precipitates was found to initially increase (the growth regime) before saturating after a  
282 certain annealing time (the coarsening regime) [28]. To illustrate the difference in the microstructure  
283 after annealing without creep deformation, samples were heat treated at 1000 °C for times equivalent to  
284 each creep test, as displayed in Fig. 6 (d-f).

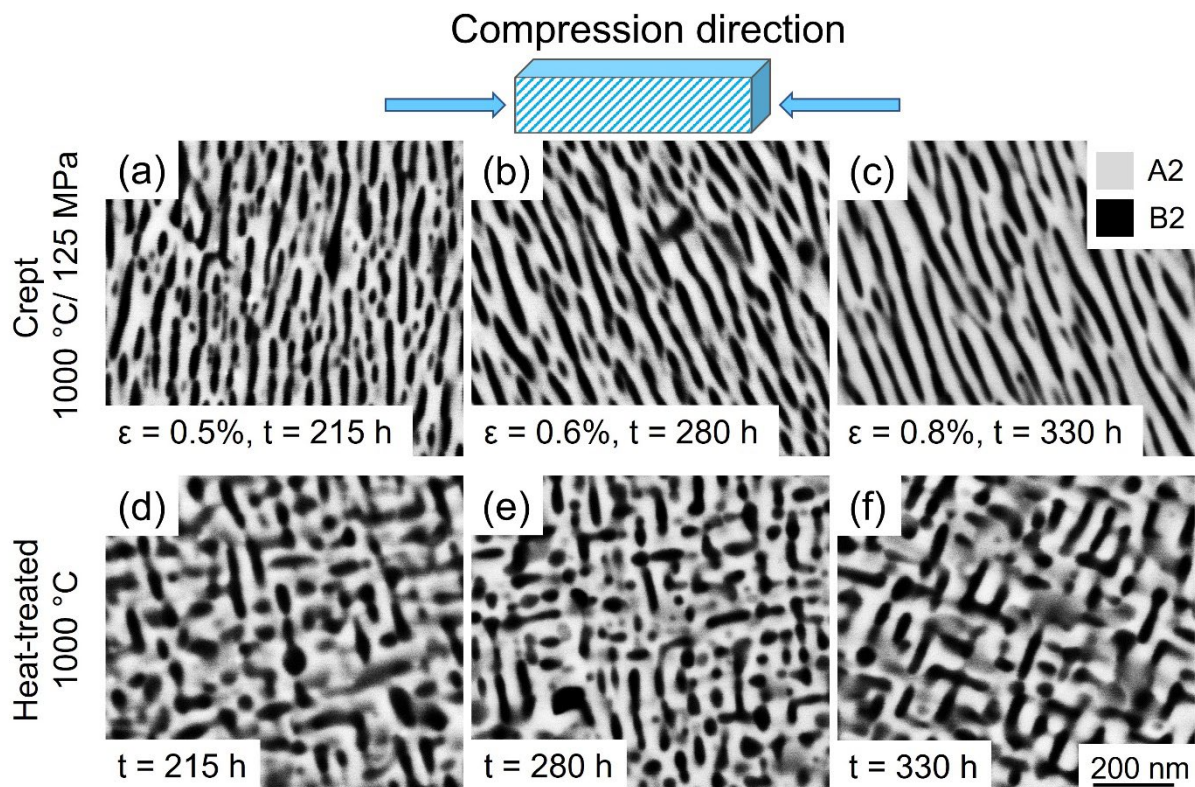


Fig. 6. SEM-BSE micrographs showing microstructure evolution of TMT-8Cr-10Al alloy after creep tests at 1000 °C with a stress of 125 MPa and strains of: (a) 0.5%, (b) 0.6%, and (c) 0.8% as well as heat treatments at 1000 °C for: (d) 215 h, (e) 280 h, and (f) 330 h.

285 Fig. 7 depicts the changes in phase fraction of the precipitates at 1000 °C after creep and heat treatment  
286 for the different times and microstructures as revealed in Fig. 6. The increase in the phase fraction of the  
287 precipitates (calculated by area fraction in the SEM-BSE images) is evident in both, the crept and  
288 annealed samples. The phase fraction of precipitates increases over the tested time range, which indicates  
289 that the precipitates are still in the process of growth during the creep deformation as opposed to being  
290 in the coarsening process where a saturation of the precipitate fraction is expected. Furthermore, it can  
291 be seen from Fig. 7 that the precipitate growth is accelerated under creep conditions as compared to the  
292 conditions without plastic deformation imposed. The impact of additional deformation on the kinetics  
293 of precipitate growth can be attributed to the dislocation activity. Dislocations might act as  
294 heterogeneous nucleation sites as well as fast diffusion paths enhancing both, the nucleation and growth  
295 rates, respectively.

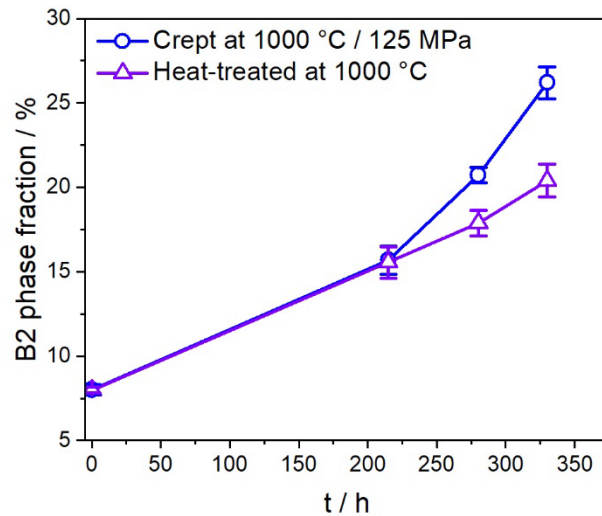


Fig. 7. The phase fraction of B2 precipitates of TMT-8Cr-10Al at 1000 °C as a function of time of creep deformation (125 MPa) and annealing.

296 Besides the change in the fraction of the precipitates, the other notable difference between the crept and  
 297 the annealed samples is in the morphology of the precipitates. In contrast to the heat-treated samples,  
 298 the crept microstructures show a pronounced directional growth similar to the rafting phenomenon of  
 299 precipitates in Al-L1<sub>2</sub> Ni-based superalloys. As mentioned earlier, the majority of the creep experiments  
 300 on Al-L1<sub>2</sub> Ni-based superalloys were performed on  $\langle 100 \rangle$  oriented single crystals. Since the samples  
 301 used in our study are poly-crystalline, the correlation between the grain orientation, and the angle  
 302 between the loading direction to the rafted precipitates is not straightforward. In Al-L1<sub>2</sub> CMSX-4, the  
 303 resistance to creep in compression was reported to decrease in the sequence of  $\langle 100 \rangle$ ,  $\langle 111 \rangle$  and  $\langle 110 \rangle$   
 304 directions parallel to the compression direction. This was attributed to the resolved shear stresses on slip  
 305 systems for each orientation [59]. This differentiation is also of great relevance for creep studies in our  
 306 poly-crystalline A2-B2 alloy since grain orientation might affect the creep response. Due to the absence  
 307 of single crystalline samples, the direct correlation of creep resistance to the grain orientation is not  
 308 possible since the obtained creep response are a result of the entire sample containing multiple grains of  
 309 different orientation.

310 In order to study such a correlation, EBSD investigations were performed on samples crept at 1000 °C  
 311 up to a strain of 0.5% (Fig. 8 (a)) and 0.8% (Fig. 8 (b)). The relationship between the loading direction  
 312 and the surface on which EBSD was performed is illustrated in Fig. 8. The IPF map is plotted such that  
 313 the directions are color-coded by the standard triangle of the compression direction in the inset. A few  
 314 grains in the IPF map for each sample in Fig. 8 are selected (marked 1-4) to represent the orientations  
 315 close to the corners of the standard triangle. The SEM-BSE images highlighting the rafting direction for  
 316 the grains marked in Fig. 8 (a) and (b) are shown in Fig. 8 (a1) to (a4) and (b1) to (b4), respectively.



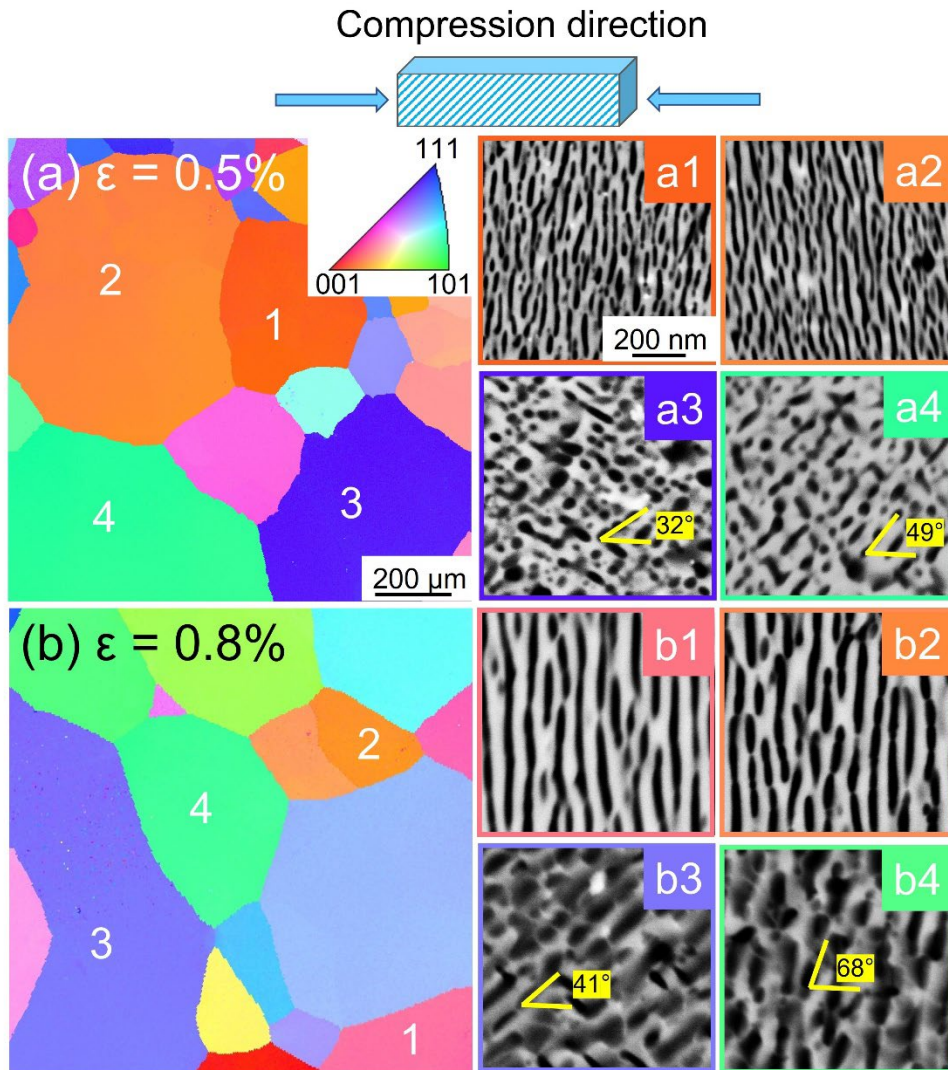


Fig. 8. EBSD-IPF maps and the direction of rafting (1-4) in each corresponding grain. The creep tests were performed at 1000 °C for total strains of (a) 0.5% and (b) 0.8%.

317 As was summarized in Table 1, N-type rafting is observed when the sign of lattice misfit (+/-) is  
 318 opposite to the loading type, e.g. compression (-)/tension (+). For the present case, this was verified as  
 319 the lattice misfit of the B2 precipitates was observed to be positive (+) via HR-TEM in Fig. 3(b-c) and  
 320 compression (-) was applied during creep. Two observations in this regard are important to conclude  
 321 from Fig. 8. Firstly, the direction of rafting with respect to the compression direction is not unique and  
 322 strongly depends on the grain orientation. Secondly, the direction of rafting is clearly perpendicular to  
 323 the compression direction for the grains where the  $\langle 100 \rangle$  direction is parallel to the compression  
 324 direction. This observation is similar to the creep results in poly-crystalline Co-based superalloys, where  
 325 rafting perpendicular to the compression direction was observed only for  $\langle 100 \rangle$  oriented grains [36,45].  
 326 The angles between the rafted structure and the loading direction lie between 45 to 70° in grain  
 327 orientations other than  $\langle 100 \rangle$ . Thus, it can be concluded that similar to the A1-L1<sub>2</sub> Ni- and Co-based  
 328 superalloys, the direction of the rafted microstructure can be predicted following the same scheme  
 329 presented in Table 1 for grains whose  $\langle 100 \rangle$  directions are close or even parallel to the compression  
 330 direction.

331 Since a significant reduction of creep resistance was obtained at temperatures  $\geq 1040$  °C (LCR in Fig. 4)  
 332 with increasing temperature as shown in Fig. 4 (b), it is required to correlate the evolution of  
 333 microstructure in the creep samples at higher temperatures. Fig. 9 (a-c) shows the microstructural  
 334 evolution of crept samples at temperatures ranging from 1030 to 1060 °C at a constant stress of 125 MPa.

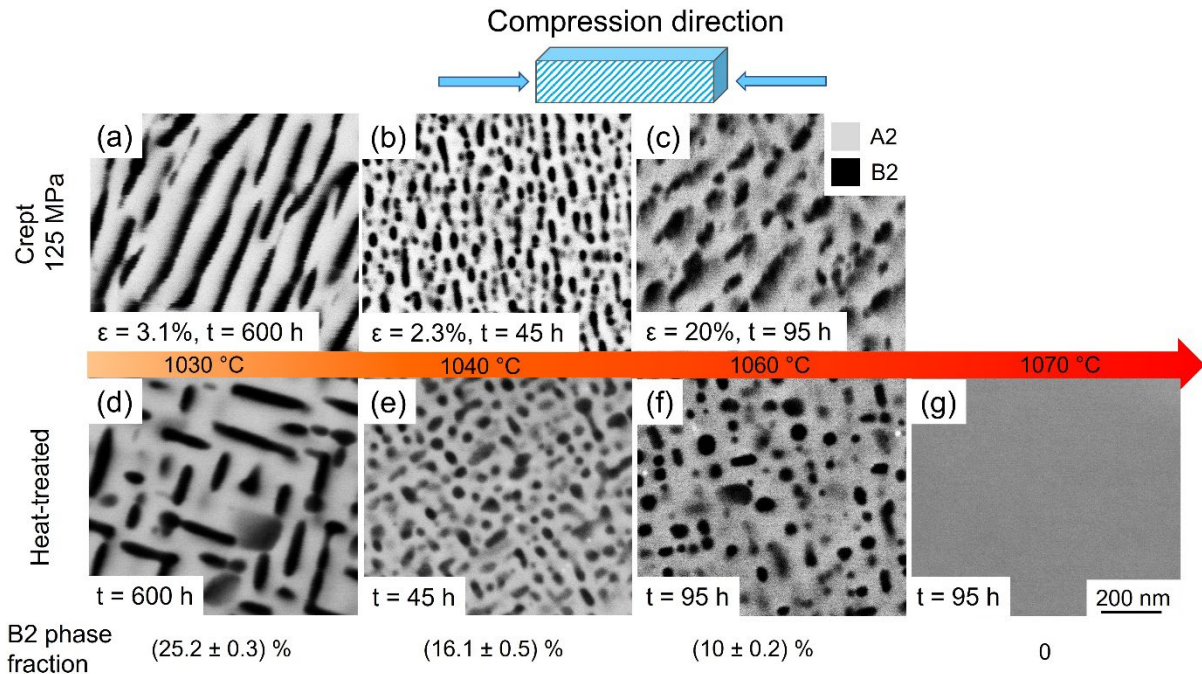


Fig. 9. Microstructure evolution of TMT-8Cr-10Al alloy after creep tests and heat treatments: (a) 1030 °C/125 MPa/3.1%, (b) 1040 °C/125 MPa/2.3%, (c) 1060 °C/125 MPa/20%. (d) 1030 °C/600 h, (e) 1040 °C/45 h, (f) 1060 °C/95 h and (g) 1070 °C/95 h. Same magnification used in all images.

335 The microstructural features after creep at 1030 °C are quite similar to those at 1000 °C in terms of  
 336 rafting of the precipitates. In contrast, the microstructure for sample crept at 1040 °C and higher appears  
 337 different. The microstructure of the sample crept at 1060 °C, shows a much smaller rafted length as  
 338 compared to the sample crept at 1030 °C, even though the total strain is over six times larger. It must be  
 339 noted however that the creep time for each temperature condition is not the same. The samples for the  
 340 microstructure Fig. 9 (a-c) were cooled inside the furnace after the completion of the creep experiments.  
 341 This furnace cooling allows the precipitates to re-nucleate during the cooling process. Thus, the  
 342 microstructures shown in Fig. 9 (a-c) may not depict the actual state of the samples during the creep  
 343 experiments. In order to visualize the state of the sample as it would have been during the creep  
 344 experiments, samples were additionally annealed at temperatures and times equivalent to the creep tests  
 345 and then subsequently quenched in water. These micrographs are shown in Fig. 9 (d-g).

346 At 1060 °C the samples still contain B2 precipitates after being annealed for 95 h. In order to determine  
 347 the temperature where the precipitates dissolve completely, one sample was also annealed at 1070 °C  
 348 for 95 h (similar to the time for 1060 °C) wherein no precipitates were found (Fig. 9 (g)). It may further  
 349 be noted that although 1060 °C (Fig. 9 (f)) is a bit higher than the previously reported 1055 °C solvus  
 350 temperature of B2 in TMT-8Cr-10Al but still exhibits B2 precipitates. Single-phase A2 devoid of B2  
 351 precipitates in TMT-8Cr-10Al is only verified for 1070 °C in these experiments. There are many aspects  
 352 which can explain this deviation. Firstly, the determination of the solvus temperature in Ref. [27]  
 353 involves the analysis of the peak temperature of the heat signature in the DSC experiment. However, the

354 formation of the present microstructure involves two reaction types, e.g. a diffusion-controlled 1<sup>st</sup> order  
 355 phase separation as well as a 2<sup>nd</sup> order ordering reaction [27]. The heat signatures of both are  
 356 superimposed in the DSC experiments. While 2<sup>nd</sup> order transitions are indeed evaluated for their  
 357 characteristic peak temperatures [60], diffusion-controlled 1<sup>st</sup> order phase transformations need to be  
 358 characterized by their extrapolated peak-onset temperatures [60–63]. As the lambda-shaped heat  
 359 signature of the 2<sup>nd</sup> order phase transformation spans down to 800 °C in the case of TMT-8Cr-10Al, the  
 360 onset of the diffusion-controlled, characteristic temperatures of the 1<sup>st</sup> order phase separation can only  
 361 be estimated from DSC experiments by the evaluation of the peak temperature. Secondly, it is important  
 362 to consider that the dissolution of the precipitates is a diffusion-controlled process, hence the dissolution  
 363 of the precipitates beyond the solvus temperature is also dependent on the time of annealing. Taking  
 364 these two aspects into account in addition to the possible errors in the different thermocouples used in  
 365 the DSC and furnaces used for annealing and in the creep testing device, it can be said the actual solvus  
 366 temperature would very close to the previously reported solvus temperature of 1055 °C with a possible  
 367 difference of 5 to 10 K.

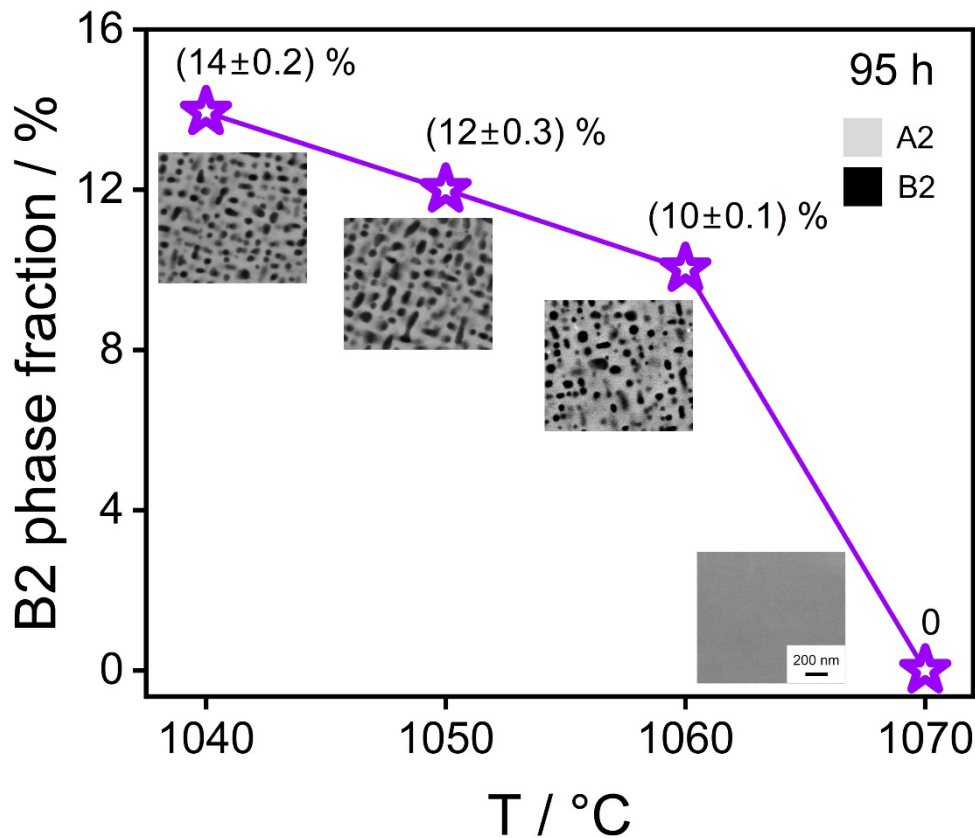


Fig. 10. The phase fraction of B2 precipitates of TMT-8Cr-10Al annealed at 1040 °C to 1070 °C for 95 h. Same magnification used in all images.

368 Although B2 precipitates were observed up to annealing temperatures of 1060 °C, not just the presence  
 369 of precipitates but the phase fraction of the precipitates also plays a key role in determining the creep  
 370 response of the TMT-8Cr-10Al. Since, the annealing times for Fig. 9 (d-g) are not constant, additional  
 371 annealing treatments were conducted at 1040 °C and 1050 °C for 95 h in order to quantitatively analyze  
 372 the temperature depending volume fraction of the precipitates. These results are plotted in Fig. 10. It is  
 373 clear that although the precipitates are still present at 1060 °C, the volume fraction decreases sharply  
 374 from 1040 °C. The higher  $\dot{\epsilon}_{\min}$  observed in the LCR regime (Fig. 4 (b)) is, hence, mainly caused by a



375 reduction in the B2 phase fraction at temperatures close to the actual solvus temperature. It is relevant  
376 to note that creep tests at 125 MPa in the single-phase A2 region  $\geq 1070$  °C repeatedly failed by too fast  
377 creep rates. Apart from benefits of the creep performance of TMT-8Cr-10Al over poly-crystalline A2,  
378 poly-crystalline B2 and also single-crystalline CMSX-4 as showcased in Section 3.2, the relevance of  
379 precipitation strengthening for causing these excellent creep properties is highlighted by these  
380 observations.

## 4. Conclusions

381 The poly-crystalline refractory high entropy alloy TMT-8Cr-10Al (at. %) with a two-phase  
382 microstructure consisting of a disordered A2 matrix and ordered B2 precipitates shows a promising  
383 creep resistance at high temperatures.

384 Compared to poly-crystalline, single-phase A2 and B2 RHEA, it can be concluded that

- 385 • Precipitation strengthening of RCCA by B2 precipitates in A2 matrix is effective as TMT-8Cr-  
386 10Al exhibits a substantially higher creep resistance over single, poly-crystalline A2 and B2  
387 RHEA. The difference amounts to about three to four orders of magnitude in creep rate at  
388 comparable stress and temperature.

389 Compared to state-of-the-art, single-crystalline A1-L1<sub>2</sub> CMSX-4, it can be concluded that

- 390 • TMT-8Cr-10Al exhibits a comparable minimum creep rate compared to CMSX-4 in the stress  
391 range of 75 to 125 MPa and temperatures ranging from 1000 to 1030 °C.
- 392 • The comparable creep resistance under these testing conditions is remarkable as TMT-8Cr-10Al  
393 is poly-crystalline compared to the single-crystalline CMSX-4. Furthermore, TMT-8Cr-10Al  
394 tested at 1030 °C is operated much closer to its solvus temperature of 1060 to 1070 °C compared  
395 to CMSX-4 with 1280 °C.

396 Regarding the microstructural changes upon creep and annealing, the following conclusions can be  
397 drawn:

- 398 • A directional coarsening or rafting of precipitates microstructure of TMT-8Cr-10Al is observed  
399 after crept at 1000 and 1030 °C with a stress of 125 MPa. The combination of positive lattice  
400 misfit and compression loading causes N-type rafting of the microstructure in grains with the  
401  $\langle 100 \rangle$  parallel to the loading direction consistent with the established theories on the formation  
402 of rafted microstructures in Ni- and Co-based superalloys.
- 403 • A reduction in the creep resistance observed at higher temperatures close to the solvus  
404 temperature is correlated to a significant reduction of the B2 phase fraction.

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## Data Availability Statement

413 The data presented in this study are available in KITopen at <https://doi.org/10.35097/0mv9vfxr8rjxbpdj>  
414 under CC BY-SA 4.0 license. Further information is available upon request with  
415 [alexander.kauffmann@kit.edu](mailto:alexander.kauffmann@kit.edu).

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